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Turbulence and turbulence-generated structural loading in wind turbine clusters

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Resume på dansk

Turbulens – i form af standardafvigelse af vindhastighedsfluktuationer – og andre strømningsskarakteristika er forskellige i henholdsvis den fri strømning og strømmingen i det indre af vindmølleparker. Derfor må dimensioneringsforudsætningerne for møller i parker ændres for at give samme sikkerhed mod brud som for enkeltstående møller. Standardafvigelsen af vindhastighedsfluktuationer er en kendt nøgleparameter, for ekstrem- såvel som udmattelseslaster, og i denne rapport søges det sandsynliggjort, at det er nok *alene* at tage hensyn til den ændrede turbulensintensitet i mølleparken ved udmattelsesberegninger. Andre strømningsparametre som turbulensens skala og horisontale og vertikale gradienter af middelvindhastigheden vides også at have indflydelse på møllernes strukturelle dynamik. På den anden side er disse parametre korreleret med turbulensen, negativt eller positivt, og dermed kan en justering af turbulensintensiteten, hvis nødvendigt, repræsentere disse. Således er der i rapporten givet modeller for gennemsnitsturbulensen i mølleparken samt for turbulensen direkte i skyggen af en anden mølle. Endvidere er principperne for addition af udmattelsesvirkningen af de forskellige lasttilfælde givet. Kombinationen af lasttilfælde involverer en vægtningsmetode omfattende hældningen af det aktuelle materiales Wöhler-kurve. Dette er i sammenhængen nyt og nødvendigt for at undgå overdreven sikkerhed med hensyn til stålkomponenter og ikke-konservatisme for glasfiberarmerede plastmaterialer. Den foreslåede metode giver betydelig reduktion i antallet af beregninger i dimensioneringsprocessen. Status for anvendelsen af modellen er, at den per august 2001 indgår i Dansk Standards standard for konstruktion af vindmøller, DS 472 (2001), samt at den er inkluderet i den tredje og sidste udgave af den internationale standard for vindmøller, IEC61400-1 (2005).

Også ekstrembelastninger under normal mølledrift i mølleskygge og effektiviteten af meget store mølleparker behandles.

Summary in English

Turbulence – in terms of standard deviation of wind speed fluctuations – and other flow characteristics are different in the interior of wind farms relative to the free flow and action must be taken to ensure sufficient structural sustainability of the wind turbines exposed to “wind farm flow”. The standard deviation of wind speed fluctuations is a known key parameter for both extreme- and fatigue loading, and it is argued and found to be justified that a model for change in turbulence intensity *alone* may account for increased fatigue loading in wind farms. Changes in scale of turbulence and horizontal flow-shear also influence the dynamic response and thus fatigue loading. However, these parameters are typically – negatively or positively – correlated with the standard deviation of wind speed fluctuations, which therefore can, if need be, represent these other variables. Thus, models for spatially averaged turbulence intensity inside the wind farm and direct-wake turbulence intensity are being devised and a method to combine the different load situations is proposed. The combination of the load cases implies a weighting method involving the slope of the considered material’s Wöhler curve. In the context, this is novel and necessary to avoid excessive safety for fatigue estimation of the structure’s steel components, and non-conservatism for fibreglass components. The proposed model offers significant reductions in computational efforts in the design process. The status for the implementation of the model is that it became part of the Danish standard for wind turbine design DS 472 (2001) in August 2001 and it is part of the corresponding international standard, IEC61400-1 (2005).

Also, extreme loading under normal operation for wake conditions and the efficiency of very large wind farms are discussed.

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A.1 Basic fatigue load concepts

A.2 Flow in the infinitely large wind farm

A.3 Momentum and energy balance in wake

Foreword

The report is to be considered as one independent thesis, in which use is made of previous work by the author – viz. the emphasised references in the reference list. Thus, while some results have previously been published, other parts appear in this report for the first time.

In particular three publications are central relative to the themes of the report. Thus, the model for effective turbulence was summarized in Frandsen and Thøgersen (2000). Herein, the background and validity of the method are dealt with in detail. The model for ambient turbulence within wind turbine clusters was reported in Frandsen and Madsen (2003). Likewise, the model for wind speed deficit in large wind farms is one of the central ideas of the report, Frandsen et al (2004).

To a large extent, the report is serving as documentation for the revision of the Danish standard on wind turbine design and safety DS 472 (2001) and the International Electrotechnical Commission's standard IEC61400-1 (2005), the aim being to compile evidence that the model for effective turbulence in its relative simplicity adequately accounts for increased fatigue loading in wind turbine clusters. The status by mid-2005 for implementation of the model is that it has become a non-normative amendment to DS 472 (2001) and IEC61400-1 (2005).

The author feels compelled to state that the efforts presented in the report are multi-disciplinary, covering areas like atmospheric boundary layer flow, wake-flow modelling, structural mechanics and materials' science. Embracing these disciplines made it necessary (at least for the author) to select and apply models that dedicated specialists may find rudimentary.

In developing the model, P. Hauge Madsen and C. Eriksson have been instrumental in their insistence on an applicable and easy-to-use form of the model.

The following colleagues have provided valuable comments to the report: N.J. Tarp-Johansen, R. Barthelmie, L. Kristensen and P. Hauge Madsen.

SEAS and Bonus Energy (now Siemens) have kindly made data available and Risø National Laboratory, the Danish Energy Agency and the EU Commission financed major parts of the work, on which the report is based.

Throughout the report the SI metric system is applied and/or assumed if nothing else is mentioned.

1 Introduction

Over a 30-year period, wind power technology has developed from being marginal to a significant contributor to the power supply, delivering by the end of 2004 approx. 20% of Danish electric energy. Over three decades, the energy production costs (DKK/kWh) have been reduced by a factor of three, bringing the technology close to competitiveness relative to conventional energy sources.

The contemporary electricity-generating wind turbine consists of the rotor with three (less frequently two) blades mounted on a hub, the main shaft, the nacelle that houses a gearbox, generator and auxiliary equipment, the tower, the control system and possibly a transformer. The machine may be operated at fixed or variable rotational speed. Limiting aerodynamic power to the capacity of the generator or optimizing power output may be done passively with stall control or actively by pitching the blades. The single-most descriptive parameter of a wind turbine is the *swept area* of the rotor, which signifies the possible kinetic energy capture. The swept area is the circular disc covered by the blades during their rotation. Though frequently being used as a short characterisation of a wind turbine, the capacity of the generator is secondary, being selected to match the size of the rotor and/or the operational strategy.

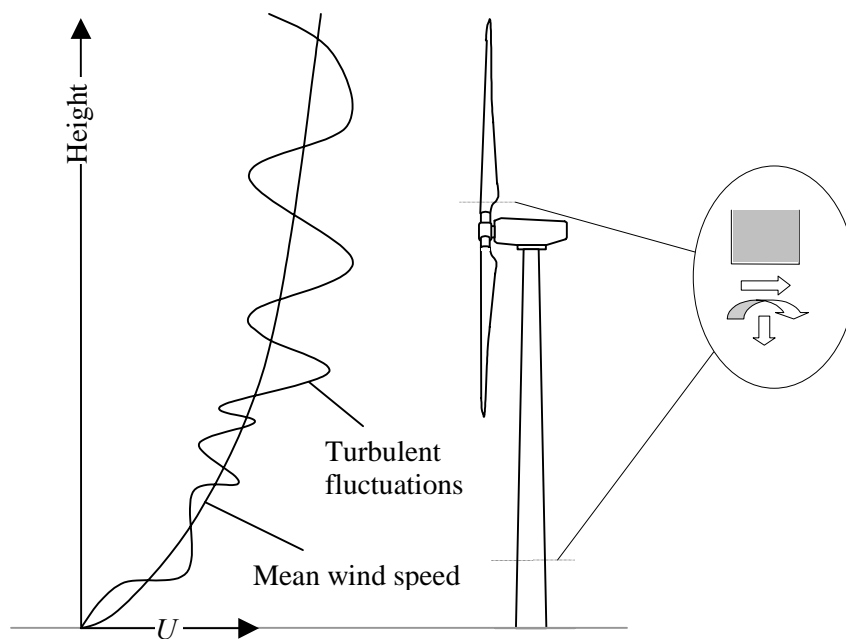


Figure 1.1 *Wind loading of a wind turbine structure. Wind speed is decomposed in its 10min mean and turbulent fluctuations around the mean. From design calculations, cross-sectional forces, deflections and material stress are determined.*

The mentioned reduction of cost of energy was achieved by refinement of the rotor aerodynamics, improvement of gearbox, generator and control system, and not least by optimisation of design against structural failure. When the wind turbine is parked with locked rotor, the loads and response calculations are similar to those of any civil engineering structure. The principal loading stems from the wind, and decomposing wind speed in a vertical mean wind speed profile and turbulent fluctuations around the mean facilitates response calculations, Figure 1.1.

When the machine is in operation, wind load is still the main contributor to structural loading, though also dynamic gravity loading of the rotating blades becomes important. Basically, wind forces on the tower are the same as for the non-operating wind turbine. However, during operation flow forces acting on the blades completely dominate. And while the wind turbine typically is shut down when the wind speed exceeds 25 m/s – far below extreme wind conditions – the blade tips on the operating wind turbine are continuously exposed to flow speeds in excess of the blade tip speed, 60-90 m/s. Thus, for the major part of its lifetime parts of the wind turbine rotor is exposed to severe flow speeds and ultimate loading of the structure may well happen during normal operation.

Designing the structure, both ultimate and fatigue loading must be considered. As it turns out, fatigue loading during normal operation frequently becomes decisive. As will become evident, the dynamic response that may result in fatigue failure is to a large extent governed by turbulent wind speed fluctuations.

When the development of a code of practise for design of wind turbines was initiated in the early 1980'ies, wind turbines were deemed “civil engineering structures” rather than the possibly more obvious “machines”. Following that choice, national and international wind turbine standards were created in the spirit of civil engineering traditions. In essence, the standards comprise of a number of *load cases*, each of which the structure must be able to withstand. A load case is a set of specific values of external conditions – i.e. mean wind speed, turbulence intensity and air density – and “states” of the wind turbine. The significance of the complex of load cases is that applying the load cases to the structure through design calculations, these will in aggregate result in (at least) the same ultimate and fatigue loading as the real-life loading over a chosen number of years.

In the context of designing a wind turbine structure, many load cases additional to those relevant for other civil engineering structures emerge. Such load cases include the load effect of turbulence generated by *operating* wind turbines, neighbouring the considered unit.

1.1 Need and purpose of work

Thus, the main purpose of the work presented was to conceive and justify a simple, yet not over-conservative model for flow conditions in wind turbine clusters – a model applicable for structural design against fatigue failure. The proposed model encompasses all physical effects of the wind farm on the airflow and offers a significant reduction in the required design computations.

The alternative to such a model is an order of magnitude more simulation runs with the aeroelastic computer codes used in contemporary wind turbine design. Without the model, simulations must be carried out for a large range of wind directions with no wake effects from neighbouring wind turbines to wind directions *with* wake effects. Also, since the distance to the neighbouring wind turbines varies – and thus the magnitude of the wake effects – separate simulations must be carried out to account for each individual wake.

The presented effort is mainly directed toward fatigue-inducing loads. However, there is a similar need for reduction of computer simulation runs in connection with extreme loading in the interior of wind farms. A rational approach to the derivation of the distribution of extremes, not conditioned on wind direction, is offered.

Somewhat off the main topic – structural loading – the report also addresses the potential problem that very large wind farms, as those being planned and built off the coasts of Denmark, may significantly affect the local wind climate, which in turn may result in disappointingly low energy production from the wind farms.

1.2 Specific background

No existing national or international standards had specific normative¹ or non-normative directives on how to deal with the irregular flow in the interior of wind farms in the context of fatigue loading of the wind turbines. The Danish standard on wind turbine design, DS 472 (1992), merely mentioned that wake effects should be taken into account and that – if simplified design rules for smaller machines (rotor diameter less than 25m) were applied – the distance between wind turbines in wind farms should be larger than 5 rotor diameters. The previous edition of the international standard IEC61400-1 (1999) limited its guidance to stating “Wake effects from neighbouring machines shall be considered for WTGS (wind turbine generator systems) operating in wind farms”. Though not being a standard as such, the Teknisk Grundlag (1992) does give specific directions on how to include wake effects when the Danish Approval Scheme for Wind Turbines is applied.

To deal with these deficiencies, numerous research efforts have addressed various parts of the problem, though loads and structural response have been investigated considerably less than measurement and modelling of the wake-airflow itself, Crespo et al (1999a). The wake-load modelling, which *has* been done primarily suggests extensive schemes of load cases to cover the real-life loads.

The Vindeby Wind Farm

One particular data set has played a central role in the analyses of this report, namely data from a large experiment set up at the Vindeby Wind Farm, see Figure 1.2. The wind farm was built to demonstrate the wind energy possibilities in the relatively shallow waters off the shores of Denmark. Thus, the wind farm was intended for gaining general operational experience and to compile data on the energy potential and structural loads offshore, including the impact of wake effects on structural loading.

The measurements at the offshore Vindeby Wind Farm – consisting of 11 450kW BONUS machines (3-bladed, stall controlled, rotor diameter 35m and hub height 38m above mean sea level) located 1.5 to 3 km off the coast of the island Lolland – were carried out over a stretch of years. The wind farm was commissioned and set into operation in September 1991. The 11 machines are arranged in two rows, with 6 in one row and 5 in the other. The orientation of the rows is 140° azimuth so as to minimize wake effects, the predominant wind direction being west-southwest. The distance between the turbines in each row is 300m (8.5 rotor diameters) and the distance between the two rows is likewise 300m. The water depth varies between 3 and 5m.

Two machines, 4W and 5E, were identically instrumented for structural measurements: flap- and edgewise bending moments on one blade, bending moment in tower base, active and reactive power (voltage and current), yaw position and operational status. Three 48m meteorological towers were erected. One tower was located on land to provide information on the change of wind characteristic when the wind was coming from land, one (SMW) was placed to the west of the wind turbines, serving basically as a reference mast, but in certain wind directions it measured double-wake conditions, and one (SMS) was placed at an imaginary wind turbine position in the western row to measure the flow in multiple-wake situations. All meteorological towers were equipped with cup anemometers in at least 5 levels, and wind direction and temperature sensors. Also, two 3-D sonic anemometers were employed. At the base of one of the sea-bottom-based towers wave heights were measured.

Sensor signals from the offshore meteorological towers were fed through multi-core cables to one of the instrumented wind turbines from where they were relayed – together with sensor signals from the wind turbines – through an optical fibre cable to the central data storage and processing computer, which was placed in a cabin at the

¹ Meaning “shall be used”.

base of the onshore meteorological tower. Structural and meteorological data were sampled continuously at 25 Hz and stored as 30-minute records.

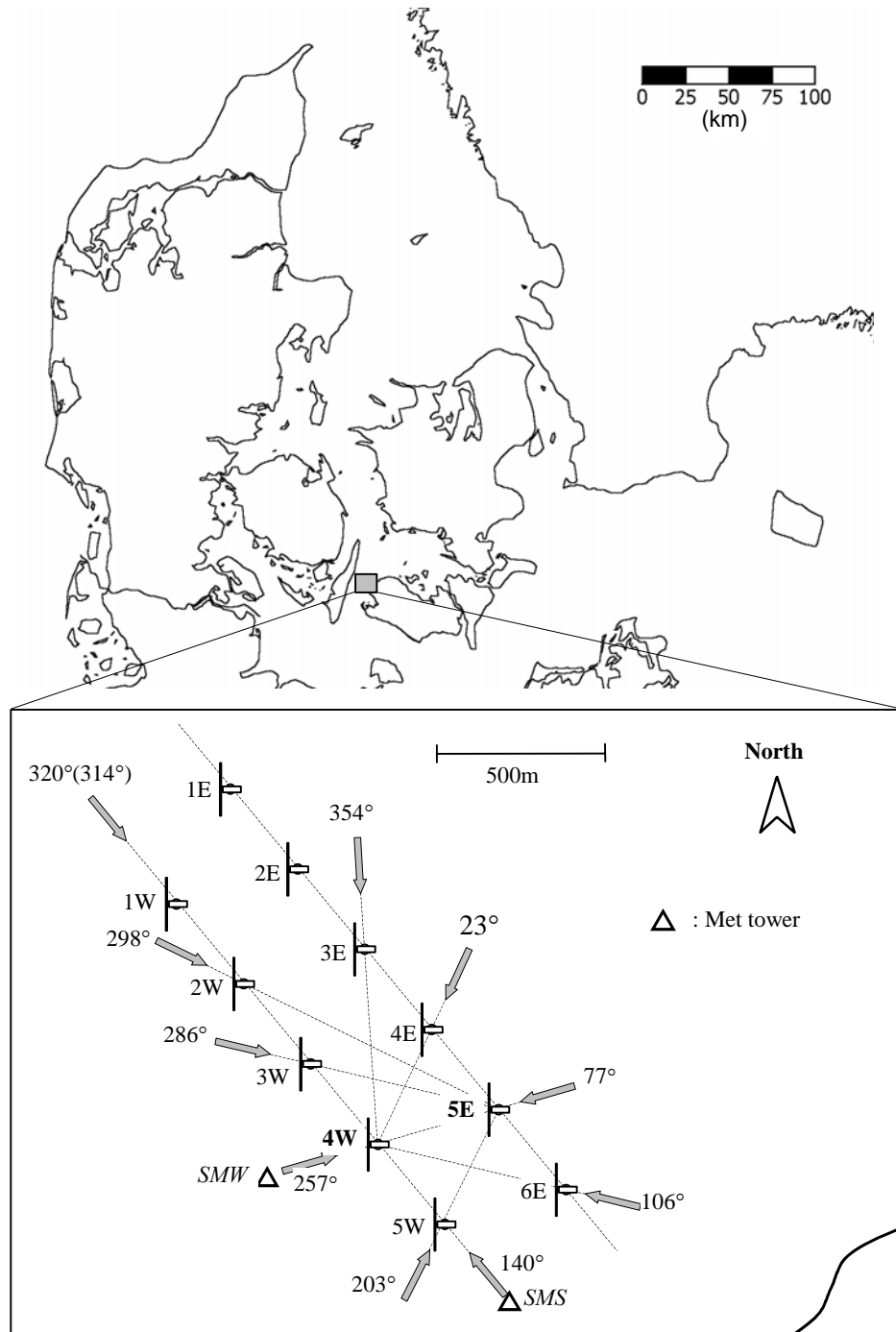


Figure 1.2 Location and layout of the Vindeby offshore wind farm. The distances between neighbour wind turbines are 300m and 330m. Arrows are indicating wind directions with wake conditions on one or both the instrumented units, 4W and 5E.

Statistics in terms of mean, standard deviation, maximum and minimum of all signals and for the structural measurements also fatigue damage "equivalent load" were computed on-line and stored. And finally, each 30-minute record was categorized (binned) according to wind speed and wind direction and stored until an adequate number of time series was accumulated in each bin.

Meteorological data have been sampled from all three meteorological towers since November 1993, and data from the two instrumented wind turbines since April 1994. The measurements were continued to the full extent for approx. 4 years.

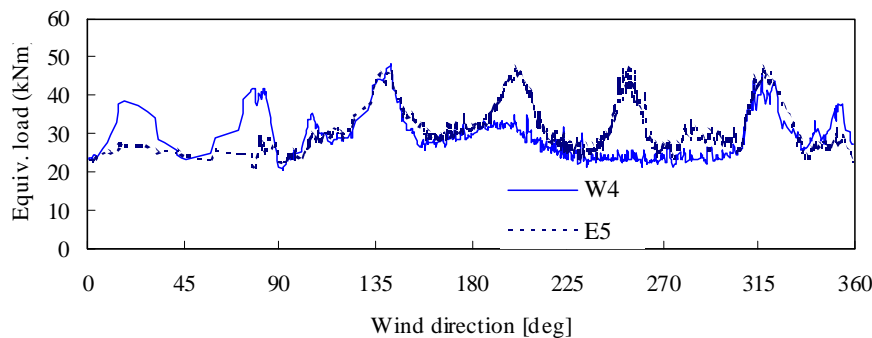


Figure 1.3 Equivalent load (flap-wise bending) for two wind turbine units, 4W and 5E, as function of wind direction, in the Vindeby wind farm, $8 < U < 9$ m/s. The curves are “running averages” of the $\frac{1}{2}$ -hour values of the wind direction-ordered equivalent loads.

Approx. 13,000 half-hour complete data series – free of gross errors and with all wind turbines in operation – were selected for the analyses in this report.

The measurement system is described in detail in Barthelmie et al (1994).

The equivalent load is the constant load amplitude at a fixed frequency that yields the same fatigue life consumption as the actual irregular load variation. The great advantage of the equivalent load quantity is that for unchanged exterior geometry of the considered structure, it immediately signifies – similar to extreme loads – material consumption. E.g. if the equivalent load is up a factor of 2, then the wall thickness of the wind turbine tower must be doubled.

Comparing flap-wise bending moments of units 4W and 5E, Figure 1.3 and Figure 1.2, it is seen that the equivalent load strongly increases under wake conditions, up to 100%. Further, it is seen that the increase in equivalent load – relative to non-wake situations – for single-wake (e.g. wind directions 140° and 257°) is about the same as for multiple-wake situations (wind direction 320°).

Also the standard deviation of wind speed fluctuations is increased in the wake of a wind turbine and it is an obvious deduction to link turbulence level and fatigue loading.

Further, the shapes of the equivalent load wake-profiles seem to approximate bell/normal distribution-shapes. Theoretical considerations and experimental evidence, e.g. Tennekes and Lumley (1972) and Engelund (1968), do suggest a bell-shape for the mean speed deficit across the axis-symmetric wake, some distance downwind of the wake-generating obstacle. As for standard deviation of wind speed fluctuations, similar basic modelling leads to zero-turbulence in the centre-wake where the presumed generation source – flow shear – is zero. Wind tunnel tests *do* indicate a slight decrease in wind speed fluctuations in the centre-wake, but basically measurements point a bell-shape cross-wake profile of standard deviation of wind speed fluctuations a short distance downwind of the obstacle, Crespo et al (1999a).

The starting point

In Denmark, Teknisk Grundlag (1992)² presented a model based on the assumption that fatigue loading under both wake and non-wake conditions is proportional to the turbulence intensity, defined as the standard deviation of wind speed fluctuations divided by the mean wind speed,

$$I = \frac{\sigma_u}{U}. \quad (1.1)$$

The recommendation puts forward a model where the design turbulence intensity is composed of the ambient (free-flow) turbulence intensity, I_0 , and the added turbulence intensity caused by the neighbouring wind turbines, I_{add} . The maximum wake turbulence intensity, I_T , and the design turbulence intensity, I_{eff} , are then calculated as

$$I_T = \sqrt{I_0^2 + I_{add}^2} \quad \text{and} \quad I_{eff} = \sqrt{I_0^2 + c^2 I_{add}^2}, \quad (1.2)$$

where c is a weighting-factor less than unity.

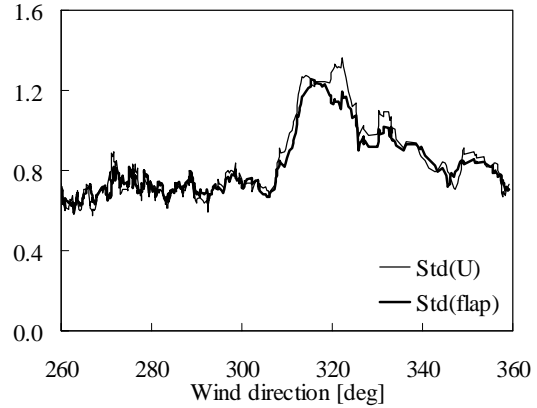


Figure 1.4 *Vindeby data, running average of data sorted after wind direction: Standard deviation of wind speed measured at hub height in SMS and flapwise blade bending moment of wind turbine 4W. The bending moment is scaled to fit at free-flow conditions at wind directions 260-300°. Scale of ordinate is arbitrary.*

Hitherto, it has not been made probable that the approach is reasonable: that it ensures that the wind farm flow conditions in general is represented well by altering only turbulence intensity. Nor has it been satisfactorily demonstrated that the concrete formulation, Eq. (1.2), is adequate.

That the idea as such could be viable is illustrated in Figure 1.3 and in further details in Figure 1.4. The standard deviation of wind speed fluctuations measured at hub height at the met mast SMS (quintuple wake) should – in a statistical sense – be equal to the same quantity experienced at the centre of the hub of wind turbine unit 4W (triple wake).

Whereas the measurement of turbulence is a point-measurement, the rotating blade effectively performs spatial averaging of the turbulence. Nevertheless, the two quantities coincide in remarkable details.

1.3 Approach

Herein, a model for "effective" or design turbulence intensity is devised. The model aims specifically at fatigue loading of operating wind turbines in wind turbine clusters. The model integrates load situations with free-flow turbulence intensity and load situations under wake conditions to give the effective turbulence intensity. When replacing the usual ambient turbulence intensity for stand-alone wind tur-

² In English: "Technical Design Basis". The document forms part of the Approval Scheme for Wind Turbines. The approval scheme is mandatory and it requires relevant standards to be adhered to. For issues not covered by standards, separate recommendations were prepared. The Approval scheme will probably yield to the international standards (IEC) emerging these years.

bines with the effective turbulence, no other flow variables need be altered to account for the increased loading in wind turbine clusters.

Thus, the model is based on the assumption that an alteration of the ambient turbulence intensity alone will be sufficient to account for added dynamic loading in wind turbine clusters. And further, that a wake-induced change of other flow variables with a load effect can be taken into account by an “extra” adjustment of the design turbulence intensity.

Applying the model when design calculations are made, the computations should be carried out as under free-flow conditions, except for the change in design turbulence intensity. For each load case, the “effective” turbulence intensity should be determined. Other flow variables such as flow shear and scale of turbulence are to be considered unchanged relative to ambient flow conditions.

For each load case/wind speed, the method’s effective turbulence intensity should be applied in the computer codes used for structural design and in turn the computed response time series be used for derivation of response statistics. Thus, the increased response shall not be determined by simply up-scaling response corresponding to the increased, effective turbulence intensity. Actually, in justification of the model it is argued that, conditioned on mean wind speed, fatigue loading is proportional to σ_u . However, for certain load situations (e.g. at wind speeds where active or passive power regulation is applied) simple up-scaling of response may be inadequate and shortcuts in design computations should not be taken.

Other *special* load cases – such as negative vertical shear of the air flow in complex terrain or a yaw misalignment of the rotor relative to the wind direction – shall be treated as “usual”, i.e. the model does not compensate for or replace non-wind farm specific load calculations.

The effective turbulence intensity is not to be understood as a physical quantity but a design quantity that will result in the same fatigue loading as the actual flow conditions.

The model was included in the Danish code for wind turbine design, DS 472 (1992), through an amendment, DS 472 (2001). Since then, the model has been applied by industry and consulting engineers and the feedback from these has been positive. The concrete formulation as laid forward in the edition 3 of the international standard IEC61400-1 (2005) is presented in Section 8.

1.4 Novelty of the work presented

The primary purpose of this report is the justification of a model that accounts for increased fatigue loading on a wind turbine due to the wake effects of neighbouring units and the increased ambient turbulence intensity inside the wind farm. In addition, extreme loading in wind farms is considered in some detail. Being off the main topic of the report, the feedback of very large wind farms to the atmospheric flow and subsequently the efficiency of large wind farms are also discussed.

The following components of the report are novel or have been presented previously as such by the author:

- Justification that standard deviation of wind speed fluctuations is the key driver for fatigue loading under both free-flow conditions and under wake conditions, Section 4.
- Model for the apparent roughness of large wind farms, including decrease in wind speed, Frandsen (1991), Frandsen (1992), Frandsen and Thøgersen (1999), Subsection 2.4 and Appendix A.2.

- Model for horizontally averaged standard deviation of wind speed fluctuations within large wind farm, Frandsen and Madsen (2003), Subsection 2.4 and Appendix A.2.
- Scheme for summation of fatigue life consumption for free-flow and wake conditions, Section 5.
- Model for effective (fatigue) turbulence in wind farms that encompasses the wake effects of all neighbouring wind turbines in one expression for the design turbulence, DS 472 (2001), Frandsen and Thøgersen (1999) and Sections 5 and 8.
- Distribution of extremes in wind farms, not conditioned on whether there is direct wake or non-wake turbulence, Section 6.
- Alternative model for the growth of internal boundary layer and decrease/increase of wind speed after a terrain roughness change, Subsection 9.1.
- Integrated model for the efficiency of very large wind farms, Frandsen et al (2004) and Subsection 9.2
- Various considerations regarding wake flow speed and turbulence, Subsection 3.1 and Appendix A.3

1.5 Structure of presentation

The presentation is structured as follows:

- In Section 2, the impact in terms of turbulence of infinitely large wind farms on the local wind climate is discussed and a model for the horizontally averaged turbulence intensity inside large wind farms is presented.
- In Section 3, the increased turbulence intensity in the immediate wake of neighbouring wind turbines is discussed and modelled.
- In Section 4, justification of the method for effective turbulence is given.
- In Section 5, it is demonstrated how a range of load cases should be combined to one load case by means of the model for effective turbulence.
- In Section 6, similar considerations regarding combination of extreme distributions for wake and non-wake situations are discussed.
- In Section 7, the model is verified against data – alternative to those from Vindeby applied in Section 4 – and compared with the model of Teknisk Grundlag (1992). Also the uncertainties related to the model are discussed.
- In Section 8, the formulation for the model for effective turbulence as adopted by IEC61400-1 (2005) is given.
- In Section 9, modelling of roughness change is commented on and an integrated analytical model for the mean wind speed deficit in – and thus the efficiency of – large wind farms are presented.
- In Section 10, concluding comments to the modelling are made.

2 Ambient flow and average wind farm flow

Firstly in this section, fundamental properties of the neutrally stratified planetary boundary layer are outlined and common engineering practice for standard deviation of turbulent wind speed fluctuations in the free-flow is described. And secondly, the *spatially averaged* level of standard deviation of wind speed fluctuations in “large” wind farms is evaluated.

2.1 Vertical shear in the free flow

For the thermally neutrally stratified planetary boundary layer in plain horizontal terrain with height-independent shear stress and with the scale of turbulence proportional to height, it is found that the vertical wind speed profile is well modelled as being logarithmic,

$$\frac{U(z)}{u_*} = \frac{1}{\kappa} \ln \frac{z}{z_0}, \quad (2.1)$$

where z is height above ground, $U(z)$ in mean wind speed as function of z and the constants u_* and z_0 are denominated friction velocity and the terrain surface roughness length, respectively. κ is von Karman’s constant which is approximately equal to 0.4. Experimentally, the roughness length is found to be of the order $\frac{1}{10}$ of the typical size of elements/

obstacles on the ground that retard the flow.

For the neutrally stratified atmospheric boundary layer up to, say, 100m above ground the expression Eq. (2.1) fits numerous experimental data excellently.

For later use, a measure of the vertical flow shear is defined as the difference in mean wind speed at the highest and the lowest wind turbine blade tip positions, respectively:

$$\tau = U\left(h_H + \frac{1}{2}D_0\right) - U\left(h_H - \frac{1}{2}D_0\right), \quad (2.2)$$

where h_H is wind turbine hub height and D_0 is the diameter of the wind turbine’s rotor. For the neutrally stratified boundary layer, τ may be estimated by means of Eq. (2.1):

$$\begin{aligned} \tau &= \frac{u_*}{0.4} \ln\left(\frac{h_H + \frac{1}{2}D_0}{z_0}\right) - \frac{u_*}{0.4} \ln\left(\frac{h_H - \frac{1}{2}D_0}{z_0}\right) \\ &\approx \sigma_u \ln\left(\frac{h_H + \frac{1}{2}D_0}{h_H - \frac{1}{2}D_0}\right) \end{aligned} \quad (2.3)$$

where σ_u is the along-wind standard deviation of turbulent wind speed fluctuations. It has been utilised, see below, that $\sigma_u \approx 2.5 \cdot u_*$. Typically, the rotor diameter is approximately equal to hub height, $D_0 \approx h_H$, and thus from Eq. (2.3) it is found that $\tau \approx \sigma_u$, i.e. in the free flow there is a difference in mean wind speed from bot-

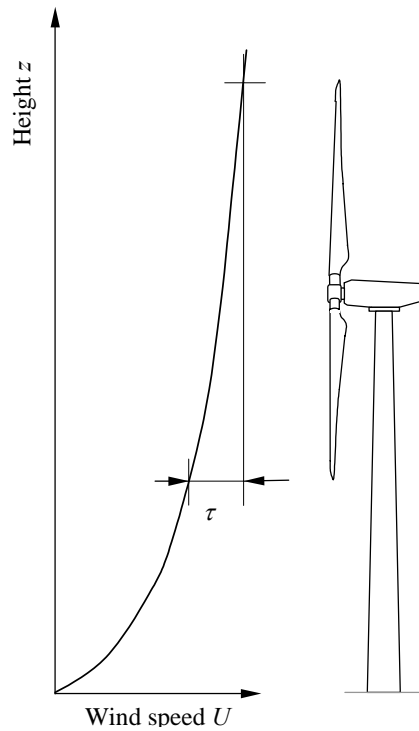


Figure 2.1 Shear over rotor. Wind from the left.

tom to top position of the blades, which is of the same size as the typical turbulent fluctuations in wind speed.

Especially for lower wind speeds there is significant variation in atmospheric stratification, which affects both vertical mean shear and fluctuations in wind speed. And for variation in stratification there is – for fixed mean wind speed and observation height – a negative correlation between vertical shear and turbulent wind speed fluctuations: low σ_u corresponds to large shear and visa versa.

2.2 Ambient turbulence in the free-flow

In the neutrally stratified atmosphere, the standard deviation of wind speed fluctuation σ_u is experimentally found to be proportional to the friction velocity u_* and it has been demonstrated that $\sigma_u \approx 2.5 \cdot u_*$. Together with Eq. (2.1) this provides the following expression for standard deviation of wind speed fluctuations:

$$\frac{U}{u_*} = \frac{1}{\kappa} \ln \frac{z}{z_0} \Rightarrow u_* = \frac{\kappa U}{\ln(z/z_0)} \Rightarrow \sigma_u \approx \frac{U}{\ln(z/z_0)}. \quad (2.4)$$

The expression, Eq. (2.4), was adapted by e.g the Danish standard for loading of civil engineering structures, DS 410 (1998), and the standard for design of wind turbine structures, DS 472 (2001). Like the friction velocity, the standard deviation of turbulent wind speed fluctuations may deviate substantially from Eq. (2.4), under stable and unstable stratification of the atmosphere. However, for extreme loading, under extreme wind conditions, it is of little consequence whether non-neutral stratification is taken into account.

In the previous edition of the international standard, IEC61400-1 (1999), the design turbulence, $\sigma_{u,IEC}$, is given by the following expression:

$$\sigma_{u,IEC} = I_{15}(15 + a \cdot U)/(a + 1). \quad (2.5)$$

Here, I_{15} is a characteristic value of hub-height turbulence intensity at the wind speed 15 m/s, a is a slope parameter, and U is the mean wind speed. The expression takes into account the frequently encountered over-representation of unstable atmospheric conditions at lower wind speeds. That IEC61400-1 (1999) bothers to include unstable weather conditions is because fatigue loads at lower wind speeds do matter.

The IEC61400-1 (1999) operates with two turbulence levels, where for “low” turbulence $(I_{15};a) = (0.16;3)$ and $(I_{15};a) = (0.18;2)$ for “high” turbulence. The expression, Eq. (2.5), is the expectation value of the standard deviation of turbulent wind speed fluctuations, as experienced/measured in nature, plus one standard deviation of the same quantity. The standard deviation, denoted $\Delta\sigma_u$, of σ_u is specified as

$$\Delta\sigma_u = 2I_{15}. \quad (2.6)$$

A little unfortunate, the constant “2” has the dimension m/s. Thus, the coefficient of variation for the standard deviation of turbulent wind speed fluctuations is

$$\delta = \frac{\Delta\sigma_u}{\sigma_{u,IEC} - \Delta\sigma_u}. \quad (2.7)$$

The quantity is plotted in Figure 2.2, for the “low turbulence” case. When wind speed increases from 10 to 20 m/s, the coefficient of variation decreases from approx. 20% to 10%.

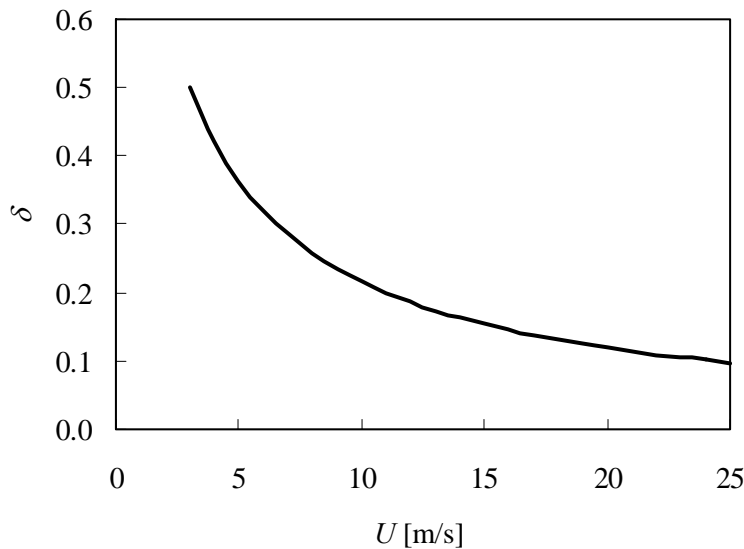


Figure 2.2 Coefficient of variation for standard deviation of turbulent wind speed fluctuations as function of wind speed, following IEC61400-1 (1999).

Thus,

$$\sigma_{u,IEC} = \bar{\sigma}_u + \Delta\sigma_u . \quad (2.8)$$

Assuming that the standard deviation of turbulent wind speed fluctuations is normal distributed, the above value constitutes a percentile of approximately 80%, i.e. for a given mean wind speed, there is a probability of 80% that σ_u is less than $\sigma_{u,IEC}$.

2.3 Scale(s) of turbulence

While in principle the scales of turbulence are easily determined from time series of the three wind speed components, these scales are usually not recorded and stored as other statistics such as mean and variance of speed fluctuations when performing measurements. The scale is determined from the integral of the auto-correlation function, the maximum of the power spectrum or possibly the up-crossing frequency of the mean of wind speed. All methods are sensitive to the assumption of stationarity of the time series. Alternatively expressed, wind speed fluctuations may have significant energy at low spectral frequencies not captured by analysis of 10min series. Therefore, the uncertainty related to determination of the scale(s) of turbulence is large. Disregarding these difficulties, the PDF of turbulence length-scale³ has been experimentally estimated, Petersen et al (1998). The result is shown in Figure 2.3. For the considered data, the scale of the along-wind component of wind velocity, measured at height 48m, typically varies between 100 and 2000m, with a maximum at $L \approx 500$ m. The variation is due to varying atmospheric stability, but also the mentioned uncertainties related to the lack of stationarity. While in the free flow, the average of length-scale is expected to be constant at a fixed height, the characteristic frequency is – by means of Taylor’s hypothesis on “frozen turbulence” – linked to the mean wind speed and length-scale by

³ The length-scale, and not the time-scale, is usually preferred since it presumably – for a given height - is independent of mean wind speed.

$$f \propto \frac{U}{L}. \quad (2.9)$$

Thus, the frequency scale varies by a factor of two when the wind speed varies between 10m/s and 20 m/s, which are wind speeds most relevant to fatigue loading of wind turbines. This mean wind speed related variation in the frequency scale is small compared to the orders of magnitude in variation of the observed turbulence scale described above.

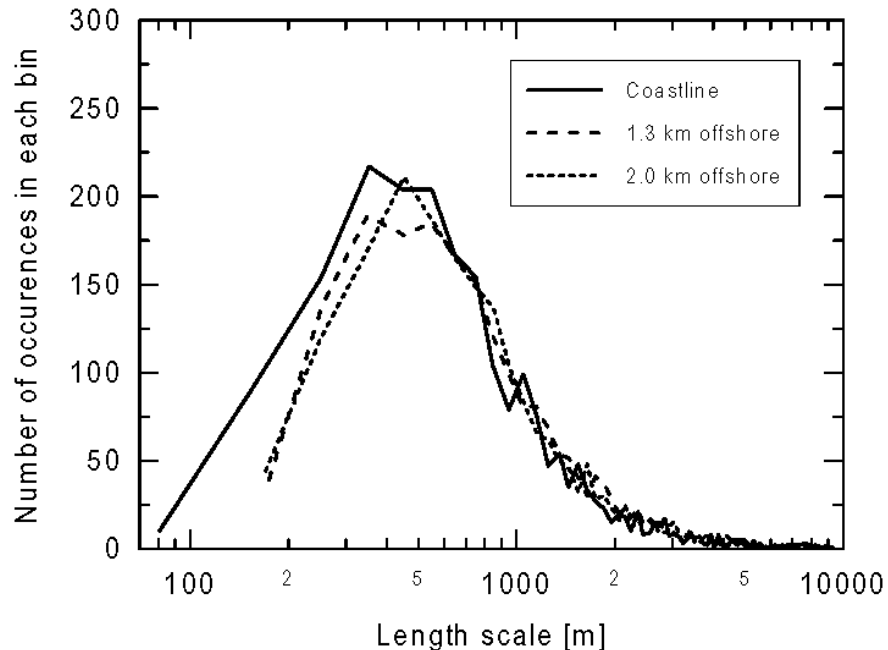


Figure 2.3 Measured PDF of length-scale of turbulence. From Petersen et al (1998).

For *wake conditions* – i.e. when the flow is affected by a nearby wind turbine – the scale of turbulence has been evaluated relative to free-flow conditions, Højstrup (1990), Crespo and Hernandez (1996) and Højstrup (1999), employing measurements and computer simulations. Both measurements and simulations showed that in the upper wake (above hub height) the scale is approximately unchanged and in the lower wake the length-scale is reduced to about half the free-flow scale. From wind tunnel measurement, Tindal et al (1993) report length-scales in the wake from 1/2 to 1/5 times the free-flow scale, depending on distance from the wind turbine. For yet another set of field measurements of length-scale 4 rotor diameters downwind, Verheij et al (1993) report reductions to about half of free-stream conditions.

In all, the apparent variability of the free-flow scale of turbulence is large compared to the reported change of scale of turbulence specifically imposed by the wakes.

Højstrup (1999) also investigated lateral and vertical coherence (normalized cross-spectral density), again finding that deviations from the free-flow case are minor.

Apart from the scale of turbulence, the general observation is that the power spectral characteristics in terms of the shapes of spectra in the wake are fairly well represented by the models applied for the free-flow, with the exception that wake turbulence tends to be more isotropic. The detailed structure of vortices generated by the wind turbine's rotor is not detectable a few rotor diameters downwind.

2.4 Ambient Turbulence within the Wind Farm

For large wind farms it is necessary to re-consider the level of “ambient” turbulence intensity. One or two rows of wind turbines away from the edge of the cluster, when there is not a wind turbine immediately upwind, turbulence may be expected to be identical to the free-flow turbulence. Further into the wind farm – the distance being depending on the density of wind turbines – no wind direction offers a flow that is unaffected by wind turbine wakes. Thus, at any given point of observation, the standard deviation of turbulent wind speed fluctuations may be described as composed of fluctuations not caused by a specific turbine, and fluctuations caused by a well-defined wake, generated by a wind turbine immediately upwind. The well-defined wake component is discussed in the Section 3, whereas the average component, here termed “ambient wind farm turbulence”, was described in Frandsen and Madsen (2003) and is treated below.

In the wind farm, the mean wind speed will be reduced relative to the free-flow wind speed. The first efforts to estimate the wind speed reduction in large clusters of wind turbines were made by Templin (1974), Newman (1977) and others, see Bossanyi et al (1980), Frandsen (1992), and Emeis and Frandsen (1993).

Adopting the view that the wind turbines can be considered roughness elements, also the *general* level of turbulence intensity will increase, i.e there will be higher turbulence not only under distinct wake conditions. In order to estimate the general decrease in mean wind speed and increase in turbulence intensity, an infinitely large wind farm is considered. Applying a simplified version of the geostrophic drag law, Jensen (1978) and the further assumptions given in Appendix A.2, the horizontally averaged, vertical wind profile down to hub height in the wind farm may be described as

$$\frac{U}{u_{*0}} = \frac{1}{\kappa} \ln \left(\frac{z}{z_{00}} \right), \quad (2.10)$$

where U and z are mean wind speed and height above ground, respectively, and the apparent, combined roughness of the ground and the wind turbines is

$$z_{00} = h_H \cdot \exp \left(- \frac{\kappa}{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}} \right), \quad c_t = \frac{\pi C_T}{8s_r s_f}, \quad (2.11)$$

where h_H is hub height, z_0 is the roughness length of the terrain surface, C_T is the wind speed dependent thrust coefficient of the wind turbines, and s_r and s_f are distances between the units in the rows and the separation between the rows, normalised with the rotor diameter⁴. The height independent above-wind-farm friction velocity becomes

$$u_{*0} = \frac{\kappa G}{\ln \left(\frac{G}{f' h_H} \right) + \frac{\kappa}{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}}, \quad (2.12)$$

and the mean wind speed at hub height

$$U_h = \frac{G}{1 + \ln \left(\frac{G}{f' h_H} \right) \frac{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}{\kappa}}, \quad (2.13)$$

⁴ If the wind turbine units are located in an irregular way, then s and s_f should be taken as averages in the wind farm.

where G is the geostrophic wind speed and $f' \approx 1.2 \cdot 10^{-4} \cdot \exp(4) = 6.5 \cdot 10^{-3}$ at latitude 55° .

As for the free flow, σ_u is assumed to be proportional to the friction velocity. In the free flow – at height h_H – turbulent fluctuations σ_0 and turbulence intensity I_0 are:

$$\sigma_0 \approx \frac{U_0}{\ln(h_H / z_0)} = \frac{u_*}{\kappa}, \quad I_0 = \frac{\sigma_0}{U_0}. \quad (2.14)$$

Similarly, turbulence over the wind farm is estimated as

$$\sigma_{T,wf} = \frac{u_{*0}}{\kappa}. \quad (2.15)$$

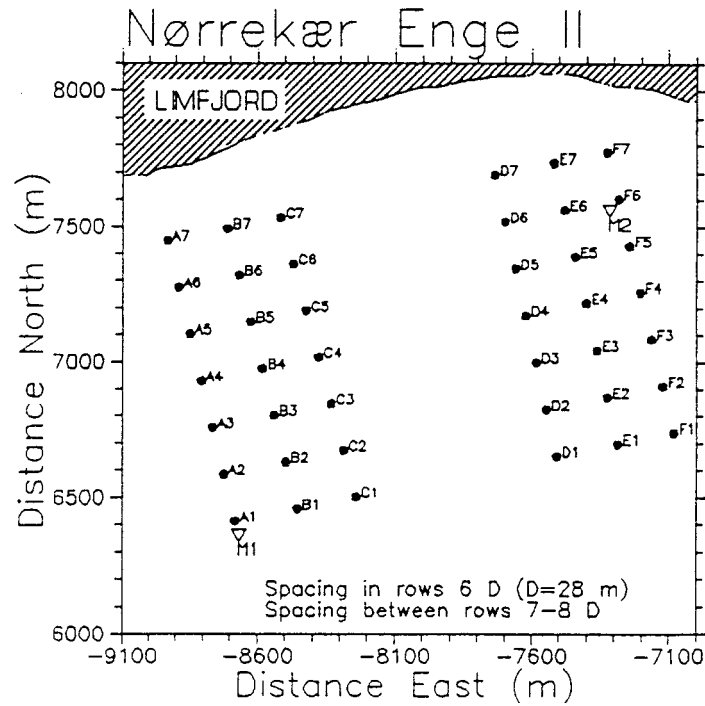


Figure 2.4 Layout of Nørrekær Enge II wind farm. 42 x 300kW, rotor diameter $D=28m$.

This expression may more generally be assumed valid some distance above the wind turbine rotors. Straining the physics, it is assumed that the expression is valid all the way down to the top position of the blades. For practical reasons, the turbulence intensity in the wind farm is defined referring to free-flow hub height wind speed, U_0 :

$$I_{T,wf} = \frac{\sigma_{T,wf}}{U_0}. \quad (2.16)$$

Experimental evidence is offered to support the model for the above-wind-farm standard deviation of turbulent wind speed fluctuations, Frandsen and Christensen (1994b). Flow measurements were carried out in the Nørrekær Enge II Wind Farm, Figure 2.4, which consists of two groups of 3 x 7 (300 kW) Nordtank units. One met mast is placed at the southwest edge of the wind farm and for southwesterly winds the free-flow characteristics are measured here. Another met mast is placed in the interior of the wind farm, in the northeasterly corner.

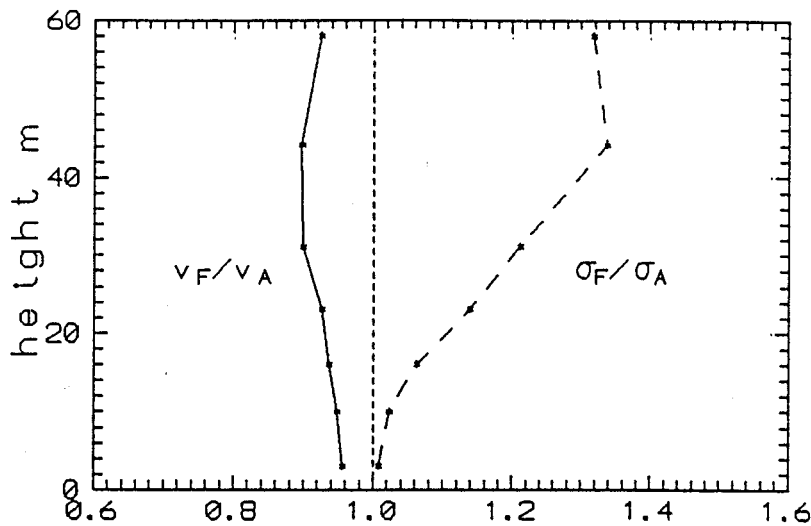


Figure 2.5 Ratios of wind speed (v_F/v_A) and standard deviation of turbulent wind speed fluctuations (σ_F/σ_A) profiles inside and outside the wind farm. Ambient wind speed between 8 and 9 m/s. The indices “A” and “F” refer to met masts M1 and M2, respectively (see Figure 2.4). From Frandsen and Christensen (1994b).

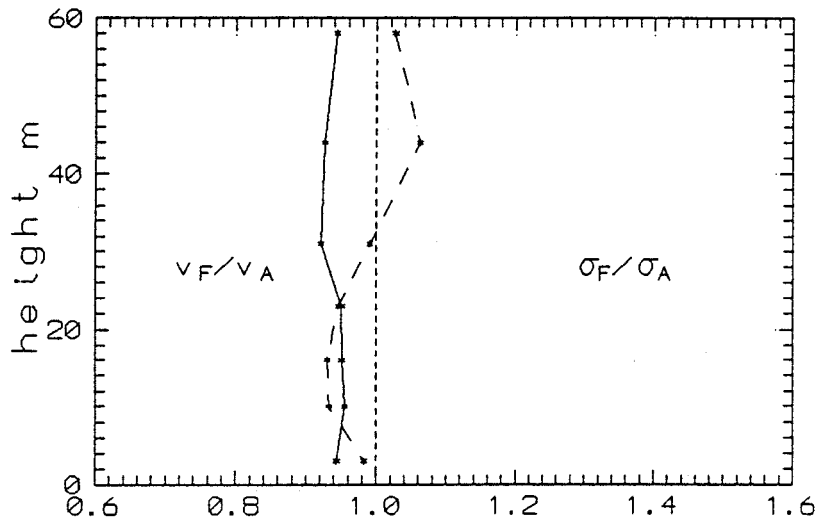


Figure 2.6 Ratios of wind speed (v_F/v_A) and standard deviation of turbulent wind speed fluctuations (σ_A/σ_F) vertical profiles inside and outside the wind farm. Ambient wind speed between 12 and 14 m/s. From Frandsen and Christensen (1994b).

Vertical profiles of mean wind speed and standard deviation of turbulent wind speed fluctuations were taken at both towers and averaged over a 30°-wind direction sector of winds from the southwest.

Figure 2.5 and Figure 2.6 show the ratio between the profiles for $8 < U < 9$ m/s and $12 < U < 14$ m/s, respectively. For the low wind speed range (high C_T values) there is an increase in the standard deviation of turbulent wind speed fluctuations above hub height of approx. 35% and near the ground σ_u is unchanged. For the higher wind speed range the increase in σ_u above hub height is approx. 10% and below “rotor height” there is an apparent decrease in σ_u of about the same magnitude. This decrease below the rotor is actually predicted by the presented model (by solving also for the friction velocity under hub height, Appendix A.2). However, in the following the standard deviation of turbulent wind speed fluctuations below hub height

is, conservatively with respect to the method for effective turbulence, assumed unchanged relative to free-flow conditions.

Figure 2.7 shows measured and modelled ratios of wind speed and standard deviation of turbulent wind speed fluctuations inside and outside the wind farm at hub height and at height 58m, respectively. The wind speed deficit at hub height is over-predicted by factor of two, which could indicate that in terms of mean wind speed, the wind farm is not “large” as assumed in the model presented above.

For σ_u above rotor height, the model deviates little from the data, indicating that in terms of standard deviation of wind speed fluctuations the Nørrekær Enge II wind farm is “large”.

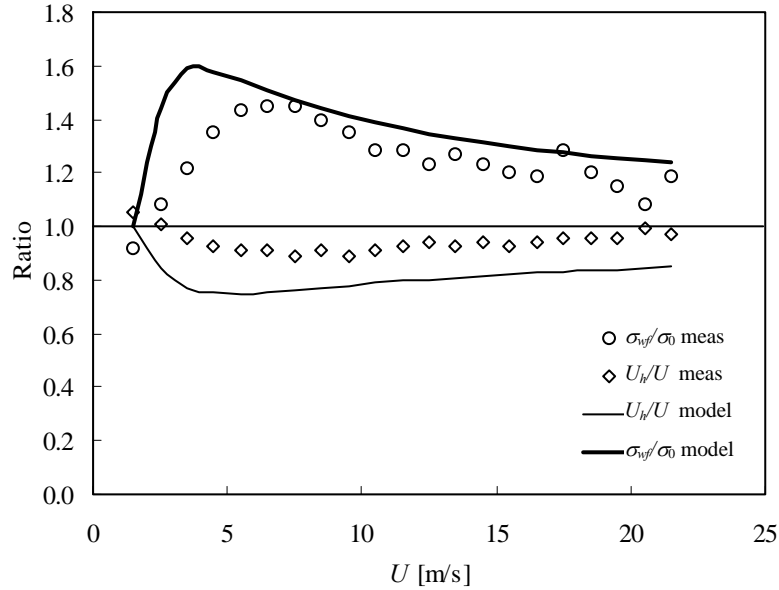


Figure 2.7 Ratios of wind speed at hub height (31m) and turbulence (58m) inside and outside the wind farm, as function of wind speed. The full lines are the model predictions.

The wind farm “ambient” turbulence may be decomposed in a component from terrain surface roughness and a component stemming from the presence of the wind turbines:

$$\sigma_{T,wf}^2 = \sigma_0^2 + \sigma_{addwf}^2, \quad I_{T,wf}^2 = I_0^2 + I_{addwf}^2 \Rightarrow I_{addwf} = \sqrt{I_{T,wf}^2 - I_0^2}, \quad (2.17)$$

where turbulence intensities are derived by dividing by the free-flow hub height wind speed. Eqs. (2.12) and (2.13) are fairly complex and a simplification is useful. In Appendix A.2, an expression for the value of I_{addwf} for large effect of the wind turbines relative to the surface roughness is derived. Although I_{addwf} is also a function of geostrophic wind speed, terrain roughness and hub height, it is fairly insensitive to changes of these parameters and the following approximation may be adopted:

$$I_{addwf} \approx \frac{a\sqrt{C_T}}{b\sqrt{C_T} + \sqrt{s_f s_r}} = \frac{0.36}{1 + 0.2\sqrt{s_f s_r} / C_T}, \quad \text{with } a = 1.8 \text{ and } b = 5. \quad (2.18)$$

The approximation is tested in Figure 2.8. For other values of G , z_0 and h , the deviation is larger, up to approx. 5% in relative terms for relevant parameter values. It should be noted that the turbulence intensity in general is a function of height of

observation, i.e. it will vary over the rotor. Also this has been neglected in recognition of the general uncertainty of the modelling effort and to arrive at the simple expression, Eq. (2.18).

Above and in Appendix A.2 it was assumed that the spatially averaged shear force/stress from the wind turbines acts on the flow at hub height. Actually, the shear imposed by the wind turbines is vertically distributed from the top to the bottom position of the blade tips. Therefore – and with support in the data presented in Figure 2.5 and Figure 2.6 – σ_u is assumed constant above the blade top position and constant the below blade bottom position. It is further assumed that σ_u varies linearly between these constant levels over the rotor layer. Thus, the average of the above-wind-farm turbulence intensity and the (chosen) below-rotor turbulence level, I_0 , is applied as “wind farm ambient turbulence intensity”:

$$I_0^* = \frac{1}{2} (\sqrt{I_{addwvf}^2 + I_0^2} + I_0). \quad (2.19)$$

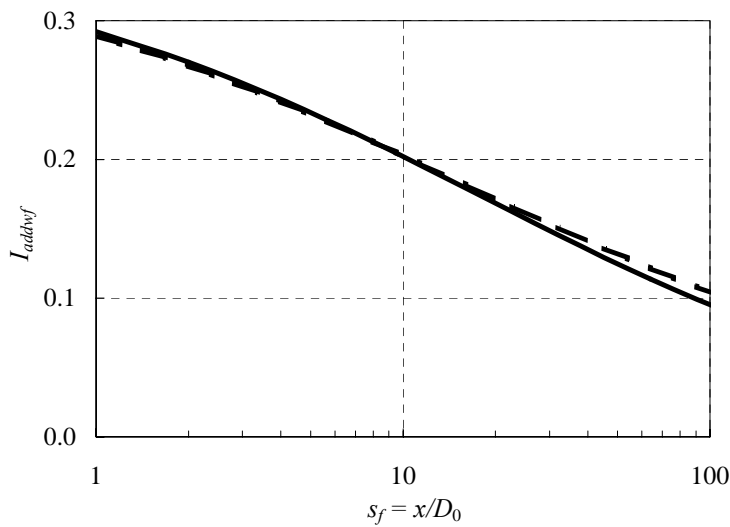


Figure 2.8 Horizontal average of added wind farm turbulence, as function of separation between rows. The solid line is from the basic equations, the broken line corresponds to the proposed approximation ($s_r=1.5$, $C_T=1$, $h=50m$, $z_0=0.01m$, $G=15m/s$).

Following its derivation, the expression (2.19) must be considered an average of direct-wake and in-between-wakes contributions and since the expression should represent ambient wind farm turbulence, it tends to be conservative. On the other hand, while the expression refers to hub height the wind turbine blades are affected by turbulence over a range of heights, with more or less turbulence at higher or lower positions, respectively. Neglecting this effect is non-conservative because addition of fatigue loading weights larger loads higher than smaller loads. Both effects are small and opposite each other.

3 Wake turbulence and shear modelling

Turbulence, near boundaries and in the “free” flow, has received considerable attention over a longer span of time. Over the last 2-3 decades, increasingly powerful computers have allowed studies of the detailed structure of turbulence – in pursuit of understanding of the basic nature of turbulence. As of now, these efforts have resulted in promising computational tools by means of numerical solutions to the Navier-Stokes’ equations for various flow problems. However, so far the efforts have had limited effect on the development of operational engineering tools for prediction of turbulence characteristics, Hunt et al (2001). In this report, no attempt is made to go beyond practical applications of statistical turbulence theory.

In the previous section a global approach was applied to investigate the interaction between the cluster of wind turbines and the airflow. In this section the effect of each wind turbine on the airflow in its immediate vicinity is considered.

Doing so, we start with considerations regarding the development of turbulence in a wind farm configuration with close spacing of the wind turbine units in the rows perpendicular to the wind direction. Results from this particular wind turbine configuration supplement existing models regarding center-wake turbulence of individual wakes as function of distance to the wake-generating unit. Then initial turbulence in the near-wake is discussed and finally the development of turbulence in the far-wake is dealt with.

3.1 Turbulence between closely-spaced machines

In an attempt to link the global model of Section 2 for the horizontally averaged turbulence to turbulence in the individual wakes, the case of wind turbines narrowly-spaced in rows – say, 2 rotor diameters or less – perpendicular to the wind direction is studied, Figure 3.1.

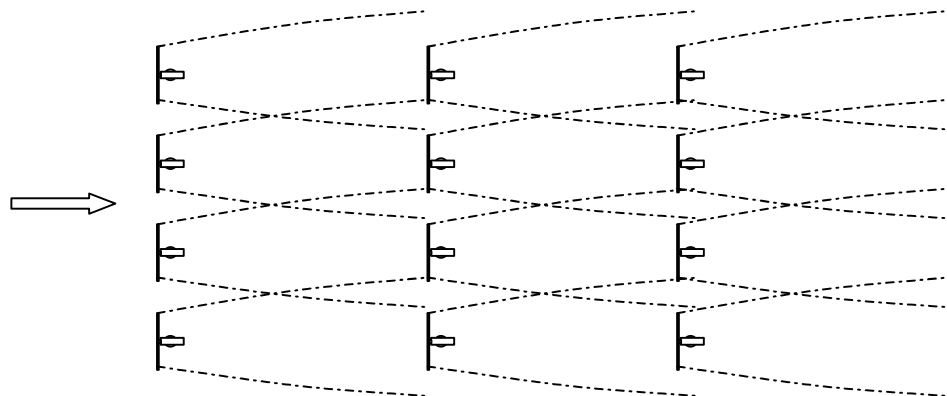


Figure 3.1 *Overlapping wakes: wind farm with wind turbines closely spaced in rows perpendicular to wind direction.*

The standard deviation of wind speed fluctuations σ_u varies between rows, being largest immediately behind each wind turbine row. In the case of nearly bladetip-to-bladetip siting of the machines in the rows⁵, it can be assumed that the wakes are significantly overlapping a few rotor diameters downwind, i.e. the standard deviation of wind speed fluctuations shortly downwind of each row does not vary later-

⁵ Despite concerns regarding fatigue, this way of siting wind turbines is usual in mountainous terrain, where there is a commercial need for intense use of land particularly well-exposed to the wind.

ally. In the turbulent boundary layer, the vertical turbulent momentum flux, $M_{vertical}$, is $\overline{\rho uw}$ where ρ , u and w are the density of air and horizontal and vertical wind speed fluctuations, respectively. The overbar denotes time averaging. The quantity $u_* = \sqrt{\overline{uw}}$ is the basic definition of the friction velocity. Further, in the neutrally stratified boundary layer, the standard deviation of wind speed fluctuations is found to be proportional to the friction velocity and thus

$$M_{vertical} \propto u_*^2 \propto \sigma_u^2. \quad (3.1)$$

In general Eq. (3.1) is taken to hold down to some distance *over* the layer embracing the wind turbine rotors (the rotor layer), where the flow by and large is horizontally homogeneous, longitudinally and laterally. Further, within the rotor layer turbulence varies between rows, being largest immediately downwind of each row and by assuming that Eq. (3.1) is also valid when the flow is longitudinally non-homogeneous, the global flow may be linked to the wake flow characteristics. The variance of wind speed fluctuations in the rotor layer – which for the particular wind farm geometry is identical to the wake – the dimensionless distance $\zeta = x/D_0$ downwind of each row of turbines can be written as

$$\phi_T^2(\zeta) = \sigma_0^2 + \phi_w^2(\zeta), \quad (3.2)$$

where σ_0^2 is the variance of wind speed fluctuations without the wind turbines, and $\phi_w^2(\zeta)$ is the variance added due to presence of the wind turbines. By assuming that the variation in variance $\phi_T^2(\zeta)$ is the same between each two rows, the development of turbulence between rows can be linked to the global impact of the wind farm. Since no external force is acting on the flow above the wind turbines, the average of the variance, taken between two rows, should be equal the variance found in the global considerations in Section 2:

$$\sigma_{T,wf}^2(s, s_r) = \left(\frac{1}{s}\right) \int_0^s \phi_T^2(\zeta) d\zeta = \sigma_0^2 + \left(\frac{1}{s}\right) \int_0^s \phi_w^2(\zeta) d\zeta. \quad (3.3)$$

where $\sigma_{T,wf}^2$ is given by Eqs. (2.17) and (2.18), $s_r = x_r/D_0$ is the wind turbine spacing in the rows (perpendicular to the wind direction) and s is the spacing between the rows. The last term on the right side of the equation corresponds to the added turbulence in the global considerations, Eq. (2.18), i.e.:

$$I_{addwf}^2 = \frac{a^2 C_T}{(b\sqrt{C_T} + \sqrt{s_r s})^2} = \frac{1}{s} \int_0^s i_w^2(\zeta) d\zeta, \quad \text{where } i_w(\zeta) = \frac{\phi_w(\zeta)}{U_0}, \quad (3.4)$$

where U_0 is ambient mean wind speed. Differentiating with respect to s , with s_r held constant, we get

$$i_w^2(s) = \frac{d}{ds} \left(a^2 C_T \frac{s}{(b\sqrt{C_T} + \sqrt{s_r s})^2} \right) = \frac{a^2}{b^2} \cdot \left\{ \frac{b\sqrt{C_T}}{b\sqrt{C_T} + \sqrt{s_r s}} \right\}^3 \Rightarrow \quad (3.5)$$

$$i_w^2(\zeta) = \frac{a^2}{b^2} \cdot \left\{ \frac{b\sqrt{C_T}}{b\sqrt{C_T} + \sqrt{s_r \zeta}} \right\}^3 \quad 0 < \zeta < s_r$$

Thus, for the particular wind farm geometry with closely in-row spacing of the wind turbines with merged one-dimensional wakes between the rows, an estimate of the wake turbulence intensity as function of downwind distance ζ from a row of wind turbines is

$$i_T(\zeta) = \sqrt{\frac{a^2}{b^2} \cdot \left\{ \frac{b\sqrt{C_T}}{b\sqrt{C_T} + \sqrt{s_r}\zeta} \right\}^3 + I_0^2} \quad 0 < \zeta < s_f, \quad (3.6)$$

where $I_0 = \sigma_0 / U$. Asymptotically, for increasing values of ζ , $i_T \propto \zeta^{-3/4}$.

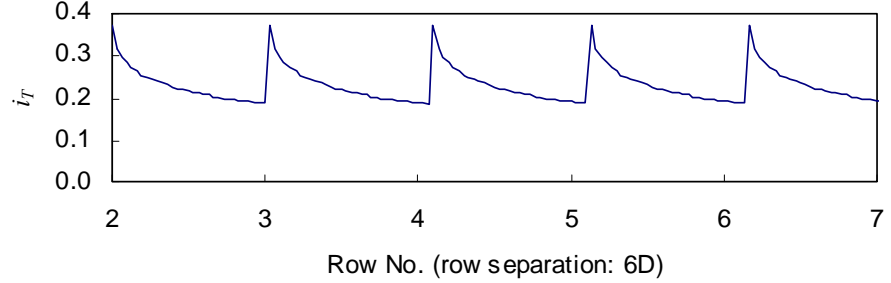


Figure. 3.2 Variation in turbulence between rows with close spacing in the rows. $C_T = 0.89$, $s_r = 2$, $s_f = 6$ and $I_0 = 10\%$.

For the chosen illustration, Figure. 3.2, turbulence intensity quickly drops off, starting at a level of 35% immediately behind each row. The derived result is for the “large” wind farm. However, it should be expected that balance occurs only a few rows from the front row: Initially – at the upwind edge of the wind farm – each row generates turbulent energy additional to the turbulence energy delivered from the row upwind,

$$\sigma_{i+1}^2 = \sigma_i^2 + \varepsilon_{i+1}^2, \quad (3.7)$$

but since there is an upper limit, viz. Eq. (3.6), to wake/boundary layer turbulence and since turbulence after the first row is already high compared to the upper limit, the added turbulence, ε_{i+1} , must vanish quickly as the row number increases. It is supported by measurements which show that the effect of multiple-wake conditions, relative to single-wake conditions, on the turbulence level is marginal, Frandsen et al (1996).

This result is applied in the following, adopting the *form* for a model of single-wake, added turbulence – as an alternative to the expressions proposed previously, Crespo et al (1999b), Teknisk Grundlag (1992) and Frandsen et al (1996).

3.2 Initial, added wake turbulence

The term initial wake turbulence is used for the turbulent wind speed fluctuations immediately behind the wind turbine rotor. In turn, “immediately behind the rotor” is a distance downwind where the air pressure in the wake has regained the ambient level and the wake has expanded to its near-wake radius, of the order 2-3 rotor diameters downwind. The basic engineering methods for calculation of propeller efficiency and loads were developed by Betz (1920), Lanchester (1915) and Glauert (1935). These early references offer quantification of the wake expansion and wake speed deficit, but not the distance it takes for the wake to expand to its initial width. For that, experimental data or numerical methods are needed. By means of momentum balance and Bernoulli’s equation, Betz (1920)⁶ developed the relationship be-

⁶ Bergey (1979) points out that Lanchester does the equivalent deductions five years earlier. Nevertheless has Betz been accredited the deductions and the rather famous limit for extraction of energy by means of a propeller.

tween the speed deficit immediately behind the rotor and the efficiency of the rotor. An alternative approach is offered in Appendix A.3, though resulting in the same expression for the power coefficient:

$$C_p = \frac{1}{2} b(2 - b)^2, \quad (3.8)$$

where the wind speed behind the rotor is $U_b = U_0(1 - b)$, U_0 being the free-flow speed and b the induction factor⁷. From the equation, the maximum obtainable efficiency of the rotor, $C_{p,\max} = \frac{16}{27} \approx 59\%$, is derived. The simplicity of the derivation of the result – though being ingenious in its own right – certainly has its limitations and was surpassed first by the blade element method, Glauert (1935), and later by methods involving numerical solution to the basic equations of motion of the air flow. Still, the Lanchester/Betz result and augmentations to that often prove useful, in particular for considerations regarding wake speed deficit. It should also be mentioned that, although advanced computational tools are being developed and to some extent used in engineering, Glauert's blade element method is still the work-horse for engineering design calculations due to its relative simplicity and not least its reasonable accuracy, Sørensen and Mikkelsen (2001).

As for initial wake turbulence, Crespo and Hernandez (1996) argue by means of local momentum and energy equations, that maximum added wake turbulence intensity is approx. 35%.

Højstrup (1999) reports from measurements that the near-wake turbulence intensity is 25-30% in flat, homogeneous terrain.

Wake turbulence data presented in the following Subsection 3.3 suggest – by means of extrapolation – that maximum, added turbulence could be of the order 35% at 2-3 rotor diameters downwind.

In Appendix A.3, turbulence has been included in the momentum and energy budgets as an extension to the Lanchester/Betz solution. Doing so, it is found that the maximum rotor efficiency of 59% can only be achieved if the wake is initially non-turbulent. For state-of-the-art wind turbines, the maximum efficiency is of the order 45+%, allowing for a maximum of approx. 20% initial turbulence intensity, which – when comparing with non-uniform wake profiles – should be understood as a sum-square-root average over the wake. Further, it is found that the maximum possible near-wake turbulence is approx. 45%, achieved at zero efficiency. Although the approach is different, the results are comparable to the result of Crespo and Hernandez (1996).

The theoretical and experimental evidence as a whole suggest, that the maximum added turbulence in the near-wake is between 30 and 45%, depending on the thrust on the rotor, on the type of wind turbine and possibly on the ambient turbulence at the site where the wind turbine is situated.

3.3 Downwind development of the wake

In the near-wake zone, the wake expanded until the pressure in the wake has reached the ambient level, probably $2-3D_0$ downwind, D_0 being the rotor diameter. At this point, a speed deficit has materialised and the emerging wake turbulence level is given by the level upwind of the turbine and the turbulence generated by the wind turbine rotor.

From the near-wake out to approx. $5-6D_0$, additional turbulence is mainly generated by the radial flow shear, dissipation starts to drain turbulent energy and the width of the wake increases and in the process the speed deficit is being reduced. The mean-

⁷ Often, the induction factor is chosen as half this value.

speed deficit profile reaches a point of balance, becoming approximately bell-shaped and having a maximum approximately in the centre of the wake. New turbulent fluctuations are now generated by radial shear and for that reason there is less generation in the centre of the wake. However, turbulent diffusion transports the turbulent fluctuations to the wake-centre, and like the profile of mean speed deficit, speed fluctuations appear with fair approximation to be bell-shaped. From $5-6D_0$ and further downwind the wake has – as to its shape, but not magnitude – “forgotten” its origin.

In the far-wake region, models for deficit and wake expansion are found in textbooks, e.g. Schlichting (1968), Tennekes and Lumley (1972) and Pope (2000), although the derived results are not in particular aimed at wakes behind wind turbines. The analyses – applying momentum conservation, self-similarity of deficit and turbulence profiles and constant eddy viscosity in the wake as tools – result for the axis-symmetric, turbulent wake in the prediction for wake expansion $D \propto x^{1/3}$,

and for wake deficit $\frac{U_b}{U_0} \propto x^{-2/3}$, where D is wake diameter and x is the distance

downwind. Also turbulent speed fluctuations are predicted by Tennekes and Lumley (1972), $\sigma_u \approx 0.35 \cdot (U_0 - U_b)$, i.e. the standard deviation of fluctuations is approx. 1/3 of the mean flow speed deficit. Specifically for wind turbines, as a means to expression of wake turbulence, the form of Eq. (1.2) is commonly used.

The maximum, additional turbulence is typically found to be a function of C_T^{n1} , C_T being the wind turbine’s thrust coefficient, and s^{n2} , where $s = \frac{x}{D_0}$ is the dimensionless distance downwind and $n1$ and $n2$ are exponents. One model for far-wake turbulence, Crespo and Hernandez (1996), is derived from computer simulations resulting in the following proposal for added wake turbulence:

$$I_{add} \propto (1 - \sqrt{1 - C_T})^{0.83} I_0^{-0.03} s^{-0.32}. \quad (3.9)$$

Another model derived purely from measurements, Quarton (1989), deviates substantially in its dependency on ambient turbulence intensity I_0 and separation s :

$$I_{add} \propto C_T^{0.7} I_0^{0.68} s^{-0.57}. \quad (3.10)$$

The deviation in the results may be explained by the fact that Quarton (1989) attempts to model the wake turbulence all the way from near-wake to far-wake. In wind turbine clusters, the separations are to be found in the range from $1.1D_0$ to $8-10D_0$; i.e. all three identified regions⁸ are represented. Therefore, for practical purposes, the theoretically better-founded models for the far-wake are only partly adequate in the context. However, it is noteworthy that more or less any model, which includes C_T and s as parameters, can be made to fit data with the “right” choice of model constants.

The mentioned wind turbine wake models have the weakness that they consecutively used for multiple-wake cases lead to non-limited wake turbulence levels.

Applied model for added wake turbulence

The above models for turbulence intensity range in dependency of wind turbine spacing from Quarton’s $s^{-0.32}$ to $s^{-3/4}$ for the special, multi-wake case of Subsection 3.1. Thus, there is not convincing convergence of the results.

Like the Quarton model, the model chosen here for added wake turbulence is engineered to fit measurements in all the mentioned three regions of the wake. Inspired

⁸ Some researchers define up to 5 regions.

by the result for the in-the-row closely spaced machines, Subsection 3.1, the following expression is adapted:

$$I_{add} = \frac{1}{c_1 + c_2 \frac{s}{\sqrt{C_T}}}. \quad (3.11)$$

In the near-wake (for $s \rightarrow 0$), this model far added wake turbulence is upward limited. For the far-wake, it has the property that $I_{add} \propto C_T^{0.5} s^{-1}$. The constants are chosen to best fit wake turbulence measured at a number of full-scale experiments.

The thrust coefficient is modelled on basis of measurements on a stall-controlled machine, Frandsen et al (1996), as

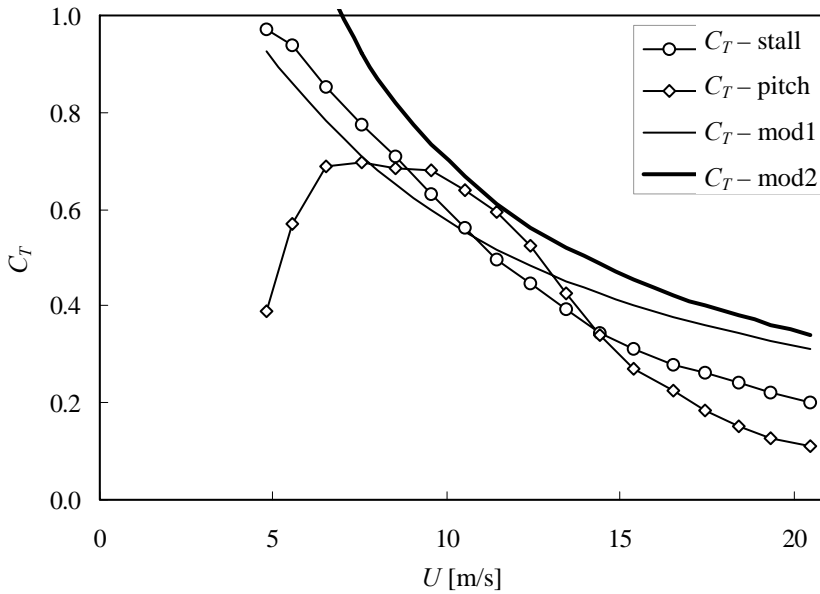


Figure 3.3 Thrust coefficients for stall controlled (circles, measured) and pitch controlled (diamonds, computed) machines, respectively. The lines are the models. The C_T -mod1 curve is the more elaborate C_T model and C_T -mod2 corresponds to the simpler expression, both given in Eq. (3.12).

$$C_T \approx \frac{3.5(2U - 3.5)}{U^2} \approx \frac{7 \text{ m/s}}{U}. \quad (3.12)$$

The model is compared with measurements and a computed C_T -curve for a stall-controlled and a pitch-controlled wind turbine, respectively, Figure 3.3. The approximation appears conservative at low and high wind speeds. In particular, it is noted that the pitch controlled rotor has a significantly smaller thrust coefficient at higher wind speeds than the suggested “default” curve.

Applying the simple model for C_T (right-most expression in Eq. (3.12)), the expression of Eq. (3.11) is fitted to the data of Figure 3.4, resulting in the following model for maximum added wake turbulence:

$$I_{add} \approx \frac{1}{1.5 + 0.3 \cdot s \cdot \sqrt{U}}. \quad (3.13)$$

From this expression, the total standard deviation of wake wind speed fluctuations is determined as $\sigma_{u,T} = U_0 \sqrt{I_{add}^2 + I_0^2}$. The model for I_{add} (instead of the best fit) weights small separations more heavily than the larger separations.

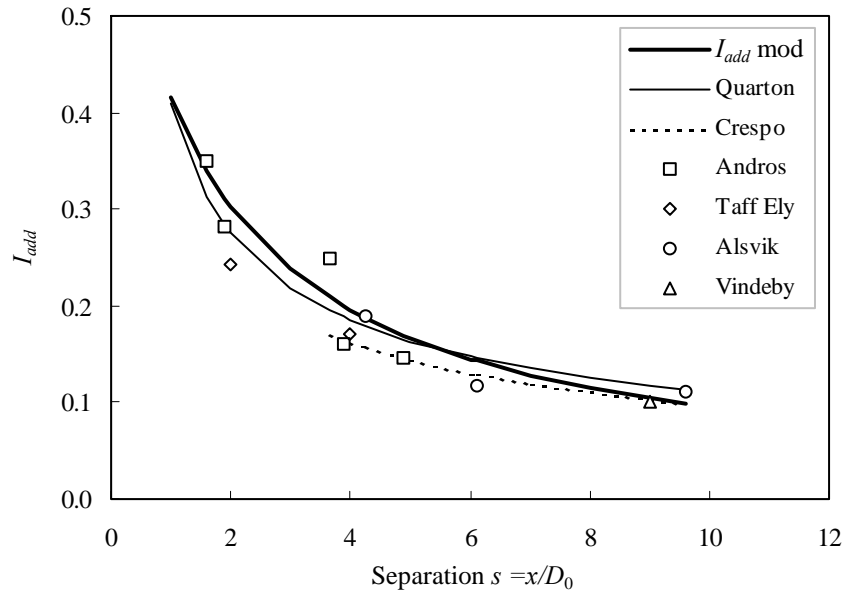


Figure 3.4 Maximum, added hub height wake turbulence measured in four different wind turbine clusters, compared with the applied model as well as the Quarton model and the Crespo and Hernandez model. $9\text{m/s} < U < 11\text{m/s}$. The data were compiled by Ghaie (1997)⁹.

Also shown in Figure 3.4 are the models of Crespo and Hernandez (1996) and Quarton (1989). The Quarton model has been made to fit Eq. (3.13) at $s = 2$ and the Crespo model was forced to fit at $s = 9$. The three models must be said to fit equally well and – on top of the previous arguments – it is therefore found justified to select the simplest solution that covers the whole relevant range of s , viz. Eq. (3.13).

The above wake-turbulence model, Eq. (3.13), is adapted in the model for effective turbulence.

3.4 Wake-generated mean flow shear

When a wind turbine is only partly affected by a wake, the wake's mean flow speed deficit will generate an alternating load on the rotor blades, Figure 3.5. For the primary purpose of “extending” the rather scarce verification measurements, a model for the flow deficit is needed. Like for the wake turbulence model it is assumed that a one-dimensional model for the wake deficit is sufficient to cover the real two-dimensional wake.

Momentum balance for a control volume around the rotor is employed to get a simple expression for the deficit. It is assumed that the deficit at a given distance downwind is constant across the wake:

⁹ Personal communication.

$$\frac{1}{2} \rho U^2 C_T \pi \left(\frac{D_0}{2} \right)^2 = \int_0^{D/2} \int_0^{2\pi} \rho r U_b (U - U_b) d\theta dr = \rho \pi \left(\frac{D}{2} \right)^2 U_b (U - U_b) \Rightarrow$$

$$\tau_w = U - U_b = \frac{1}{2} \frac{U^2}{U_b} C_T \left(\frac{D_0}{D} \right)^2 \approx \frac{1}{2} U C_T \left(\frac{D_0}{D} \right)^2 \quad (3.14)$$

where U_b and D are the wake mean wind speed and width, respectively. The expression is valid for wake wind speeds larger than approximately half the free-stream wind speed¹⁰. The wake width is discussed in further detail in the context of energy extraction, Section 9. At this point, a model with linearly expanding wake is applied, Jensen (1983):

$$\frac{D_0}{D} = \frac{1}{1 + \beta_0 \cdot s}, \quad (3.15)$$

where $s=x/D_0$ is the non-dimensional downwind distance and β_0 a constant. The linear expansion of the wake is contrary to both experimental and theoretical common knowledge, viz. the references made previously in this section. In terms of energy extraction by the wind turbines in clusters,

the linear expansion model will prove inadequate, see Section 9. However, in the

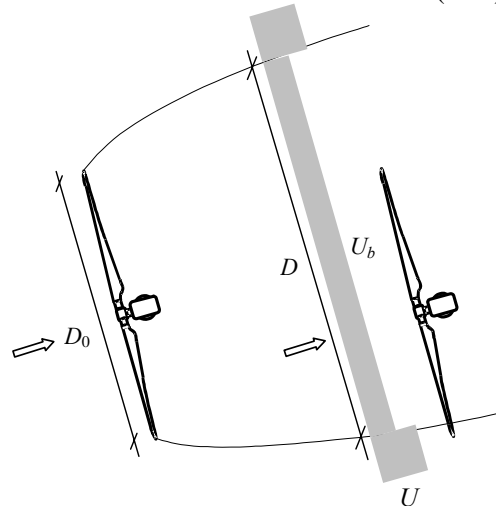


Figure 3.5 Wake deficit as experienced by downwind wind turbine.

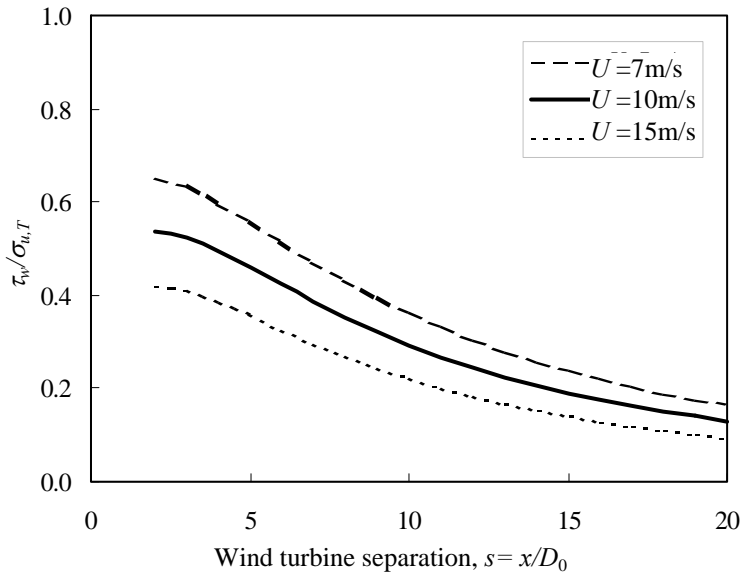


Figure 3.6 Ratio of models for wake mean wind speed deficit τ_w and wake maximum turbulence $\sigma_{u,T}$, as function of wind turbine separation.

¹⁰ A similar result is obtained by assuming a bell-shaped deficit in the wake, $\propto \exp(-(r/D)^2)$, integrating to infinity.

context of structural loading the linearity assumption proves useful and is justified by the general level of accuracy of the proposed model.

According to Jensen (1983), β_0 is 0.15-0.2, increasing with ambient turbulence intensity. The specific choice is of some importance in developing the model for effective turbulence. However, full-scale experimental evidence barely allows identification of differences in wake expansion and a fixed value of $\beta_0 = 0.17$, not dependent of ambient turbulence intensity, is applied in the following.

The ratio of the modelled wake deficit and maximum standard deviation of turbulent speed fluctuations in the wake, $\tau_w/\sigma_{u,T}$, is plotted in Figure 3.6, showing that the ratio is expected to decrease for increasing separation between the affected and the wake-generating wind turbine. Thus, for larger separations the deficit becomes less important relative to turbulent wind speed fluctuations.

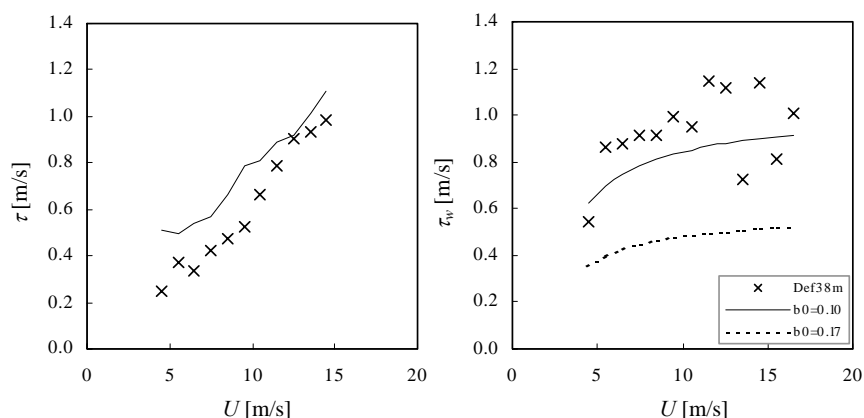


Figure 3.7 Vertical tip-to-tip shear in left plot and maximum wake mean speed deficit in right plot. Crosses are bin-averaged data values and lines are models, Eqs (2.2) and (3.14). β_0 stands for β_0 . Data from Vindeby.

The models for free-flow vertical shear and wake flow deficit are compared with data from the Vindeby experiment, Figure 3.7. The vertical free-flow shear was measured as the difference between wind speed at $z = 7\text{m}$ and $z = 38\text{m}$, which difference was transformed to the difference between lowest blade tip position (20m) and highest tip position, 55m, assuming neutral stratification and logarithmic vertical wind profile, see left plot in Figure 3.7. The wake speed deficit data – right plot in Figure 3.7 – are the difference between the wind speeds measured at hub height in the western met mast (SMW) and the southern met mast (SMS), see Figure 1.2, for wind directions $310\text{-}320^\circ$. For the wake-expansion constant β_0 , values of 0.1 and 0.17 have been tested, the lower value giving the best fit, possibly because of the low ambient turbulence intensity encountered offshore. In the following the higher value of β_0 is maintained, which constitutes a conservatism as to the validity of the method since it will make the sensitivity to wake deficit appear higher than it is.

3.5 Wake expansion and shape of turbulence profile

The wake profile of turbulent fluctuations can with good approximation be considered to be a superposition of the axis-symmetric wake turbulence generated by the rotor and the vertical free-flow turbulence profile. However, only when the mean wind direction is parallel to the connection line between the wake-generating and wake-affected turbine ($\theta = 0$) will the wake loading be symmetric. For other wind directions – as illustrated in Figure 3.8 – the real two-dimensional wake exposes the

downwind unit to a non-symmetric and inhomogeneous field of turbulent wind speed fluctuations. The degree of in-homogeneity varies with turbine separation. For closely spaced wind turbines (corresponding to the illustration in Figure 3.8) the variation of σ_u over the wake-affected unit is large and non-linear. For large separations the wake-affected unit's rotor tends to extend over only a limited part of the wake, consequently with lesser and more linear variation in σ_u over the rotor.

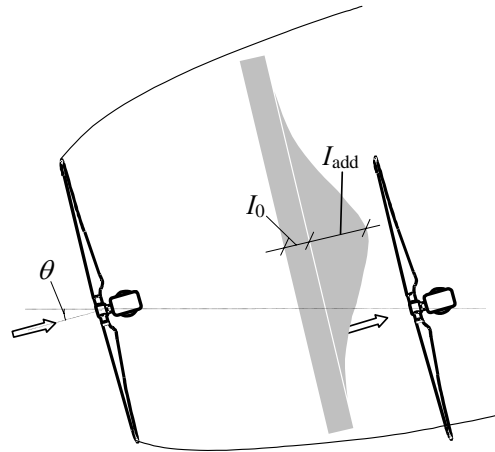


Figure 3.8 *Illustration of wake turbulence as experienced by downwind turbine.*

An additional complication is that at a given mean wind direction, the wake appears to be meandering. The phenomena is to some extent a matter of definition: with a short averaging time applied for determination of the (mean) wake profile, large-scale turbulent eddies will not be captured in the averaging but will appear as meandering. For large averaging times large turbulent fluctuations are captured in the averaging. For typically applied averaging times (10-30min), meandering *does* appear to have the effect that the mean wind speed deficit “sweeps” over the rotor thereby adding to the turbulent fluctuations of the non-meandering wake. The apparent lateral movement of the wake has been investigated by Thomsen and Madsen (2005), who carried out numerical simulations for the wake and response of a turbine affected by the wake. The response simulations were in agreement with measurements, which displayed an intermittent character possibly consistent with a distinct difference between the scales of (smaller) turbulent fluctuations and meandering scales.

An alternative explanation to the low-frequency, lateral movement of the wake is vortex shedding similar to what is seen behind linear bluff bodies. Medici and Alfredsson (2004) measured wind speeds in the wake behind a model wind turbine in a wind tunnel under low ambient-turbulence conditions. They identified wind speed fluctuations near the edge of the wake at a frequency of $\frac{U \cdot St}{D_0}$, where U is mean

wind speed, D_0 is rotor diameter and St is the Strouhal number, which for the reported measurements took values between 0.12 and 0.2.

These findings add to the understanding of the load impact of the wake, but also add complexity. In here, no distinction is made between wind speed fluctuations of different kinds. The presumption is that the rotor structure is insensitive to the detailed character and origin of fluctuations.

Thus, with the rotor-generated wake turbulence profile assumed bell-shaped, the turbulence intensity as experienced in the centre of the wake affected rotor can be expressed as

$$I = I_0 \left(1 + \alpha \exp \left(- \left[\frac{\theta}{\theta_w} \right]^2 \right) \right), \quad (3.16)$$

where I_0 is ambient turbulence, θ is the angle between the connection line of the two wind turbine units and the wind direction, Figure 3.8, and θ_w is defined below. Referring to Eq. (1.2), the constant α can be expressed by the ambient turbulence intensity and maximum added wake turbulence intensity:

$$I_T = I_0(1 + \alpha) = \sqrt{I_{add}^2 + I_0^2} \Rightarrow \alpha = \sqrt{\left(\frac{I_{add}}{I_0}\right)^2 + 1} - 1 \quad (3.17)$$

In Eq. (3.16), θ_w is a characteristic view-angle of the wake-generating unit, seen from the wake-effected unit. The view-angle from the wake-affected to the wake-generating wind turbine is deduced by means of the wake-width, Eq. (3.15).

Despite the lack of convincing theoretical background, the expression of Eq. (3.15) fits data fairly well for a broad range of separations. Eq. (3.15) leads to the following model for the characteristic view-angle θ_w in Eq. (3.16):

$$\theta_w \cong \frac{1}{2} \left(\frac{180}{\pi} \cdot \tan^{-1}(1/s) + 10^\circ \right) \quad [\text{deg}]. \quad (3.18)$$

Thus, the hypothesis is that for a given mean wind direction, θ , the turbulence intensity experienced in the centre of the wake-effected rotor applied over the whole rotor yields approximately the same response as if applying the spatially distributed turbulence intensity. The assumption was graphically tested by mean of measurements in Figure 1.4 and is exposed to further testing in the following section.

3.6 Summary

A model for the horizontal turbulence intensity profile as experienced in the wake of a wind turbine was devised. The proposed model for the wake-generating wind turbine's thrust coefficient is chosen so that, for all wind speeds, it is higher than what is measured/calculated for stall-controlled and pitch-controlled wind turbines, respectively. The resulting model for wake turbulence as function of distance to the wake-generating wind turbine is presumed conservative.

4 Method and justification

As stated in Subsection 1.3, the approach of the proposed method is based on the observation that the along-wind turbulent fluctuations, characterized by their standard deviation σ_u , is the single-most important external quantity that causes dynamic response of the wind turbine structure, and that this is the case whether the wind turbine is exposed to the free flow or in the wake of another wind turbine. An additional assumption is that should indeed other flow variables change from non-wake to wake conditions, these variables are either of lesser significance to structural response or consistently collinear with σ_u .

A direct approach would be to describe in detail the air flow as experienced by a wind turbine unit inside the wind farm, then derive from the flow description the loads and subsequently run the needed number of simulations. However, the details of the flow structure inside the wind farm are complicated and presently not well understood. In addition, the applied computer codes set limits to the overall accuracy of the design process. Thus a study, Heijdra (2003), that compared computational result from 8 different aeroelastic codes with measurements from 4 wind turbines showed large discrepancies between the codes. For tower base overturning moments and flapwise blade bending moment, the standard deviation of the computational results from the different codes were determined. Three times the standard deviation¹¹ of the 8 computed estimates of equivalent load for flapwise blade and tower bending were 25-30% and 25-100%, respectively. The discrepancy signifies a considerable uncertainty in present design computations. Heijdra (2003) points to several potential reasons for the large discrepancies, amongst which the lack of precise information on the flow input.

Therefore, The proposed model for effective turbulence is also justified by the uncertainties in design calculations in general and by the limited understanding of the flow specifically.

4.1 General on loads on wind turbines

However, the proposed approach is first of all justified because the dynamic response of the wind turbine structure basically *is* proportional to wind speed fluctuations and because σ_u in general terms *is* a key parameter for the dynamic response of any wind-exposed structure when dealing with fatigue.

Consider first a cylinder, e.g. a chimney-stack or a (separate) wind turbine tower, extending vertically from a fairly flat ground. Disregarding lateral wind speed fluctuations, the wind load on a unit-section of the chimney, Figure 4.1, is

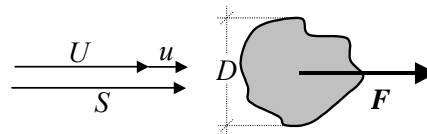


Figure 4.1 *Wind load on a cylinder, e.g. a wind turbine tower.*

$$F = \frac{1}{2} \rho S^2 D C_D, \quad S(t, z) = U(z) + u(t, z), \quad (4.1)$$

where ρ is the air density, D the cross-wind extension of the cylinder, C_D the drag coefficient of the cylinder, and S the flow speed relative to the cylinder at time t and height z in the direction of the mean wind¹². The mean wind speed is not time-

¹¹ If the computational results be normal distributed, three times the standard deviation signifies a range around the mean that contains approx. 85% of the results.

¹² In the context being of no consequence, it is neglected that the movement of the structure also influences the flow speed, giving rise to aerodynamic damping.

dependent, and for neutrally stratified atmosphere the standard deviation of the turbulent wind speed variations u is not height-dependent, see Subsection 2.1. To make the load linear in u , the load per unit length of the cylinder is approximated as

$$F(t, z) = \frac{1}{2} \rho D C_D (U^2 + u^2 + 2Uu) \approx \frac{1}{2} \rho D C_D U^2(z) + \rho D C_D U(z)u(t, z). \quad (4.2)$$

Assuming that the dynamics of the cylinder are described well by its first mode of vibration (with deflection in the along-wind direction) the deflection of the cylinder can be described by the equation

$$\ddot{y} + 2\xi\omega_0\dot{y} + \omega_0^2 y = \frac{1}{m_e} \int_0^{H_C} F(t, z)\zeta(z)dz \approx \frac{\bar{F}}{m_e} + \frac{\rho D C_D}{m_e} \int_0^{H_C} U(z)u(t, z)\zeta(z)dz, \quad (4.3)$$

where y is the generalized deflection of the cylinder, ω_0 the eigenfrequency, ξ the damping ratio, H_C the height of the cylinder, \bar{F} the generalised mean load, ζ is the normalized natural mode shape and m_e is the equivalent mass of the system. Through some manipulations of Eq. (4.3) including restrictions on the power spectrum of wind speed, the power spectrum and variance of y are found:

$$S_y(\omega) \propto \sigma_u^2 |H(\omega)|^2 S_u(\omega) F_{adm}(\omega) \Rightarrow \sigma_y^2 \propto \sigma_u^2 \int_0^\infty |H(\omega)|^2 S_u(\omega) F_{adm}(\omega) d\omega, \quad (4.4)$$

where $|H|^2 = \left\{ \left((\omega/\omega_0)^2 - 1 \right)^2 + \left(2\xi(\omega/\omega_0) \right)^2 \right\}^{-1}$ is the squared frequency transfer function, F_{adm} is the structural admittance function that accounts for the extension of the turbulent eddies relative to the height of the cylinder and the different deflection of different sections of the cylinder. S_u is the normalised power spectrum of u at a reference position, e.g. the top of the cylinder. And further, if u is normal distributed so is the response.

In turn, if the response is narrow band (low damping of the structure), then the *amplitude* process will be Rayleigh distributed, and for that case Crandall and Mark (1963) have implicitly demonstrated that the equivalent load is proportional to the standard deviation, σ_y , see appendix A.1. Thus, in case of a linear structural system subjected to entirely random Gaussian turbulent loading, the equivalent load of the response will be exactly proportional to the standard deviation of wind speed fluctuation.

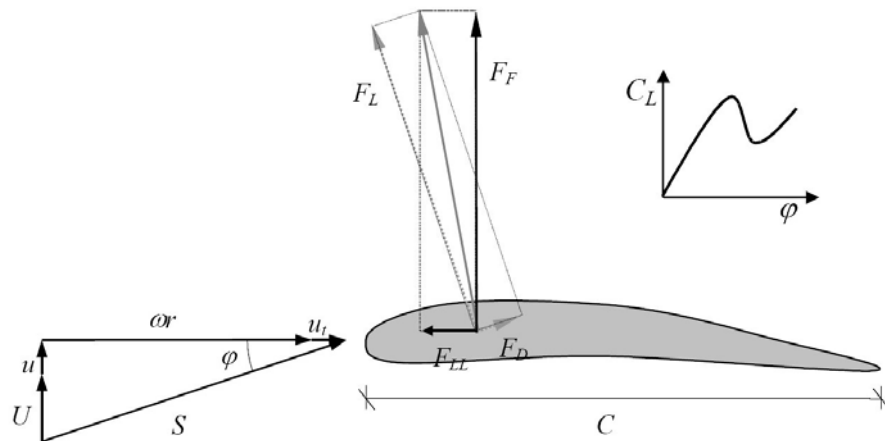


Figure 4.2 Section of wind turbine blade and, lift coefficient curve. U is mean wind speed, u turbulent wind speed fluctuations in mean wind direction, u_t wind speed fluctuations perpendicular to mean flow direction, S resulting flow speed, ϕ angle of attack, ω rotational speed of section, r distance from rotor centre to blade section, C chord length of section, F_L lift force on section, F_D drag force on section, F_F and F_{LL} projections of resulting blade force onto directions perpendicular and parallel to rotor disc, respectively, and $C_L = C_L(\phi)$ lift coefficient as function of angle of attack.

tuations.

However, for the operating wind turbine, the wind loads on the tower are marginal compared to the blade loads, Figure 4.2. Loads on a blade section are characterised by a lift and a drag force, being defined as perpendicular and parallel, respectively, to the flow direction as experienced by the moving blade section.

Lift and drag forces are given by $F_L = \frac{1}{2} \rho C S^2 C_L$ and $F_D = \frac{1}{2} \rho C S^2 C_D$, i.e. these are proportional to the air density ρ , to the cord length C , and the square of the resulting flow speed S experienced by the rotating blade section. What characterise the specific blade profile are the lift and drag coefficients, which are functions of the flow-angle of attack φ . The coefficients are determined from wind tunnel tests – and more recently also by means of numerical methods. The outer part of the blade delivers the largest contribution to the energy capture and experiences the largest loads, basically because there the speed of the blade, ωr , dominates in the composition of S^2 when combined with the wind speed. For contemporary wind turbines, the maximum tip speed during normal operation is of the order 60-90 m/s and thus it could be said that a wind turbine blade most of its life is exposed to hurricane-level wind speeds. To obtain the force that drives the rotor, the resulting force on the blade section is decomposed in a force in the rotor plane, F_{LL} , and a force perpendicular to the rotor plane, F_F .

In reality, the rotor interacts with the flow by slowing it down in the process of extracting energy. The rotor induces flow-speed changes in both the axial and tangential direction. Glauert (1935) demonstrated how in practical terms to compute the blade loads by dividing the rotor into annular sections, by setting up the balance of momentum between the blade section and the corresponding flow section and by an iterative process deriving the *induction factors* and in turn the sectional forces. The underlying assumptions of Glauert's *blade element method* were examined by Sørensen and Mikkelsen (2001), showing an acceptable degree of validity. Therefore, and for its relative simplicity, the blade element method is still the workhorse in engineering calculations.

For strength calculations, the out-of-plane force F_F is the more important since it is acting in the weak direction of the blade structure and at the same time the effect of F_F integrated over the rotor propagates more or less directly to the tower. A relevant question, when devising the turbulence model for wind turbine clusters, is whether also crosswind turbulent fluctuations – u_t in Figure 4.2 – must be considered. By means of approximations this may be evaluated as follows. In design load calculations, a feasible technique when dealing with the dynamic load component is to “freeze” the induction factors at values corresponding to the mean wind speed, thus avoiding the mentioned iteration sequence at each time step in the computer simulations. The induction factors are therefore disregarded, and it is noted that for the relevant (small) values of φ , $F_F \approx F_L$. Further, theory for thin airfoils and incompressible flow leads to a lift coefficient of $C_L = 2\pi\varphi$, φ in rad, for angles of attack below stall (for real airfoils the coefficient is a little less than 2π). Turbulent fluctuations are small compared with the mean wind speed, $u \ll U$ and $u_t \ll U$, and as previously mentioned, for the important outer parts of the blade, $\omega r \gg U$. Crosswind fluctuations are of the order 50% of along-wind fluctuations. The angle of attack is approximately $\varphi \approx \frac{U + u}{\omega r}$, where the pitch setting of the blade has been neglected. With these assumptions and dismissing second order terms, the following approximation for the flapwise force (per unit length of the blade) acting on the outer part of the blade can be deduced:

$$\begin{aligned}
F_F &\approx \frac{1}{2} \rho C C_L S^2 \approx \frac{1}{2} \rho C 2\pi \frac{U+u}{\omega r} \left((\omega r + u_t)^2 + (U+u)^2 \right) \\
&\approx \rho C \pi \bar{S}^2 \frac{U+u}{\omega r} \left(1 + 2 \frac{u_t}{\omega r} \right),
\end{aligned} \tag{4.5}$$

where $\bar{S}^2 = (\omega r)^2 + U^2$. A suitable measure of relative importance of u and u_t could be to compare the impact of the standard deviations of the two quantities on the flapwise loading of a rigid blade:

$$\frac{\frac{\partial F_F}{\partial u_t} \sigma_{u_t}}{\frac{\partial F_F}{\partial u} \sigma_u} \approx \frac{2 \frac{U}{\omega r} \cdot \frac{1}{\omega r} 0.5 \sigma_u}{\frac{1}{\omega r} \sigma_u} = \frac{U}{\omega r}, \tag{4.6}$$

where σ_{u_t} is standard deviation of u_t . With a typical blade tip speed of $\omega R = 70$ m/s and for typical operation wind speeds of $U = 10$ - 15 m/s, the above ratio of impact is 0.14 to 0.21. Thus, the crosswind turbulence does play a role in blade loading, but it can be expected to be much less than that of the along wind turbulence.

Because of the rotating sub-structure of the wind turbine – the rotor – the complete equations of the dynamics have periodic coefficients. If the tower structure is relatively stiff the effect of the non-constant coefficients is small, and in this case it is possible to separate rotor and tower calculations. If – on the other hand – the structure is “soft”, the designer seeks to eliminate the effects of the periodicity of the coefficients in avoidance of coinciding structural eigenfrequencies and rotor-rotational frequencies and multiple thereof. Thus, in navigating around such possible dynamic amplifications, also the effects of the non-constant coefficients will be less penetrating.

In design calculations, response of the wind turbine structure is determined by finite element or similar models, and by computational means the coupled equations of dynamics of the structure,

$$\mathbf{M}(t)\{\ddot{y}\} + \mathbf{C}(t)\{\dot{y}\} + \mathbf{K}(t)\{y\} = \{p(t, \dot{y}, \ddot{y})\}, \tag{4.7}$$

is solved. $\mathbf{M}(t)$, $\mathbf{C}(t)$ and $\mathbf{K}(t)$ are the time-dependent mass, damping and stiffness matrices, respectively, $\{y\}$ is the vector of translations and rotations and $\{p(t, \dot{y}, \ddot{y})\}$ is the force vector. Due to the time dependency of the coefficients, frequency domain methods are difficult, but possible to apply on the system. These are avoided presently also because transient load situations like start and stop only with great difficulties can be handled in the frequency domain and because the loading may be non-linear in the deflection and derivatives thereof.

Therefore, solutions are found by time integration, over e.g. 10-minute periods, from which relevant statistics are extracted. The statistics would include mean, standard deviation, power spectra and extremes, from which ultimate loads/stresses are derived.

Further, from the response time series the load spectra – the range of loading as function of the number of cycles – which are used for fatigue-life time estimation, are determined. Initially in this section it was argued that standard deviation of a selected response quantity σ_y is proportional to wind speed fluctuations σ_u , and in turn that also fatigue in terms of the equivalent load is proportional to σ_u .

Fatigue occurs when the structure is subjected to repeated cyclic loading. Laboratory tests give the relation between the amplitude of stress and number of cycles that a given material can withstand before failure occurs. The results of the lab tests are the SN- or Wöhler curves. These have for a significant range of cycles the form $n_{fat} \propto s_{fat}^{-m}$, where s_{fat} is the stress range, n_{fat} the number of cycles at that stress

range which will cause fatigue-failure and m is an exponent depending on the material. For fibreglass reinforced polyesters values of the exponent are of the order 10-12 and for steel $\approx 3-5$. In reality, the failure-causing amplitudes are limited for low and high numbers of cycles. Other models for material fatigue include also the mean load level, but for simplicity the refinement is neglected in the present context. Should the mean load level be relatively high compared to the dynamic variations, it is possible to adjust the modelling of the effective turbulence by means of the Goodman criterion, see Subsection 5.2.

When the stress amplitudes are not constant but e.g. random, the Palmgren-Miner rule of partial damage is applied, Miner (1945) and Appendix A.1. Computing the load spectrum the rainflow counting method is used, Matsuishi and Endo (1968). To facilitate simplified combination of several load cases, the concept of damage rate may be applied, Crandall and Mark (1963) and Appendix A.1. In connection with wind turbines, the concept of equivalent load was defined and applied for the first time by Madsen et al (1984), and later on in the context of increased loading under wake conditions by Stiesdal (1991).

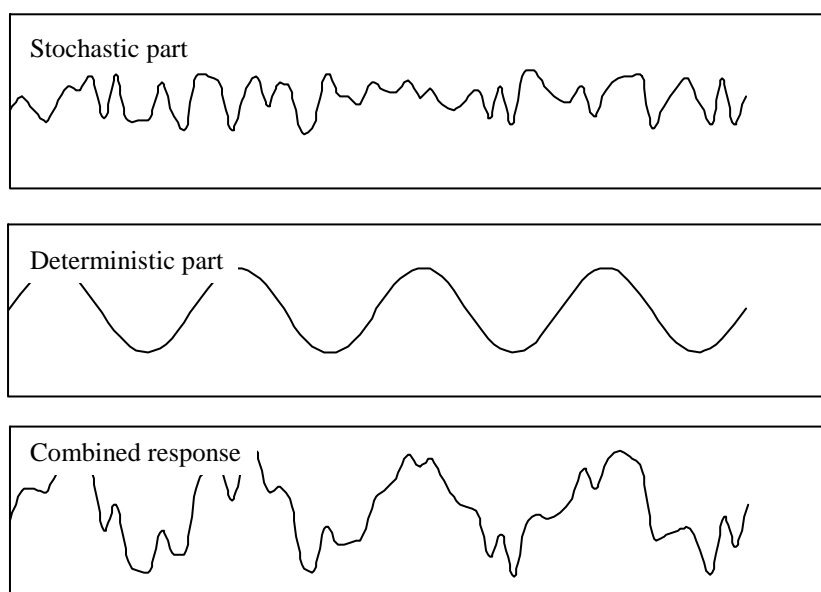


Figure 4.3 *Illustration of response of wind turbine component, combined of a deterministic and a stochastic component.*

Many wind turbines, presently in operation, are equipped with induction generators or variations hereof. The implication is that variation in revolution speed is limited to a few percent. Therefore, the typical structural response of wind turbine components is composed of a near periodic deterministic term caused by cyclic gravity loading on the blades and vertical and horizontal mean flow-shear, and a stochastic term caused by turbulent wind speed fluctuations. The combined response is illustrated in Figure 4.3. The periodic deterministic loading can be expressed as a Fourier series with the frequency of revolution of the rotor as the fundamental frequency. In the illustration, the periodic signal is simply a sinusoid. Due to the rotation of the blades, the stochastic response has spectral peaks at the revolution frequency and harmonics hereof, Kristensen and Frandsen (1982). Though both have spectral concentration of energy at the rotation frequency, the stochastic and deterministic load components are not correlated. Appendix A.1 summarises analytical work by Madsen et al (1984), based on Rice (1944), of the fatigue-effect of combined deterministic and stochastic response on wind turbine blades. The bending moments with a dominating deterministic component stemming from gravity are the driving bending moments of the blades and the bending moments of the trans-

mission shafts.

Most new wind turbines are operated with variable speed of the rotor, in contrast to the majority of existing wind turbines. Variable-speed operation gives higher energy extraction, additional options for the interaction with the grid and/or possibilities for structural-load limitation. Depending on the purpose for which it was designed, variable-speed operation may cause correlation between the frequency of revolution and the stochastic loading. It is, however, stipulated that the *amplitude* of such frequency-modulated loads (gravity and shear) is not correlated to the stochastic loads.

The flap-wise blade bending moment is chosen for the below analysis. Usually, flap-wise bending moment does have a deterministic component due to coning and tilt of the rotor and mean vertical shear of the airflow. Of these, the importance of the vertical shear is included in the analysis.

4.2 Linearising equivalent load

Analytical considerations regarding narrow-band random processes lead to a fixed ratio of the equivalent load e and the standard deviation of response σ_y , Appendix A.1. In all, it was found to be a reliable presumption that equivalent load is proportional to standard deviation of wind speed, σ_u :

$$\text{For fixed } U: \sigma_y \propto \sigma_u \wedge e \propto \sigma_y \Rightarrow e \propto I. \quad (4.8)$$

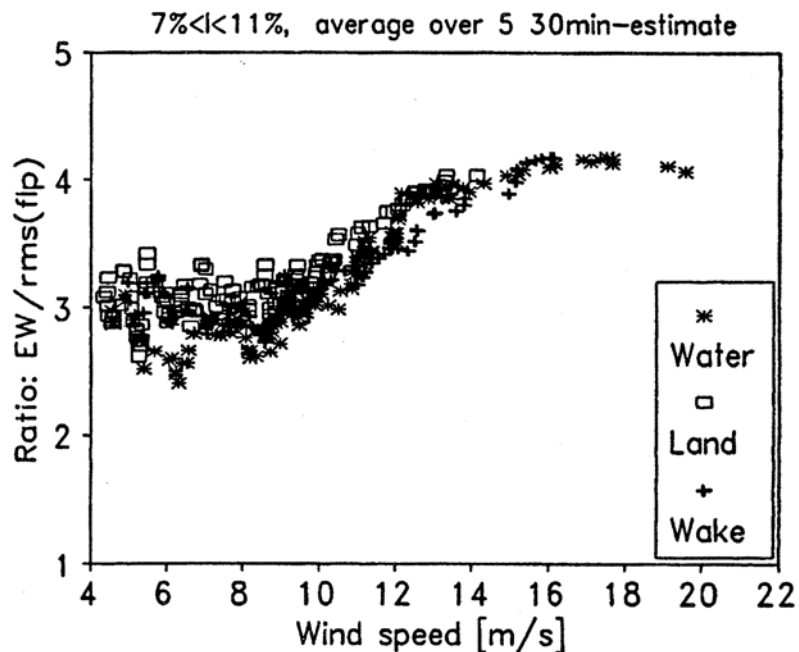


Figure 4.4 Flap-wise bending of blade: ratio of measured equivalent load (EW : flap-wise bending moment, $m = 5$) to standard deviation of the same quantity, as a function of the wind speed for various in-flow conditions. From Frandsen et. al (1996).

The mentioned link between e and σ_y is supported by measurements, Figure 4.4 . It is seen that for a given mean wind speed, the ratio equivalent load and the standard deviation of response fluctuations is approximately constant for different in-flow conditions including also wake conditions.

The ratio increases with increasing wind speed, indicating that equivalent load is more sensitive to the frequency-scale of turbulence (which increases with wind speed, see Subsection 2.2) than to σ_y . And/or that response becomes increasingly narrow-band as wind speed increases, which effectively moves the process to higher frequencies and thus higher equivalent load, see Appendix A.1. That the ratio between e and σ_y varies with mean wind speed is of no consequence to the validity of the method since the model for effective turbulence is conditioned on wind speed.

A priori, it is expected that equivalent load – like σ_y – also is a function of other variables. Disregarding phenomena like stall-induced vibrations and flutter¹³, vertical shear of the mean flow and scale of turbulence are known sources to dynamic loading. Vertical shear and the time scale of turbulence will vary as function of mean wind speed. In addition in the interior of the wind farm, the wakes of the neighbouring machines result in a complicated flow-structure, which herein is idealised as horizontal variation of the mean flow and turbulent fluctuations.

Thus, with approximate linearity between input and output quantities of the dynamic equation of the structure, Eq. (4.7), the response in terms of equivalent load, determined for 10-30min periods, may be written as

$$e(U, \theta) \approx \alpha_{\sigma_u} \sigma_u + \alpha_{\tau} \tau + \alpha_w \tau_w + \alpha_f f, \quad \alpha_* = \frac{\partial e}{\partial (*)}, \quad (4.9)$$

where $\tau = \tau(U, \theta)$ is the vertical mean shear of the flow, $\sigma_u = \sigma_u(U, \theta)$ is standard deviation of wind speed fluctuations, $\tau_w = \tau_w(U, \theta)$ is the wake-induced mean flow speed deficit (horizontal shear) and $f = f(U, \theta)$ is a frequency-scale of turbulence. Mean wind speed is denominated U and θ is the wind direction. The quantities α_* are the sensitivity coefficients of the respective variables.

While turbulent wind speed fluctuations are random, the vertical and horizontal shear are by definition deterministic. Rice (1944) dealt with the combination of a deterministic (sinusoidal) and a random process, see Appendix A.1. The result of Rice is not a linear rule of combination as suggested above. However, the linearization is adequate in a fairly wide range of the ratio between the stochastic and the deterministic component.

The four defined input variables will in general have a non-trivial joint-PDF. The distribution of input variables and their correlation are functions of mean wind speed and direction. Denoting the conditional joint-PDF of the input variables $g = g(\sigma_u, \tau, \tau_w, f | U, \theta)$, the integrated equivalent load raised to the power m may be written as:

$$e^m(U, \theta) = \iiint (\alpha_{\sigma_u} \sigma_u + \alpha_{\tau} \tau + \alpha_w \tau_w + \alpha_f f)^m g(\sigma_u, \tau, \tau_w, f | U, \theta) d\sigma_u d\tau d\tau_w df \quad (4.10)$$

Though not indicated in Eqs. (4.9) and (4.10), the sensitivity coefficients may be functions of wind speed. The extent to which the sensitivity coefficients are functions of wind direction when considering the interior of a wind farm is investigated in the following. Thus, even with the chosen simplified approach, the task is getting more complicated than desired.

For the comprehensive analysis of fatigue, the joint-PDF must be known in detail, and deriving it from data is not straightforward. The challenging issue is touched

¹³ Such vibrations may be initiated without external excitation and can not be neglected. However, in designing a wind turbine structure separate efforts are put into avoiding these vibrations and in terms of fatigue these should be dealt with separately.

upon in Subsection 5.1 by considering merely the PDF of standard deviation of wind speed fluctuations. Although being potentially important, the joint-PDF of the input variables is an issue not specific to wake loads.

On the background of these considerations, the task of justifying the method boils down to demonstrating – for fixed wind speed – that

- Of the four input variables, standard deviation of wind speed is the dominating variable and that – to a reasonable degree – the sensitivity coefficient α_{σ_u} is the same under free-flow and wake conditions.

Or, if this is not or only partly the case, that

- The sum of the three rightmost terms in Eq. (4.9) is the same under free-flow or wake conditions.

Thus, the above first bullet states that Eq. (4.9) is valid under both free-flow and wake conditions, and that it is sufficient to assume that only σ_u is different under free-flow and wake conditions.

The second bullet states that should the other flow variables be of importance or should the sensitivity coefficients change from free-flow to wake conditions, they may do so without jeopardising the method if they cancel out, i.e. independently of wind direction

$$\alpha_{\tau}\tau + \alpha_w\tau_w + \alpha_f f \approx \text{constant}. \quad (4.11)$$

Should it happen that Eq. (4.11) does not represent reality sufficiently well, there is a third option: The resulting formula of the method has a structure that allows for adjustments, e.g. by exaggerating σ_u under wake conditions to account for the effect of the other variables.

Below, the justifiability of the above statements is investigated. The sensitivity coefficients are evaluated by means of data, and the effect of the simplification suggested by Eq. (4.11) is tested.

For convenience, the equivalent load $e(U, \theta)$ is simply written as e in the following.

Initially – because of the expected limited wake-related alterations of the time scale of turbulence relative to the free-flow – the scale of turbulence is neglected, reducing Eq. (4.9) to

$$e \approx \alpha_{\sigma_u}\sigma_u + \alpha_{\tau}\tau + \alpha_w\tau_w. \quad (4.12)$$

4.3 Sensitivity coefficients

The sensitivity coefficients are specific to the wind turbine structure. However, the validity of the method does not depend on the absolute magnitude of the sensitivity coefficients, but on the ratios between these.

The sensitivity coefficients may be derived computational or from measurements. The computational method would ensure that many different wind turbine structures could be tested, the drawback being that the state-of-the-art of computer codes would call for experimental verification of the codes themselves.

The experimental approach offers a direct estimation of the sensitivity coefficients, the disadvantages being the limited availability of relevant data and the uncertainties in measurements. An additional problem is that – under wake conditions – it is difficult to measure the flow to which the wind turbine is exposed: the wake develops so if the met mast is placed far from the wake-exposed machine, the measured flow is different from the flow reaching the machine. On the other hand if the met mast is too close to the machine, the test unit itself disturbs the measured flow field. At the Vindeby experiment the problem was dealt with by installation of a tower

(SMS) in a fictitious wind turbine “grid point”, see Figure 1.2. This way, for certain wind directions, the tower is – in a statistical sense – exposed to the same flow as the instrumented wind turbine units. Obviously, the met-mast measurements do not completely reflect what the wind turbine experiences, since the met mast only covers a vertical line. However, the general idea is that the rotor-centre wind turbulence level is an adequate representative of the integrated rotor loading and therefore the met mast’s anemometer at hub height provide the relevant information.

Multivariate regression analysis was performed on the Vindeby data, for loading under both free flow and wake conditions. Data were selected from the wind direction sector 295°-317°, covering free-flow to full-wake conditions. In the regression analyses, the independent, measured variables were

- Standard deviation of turbulent wind speed fluctuations measured at hub height at the southern mast (SMS). Statistically, the standard deviation of turbulent speed fluctuations measured here corresponds to what is experienced by the instrumented 4W wind turbine unit.
- Vertical shear, $\tau = U_{48} - U_{20}$, measured in the western met mast (SMW). This shear is expected to be approx. 15% less than the shear between the uppermost to the lowermost position of the blades (20.5m and 55.5m). However, the load on the outer 1/3 of the blade represents typically 3/4 of the total blade loading, and the chosen measure of shear is therefore assumed to approximate what is effectively experienced by the rotating blade in terms flapwise blade bending.
- Wake deficit, $\tau_w = U_{38,SMW} - U_{38,SMS}$, measured as the difference between hub height wind speed in the two masts: for the selected wind direction range, the wind speed in the western mast represents free flow conditions.

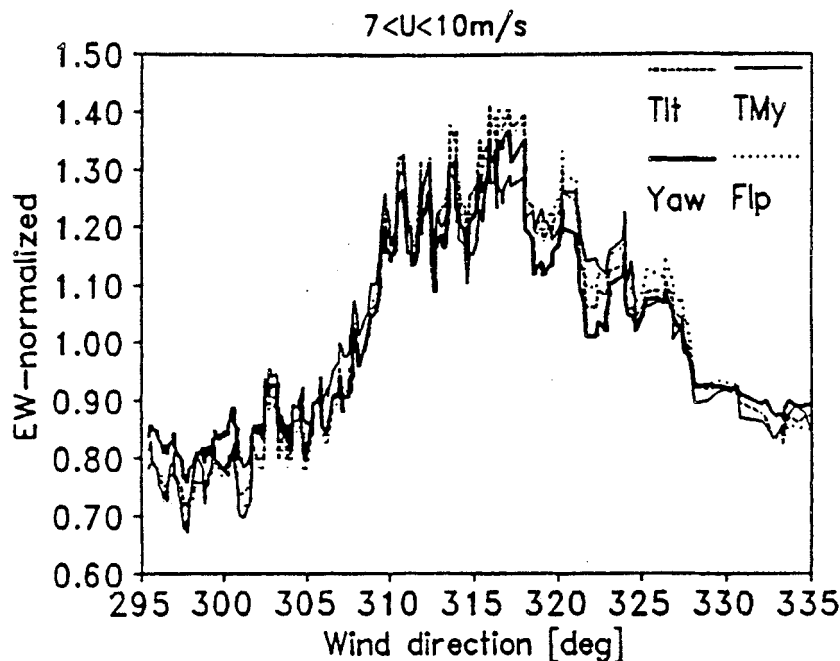


Figure 4.5 Normalised (so mean is equal to 1) equivalent loads for characteristic loads on a wind turbine – along-wind bending of tower (TMy), flap-wise bending of blades (Flp), yaw moment (Yaw) and tilt moment (Tlt). Vindeby Bonus wind turbine – multiple-wake case; turbine separations 8-9 rotor diameters. From Frandsen et al (1996).

The equivalent load of flap-wise blade bending moment on one blade on the 4W unit is selected as the dependent variable. For the wind turbine in question, any of

the main cross-sectional loads could have been chosen, Figure 4.5. The figure shows running mean of measured equivalent loads of blade root, tilt, yaw and tower-base bending moments, all normalised with the average of the considered wind direction range. The data, plotted as function of wind direction, show the effect of varying degrees of wake conditions and it is seen that the four loads for this particular wind turbine respond similarly to the wake environment.

The regression analysis is carried out in 2-degree wind direction bins in the range 295° to 317°, and all data, disregarding mean wind speed, are included. The input data are derived from 30min statistics (mean and standard deviation of the input variables). The sensitivity coefficients as function of wind direction are shown in Figure 4.6 and Figure 4.7. The correlation between the data and the regression result (“multiple-R”) is 0.7-0.9 and the standard error approx. 5kNm, corresponding to 10-20%. The correlation between input variables varies between 0 and 0.5. The low correlation implies that straightforward multivariate regression analysis is applicable. Applying regression on averages over many pre-averaged realisations of the statistics e.g. in wind speed or wind direction bins causes lesser standard error, but less reliable results due to the collinearity imposed by the averaging.

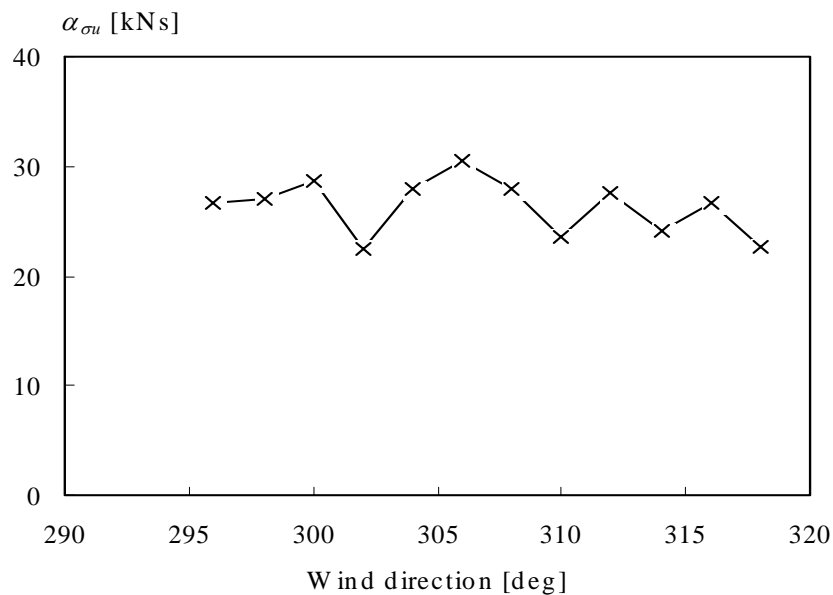


Figure 4.6 Sensitivity coefficient for turbulence, α_{σ_u} , as function of wind direction – to the left free-flow and to the right full wake condition.

The sensitivity to standard deviation of turbulent wind speed fluctuations, Figure 4.6, is insignificantly dependent of wind direction, i.e. not dependent on whether wake conditions are experienced or not. The average is approx. 25 kNs.

Since for non-wake conditions the defined wake deficit is zero, the sensitivity coefficient is here poorly determined. The data analysis, Figure 4.7, displays essentially that, with a variation of α_w of approx. 5 kNs around zero for wind directions 295° to 305°, and a gradual increase to approx. 10 kNs for centre-wake condition at 315°. For illustration of this point, the function $10 \cdot \exp(-((\theta - 315)/8)^2)$ is fitted to the regression result, assuming a maximum value of the sensitivity coefficient at centre-wake condition at 315°.

The sensitivity coefficient corresponding to free-flow vertical shear decreases as wake conditions increase, although there is an increase around 308°-310°, the wind direction where the wake influence becomes significant. The decrease in the sensitivity coefficient under wake condition could be expected since the shear experi-

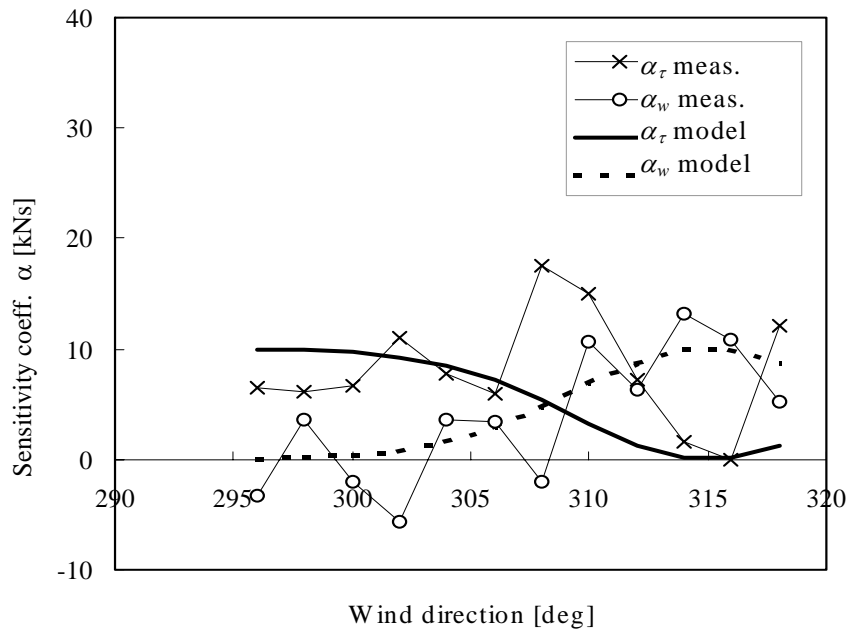


Figure 4.7 Sensitivity coefficients for vertical, free-flow shear α_w and wake deficit, α_τ . Smooth lines are visual fits.

enced by the wind turbine is much distorted relative to the free-flow shear. The function $10 \cdot (1 - \exp(-((\theta - 315)/8)^2))$ was fitted visually.

The ranges of variation of the chosen input variables are approximately the same and therefore the magnitudes of the sensitivity coefficients also indicate the relative load-wise importance of the variables. In turn, this implies that turbulent fluctuations weight approximately three times more than vertical free-flow shear and wake mean deficit.

A demonstration of the regression result (with sensitivity coefficients $\alpha_{\sigma_u} = 25$ kNs, $\alpha_\tau = 10$ kNs, $\alpha_w = 0$ kNs for free-flow conditions and $\alpha_{\sigma_u} = 25$ kNs, $\alpha_\tau = 0$ kNs, $\alpha_w = 10$ kNs for wake condition) is shown in Figure 4.8, where measured, bin-wised averages of loads under free-flow and wake condition, respectively, are shown together with predictions. For input variables, the measured, bin-wise averaged values have been applied. In particular it is noted that the regression analysis captures the drop-off in e for wake conditions at 10-12 m/s and thus shows the tight relationship between response and turbulence also under wake conditions.

When deriving the coefficients conditioned on wind speed, the significance of the regression results improves. However, for the Vindeby wind turbines the equivalent load is by and large a linear function of σ_u for non-wake conditions disregarding the mean wind speed, Frandsen et al (1996). Therefore, performing the regression analysis not conditioned on wind speed demonstrates the robustness of the approach.

Finally, it should be noted that the uncertainty on the estimated sensitivity coefficients is considerable as is evident especially from Figure 4.7.

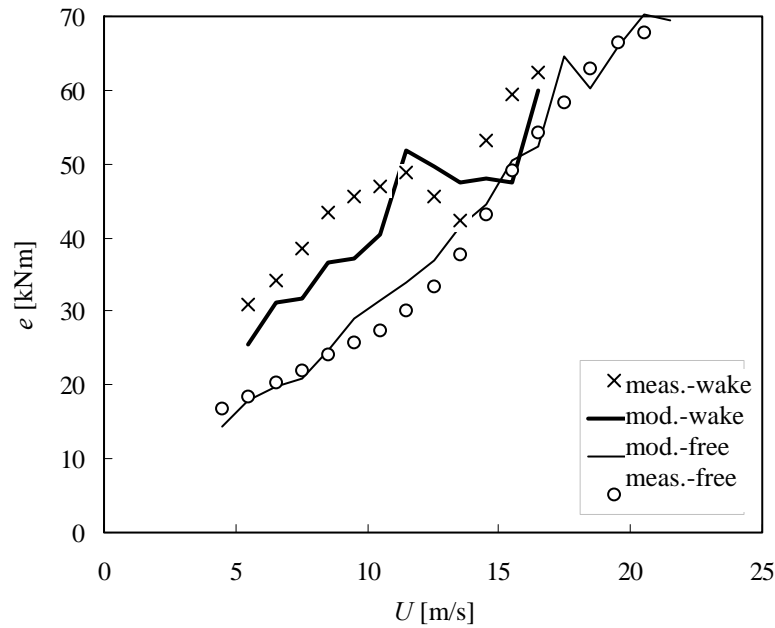


Figure 4.8 Measured bin-wise averaged equivalent loads, flap-wise bending, under free-flow (circles) and under centre-wake conditions (crosses), respectively. The lines are predictions, applying the sensitivity coefficients and measured turbulent fluctuations, free vertical shear and wake deficit. Data from Vindeby.

4.4 Extending measurements

The approach of the method for effective turbulence was tested by means of data from one wind farm, i.e. the important variable *wind turbine separation* is fixed. To be able to cover other separations, it is necessary to extend the measurements by means of analytical models for the input variables.

Thus, in the following the experimentally determined sensitivity coefficients are combined with the models for input variables, given in Section 3. In the *free flow*, the standard deviation of wind speed fluctuations is modelled as

$$\sigma_u = \frac{U}{\ln(z/z_0)}, \quad (4.13)$$

where U is mean wind speed, z is height above ground and z_0 is terrain surface roughness.

For *wake* conditions, the centre-wake, maximum standard deviation of wind speed fluctuations is modelled as

$$\sigma_{u,T} = \sqrt{\sigma_{add}^2 + \sigma_u^2}, \quad \text{where} \quad \sigma_{add} = \frac{U}{1.5 + 0.8 \frac{s}{\sqrt{C_T}}}. \quad (4.14)$$

σ_{add} is maximum added wake turbulence, $s=x/D_0$ is the dimensionless distance to the wake-generating wind turbine, x is the distance between wake-generating and wake affected wind turbine, D_0 is the rotor diameter and C_T is the thrust coefficient of the wake-generating wind turbine.

The models for input variables are compared with the wind speed bin-wise averaged measurements in Figure 4.9, in general showing good agreement. The discrepancy of the model for wake turbulence at high wind speeds is likely due to scarcity of data.

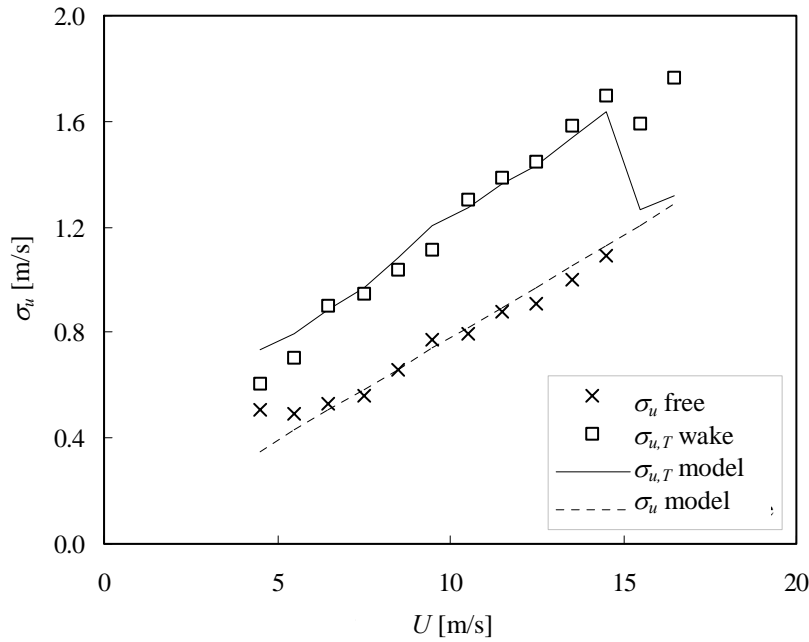


Figure 4.9 Measurements (squares and crosses) and models (full and broken lines) for free-flow and wake standard deviation of wind speed, σ_u . Vindeby data: free-flow measured in western met mast (SMW) and wake measured in southern met mast (SMS), for wind directions 310-320°.

Next, the implication of disregarding the mean wake deficit in the model for effective turbulence is tested. Specifically, this was done by

1. Applying the *experimentally* found sensitivity coefficients, for free-flow and full wake conditions,
2. By applying two different sets of sensitivity coefficients to account for the rather uncertain determination of α_τ and α_w , and by
3. Applying the *models* for free-flow and wake standard deviation of turbulent wind speed fluctuations (Eqs. (4.13) and (4.14)), free-flow vertical shear, (Eq. (2.2) and wake deficit (Eqs. (3.14)/(3.15)).

The two applied sets of the sensitivity coefficients are: [$\alpha_{\sigma_u} = 25\text{kNs}$, $\alpha_\tau = 0\text{kNs}$, $\alpha_w = 10\text{kNs}$] and [$\alpha_{\sigma_u} = 25\text{kNs}$, $\alpha_\tau = 5\text{kNs}$, $\alpha_w = 5\text{kNs}$].

These choices lead to the following equivalent-load predictions under maximum wake conditions:

$$\begin{aligned} e_w &\approx \alpha_{\sigma_u} \sigma_u + \alpha_\tau \tau + \alpha_{\tau_w} \tau_w = 25 \cdot \sigma_u + 5 \cdot \tau + 5 \cdot \tau_w \quad \text{and} \\ e_w &\approx 25 \cdot \sigma_u + 10 \cdot \tau_w \end{aligned} \quad (4.15)$$

In the method for effective turbulence, on the other hand, the presumption is that $\tau_w = 0$ since the wake deficit is not explicitly taken into account. In Figure 4.10 to Figure 4.12, the method (in the considered case $e_w = 25 \cdot \sigma_u$) is compared with the two empirically derived alternatives, Eq. (4.15). In Figure 4.10, the equivalent load

is plotted as function of the dimensionless wind turbine separation. Wind speed is $U = 10\text{m/s}$ and ambient turbulence intensity is $I_0 = 12\%$, and for both cases, the model for effective turbulence slightly under-predicts for wind turbine separations $s = x/D_0$

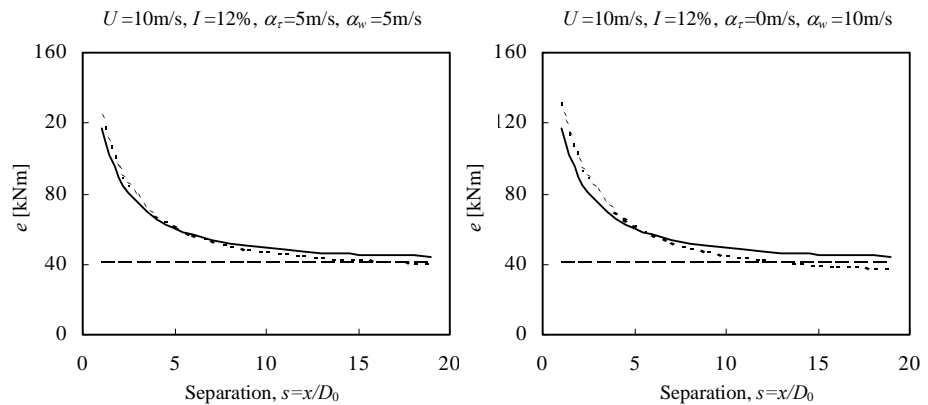


Figure 4.10 Equivalent load as function of wind turbine separation. Full lines are the effective turbulence method, broken lines are the alternative results, derived from Eqs. (4.15), and the horizontal lines are free-flow equivalent loads.

< 5 and over-predicts for $s > 5$.

In Figure 4.11, the sensitivity of equivalent load to ambient standard deviation of turbulent wind speed fluctuations is illustrated. The model for effective turbulence under-predicts for small separations and over-predicts for larger separations.

Finally, Figure 4.12 shows equivalent load for high and low free-flow wind speed.

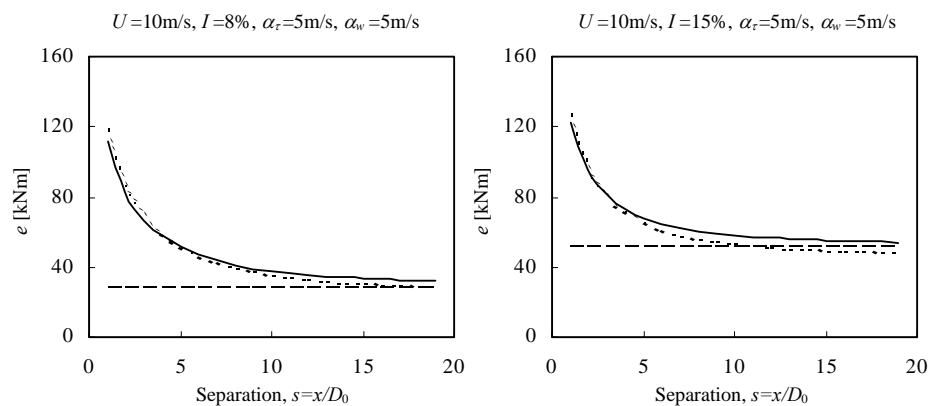


Figure 4.11 Equivalent load as function of wind turbine separation. In both plots, full lines are the effective turbulence method, broken lines are alternatives corresponding to Eq. (4.15). Ambient turbulence intensity is $I_0 = 8\%$ (left) and $I_0 = 15\%$, and the horizontal lines are free-flow equivalent loads.

Again, there is a slight tendency of the model for effective turbulence to under-predict for closely spaced machines and over-predict for larger separations. Comparing Figure 4.10 with Figure 4.12, note the importance of free-flow wind speed – this because, everything else being equal, the turbulence level (σ_u) is high for high wind speeds.

The under-prediction of equivalent load at small separations was foreseen with the result presented in Figure 3.6 that showed the wake-speed deficit expectedly is relatively more important for small separations.

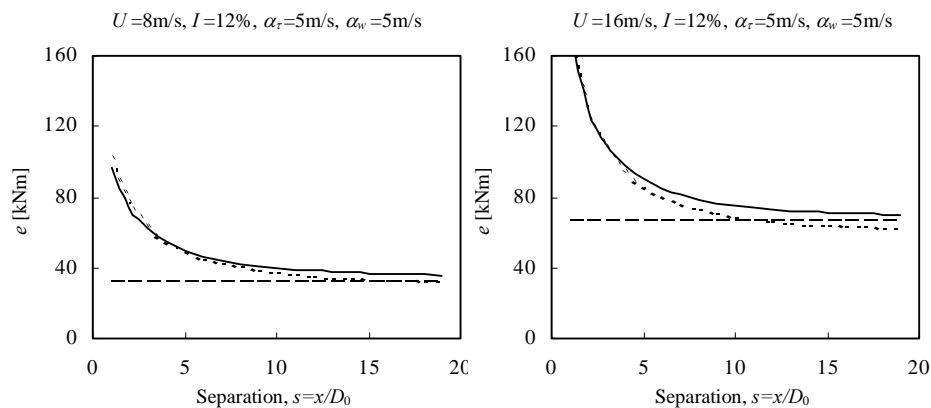


Figure 4.12 *Equivalent load as function of wind turbine separation. In both plots, full lines are the method, and broken lines correspond to the alternative, Eq. (4.15). Free-flow wind speed is $U = 8\text{m/s}$ (left) and $U = 16\text{m/s}$, and the horizontal lines are free-flow equivalent loads.*

4.5 Summary

The validity of the approach in terms of fatigue loading under (full) wake conditions has been assessed, showing that it is feasible, though not perfect.

The model and its approach are evaluated further by means of alternative measurements in Section 7.

5 Combination of fatigue load cases

The model for effective turbulence is conditioned on wind speed, i.e. fatigue load calculations should be carried out separately for a number of wind speed bins, and for each wind speed bin the effective turbulence intensity should be established. Further, it is assumed that the standard deviation of turbulent wind speed fluctuations, $\sigma_u(\theta, U)$, is a deterministic function of wind direction with a superimposed, wind-direction independent random component. Disregarding, by arguments given in Section 4, other flow variables than standard deviation of wind speed fluctuations, Eq. (4.10) reduces to

$$e(U, \theta) = \alpha_{\sigma_u} \cdot \sigma_{u,eff}(U, \theta), \quad \text{where}$$

$$\sigma_{u,eff}(U, \theta) = \left[\int_0^\infty (\alpha_{\sigma_u} \cdot \sigma_u(\theta, U))^m g(\sigma_u|U, \theta) d\sigma_u \right]^{\frac{1}{m}}, \quad (5.1)$$

where m is the Wöhler curve exponent, α_{σ_u} is the sensitivity coefficient for wind speed fluctuations defined in Section 4 and $g(\sigma_u|U, \theta)$ is the PDF of σ_u , conditioned of wind speed and direction. $\sigma_{u,eff}(U, \theta)$ is the fixed standard deviation of wind speed that causes the same fatigue as the varying quantity. Denominating the distribution of wind direction conditioned on wind speed $f_{wd}(\theta|U)$, the integrated equivalent load at wind speed U becomes

$$e_U = \alpha_{\sigma_u} \cdot U \cdot I_{eff}, \quad (5.2)$$

where

$$I_{eff} = \left[\int_{-180}^{180} I^m f_{wd}(\theta|U) d\theta \right]^{\frac{1}{m}}, \quad I = \frac{\sigma_{u,eff}(\theta, U)}{U} \quad (5.3)$$

is the effective turbulence intensity for mean wind speed U .

5.1 Random variation in e in the free flow

For higher wind speeds – which are most relevant for estimation of extreme loading because everything else being equal standard deviation of wind speed fluctuations are high – random variations in turbulence (σ_u) are relatively small. For lower wind speeds, there may be significant variations due to random variability of atmospheric stratification and to nonstationarity in wind speed. This is illustrated by the data in Figure 5.1, where the standard deviation of wind speed σ_u (based on statistics of half-hour time series) is plotted versus wind speed. The plot to the left in Figure 5.1 contains approx. 8000 recorded values in a wind direction sector where the upwind terrain surface is homogeneous (water). The data scatter is large and apparently σ_u reaches levels at $U = 5-10\text{m/s}$ that would only be expected at much higher wind speeds, $U = 20-30\text{m/s}$. Turbulent wind speed fluctuations are site-specific, but the data are nevertheless representative for what is seen at most sites. Though infrequent, the apparently high turbulence levels at low wind speeds are troublesome. If the data are applied directly to compute an m -weighted turbulence intensity for fibreglass, the effective turbulence intensity appears to be 16%, which is twice the value obtained by linear averaging.

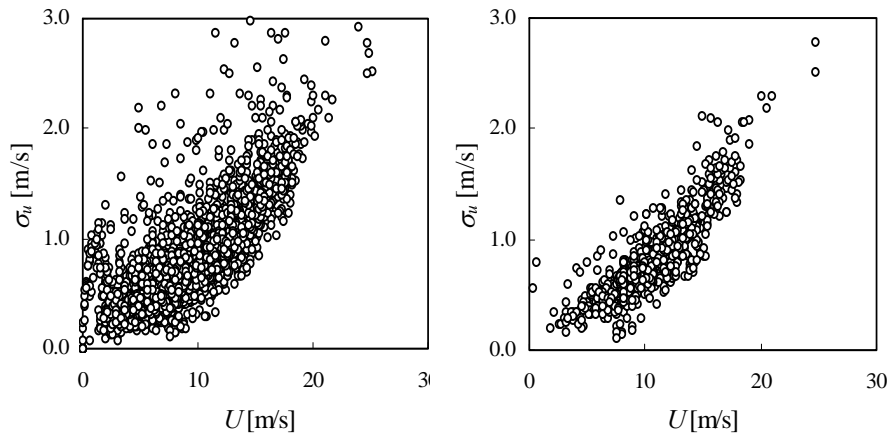


Figure 5.1 *Standard deviation of wind speed – computed for half-hour periods – as function of wind speed. Measurement-height is 38m and wind direction sector 250°-300°, see Figure 1.2. Left plot shows all data and in the right plot the data have been filtered to eliminate instationary time sequences.*

To screen the extent, to which the scatter stems from instationarity, the data were filtered by requiring that the change in mean wind speed around each considered period shall be considerably less than the expected σ_u . Specifically, the selection criterion was

$$U_i \text{ accepted if } \left| \frac{U_{i+1} - U_{i-1}}{U_i} \right| < 0.02, \quad (5.4)$$

where U_{i-1} and U_{i+1} are mean wind speeds the periods immediately before and after the considered period, respectively. The selection criterion eliminates approx. 80% of the data. The data remaining are shown in the right plot of Figure 5.1. The scatter has been reduced significantly and notably the very high values of σ_u at lower wind speeds have vanished all together.

The result of the filtering demonstrates a well-known generic problem in description of turbulence, namely that the much favoured phrase “consider a stationary, random process...” applies to only a fraction of real-world time series of wind speeds under, say, 15-20m/s.

Eliminating the non-stationary time series leaves series that for practical purposes may be considered stationary. In the context of fatigue, the question comes up whether load-sequences, which *do* add significantly to fatigue, are left out if excluding the non-stationary flow conditions. Actually, Larsen and Thomsen (1996) have dealt with the problem from a slightly different view angle by investigating the missed fatigue life consumption caused by neglecting the shift of mean load level from a computer simulation period at one mean wind speed to the next simulation period at a different mean wind speed. It was found that in terms of equivalent load, the error made by neglecting the low-cycle loading is less than 5% for a Wöhler exponent of 7, though depending on the structural component considered.

Further, Kashef and Winterstein (1999) performed multivariate regression analysis on several data sets consisting of damage rates of stresses in a wind turbine component (output variable) and a one-point measurement of wind speed and standard deviation of wind speed fluctuations (input variables). Trying to improve the quality of the results they high-pass filtered the input variables. This rendered the result that

the filtered σ_u considerably better “explained” the damage. This result indicates that the above filtered data (right plot in Figure 5.1) are the more relevant for fatigue loading.

Without drawing a conclusion, it is noted that the problem is potentially significant, but since it is not unique to wake-generated loading, it will not be dealt with in further detail. Therefore, verifying the below model for free-flow effective turbulence, the filtered turbulence data are applied.

From the definition of σ_u , it is clear that it cannot be normally distributed since it is downward limited. It is, nevertheless, a common and practical assumption. Thus, assuming the turbulence intensity, for fixed wind speed, normally distributed and denominating the mean of turbulence intensity, determined from 10min, series I_0 and deviations from the mean i , the fatigue-weighted design turbulence becomes

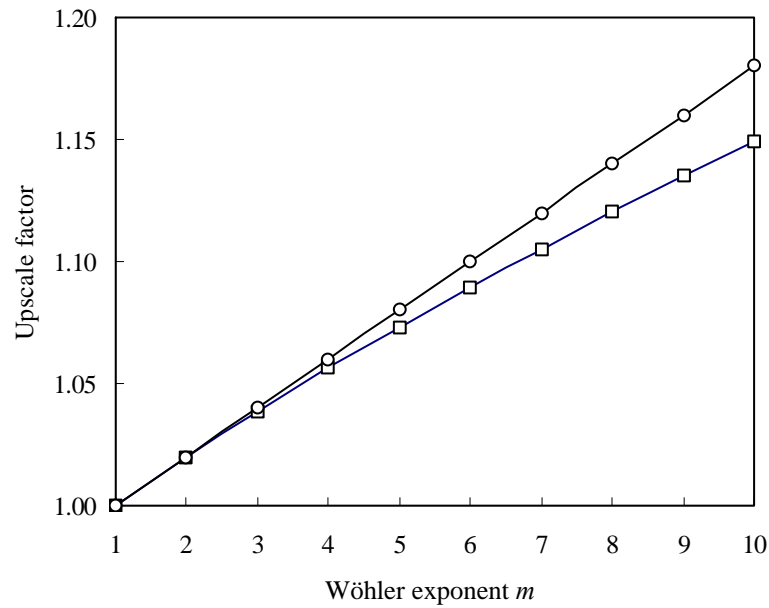


Figure 5.2 Effective turbulence as function of m , for a coefficient of variation of turbulence intensity of 20%. The line with the circles is the approximation.

$$I_m = \left[\int_{-\infty}^{\infty} I^m f(I) dI \right]^{\frac{1}{m}} = \left[\int_{-\infty}^{\infty} (I_0 + i)^m f(i) di \right]^{\frac{1}{m}}, \quad I = I_0 + i, \quad (5.5)$$

where m is the Wöhler curve exponent of the material of the structural component in question, $\bar{I} = I_0$, $\bar{i} = 0$ and f is the PDF of turbulence intensity,

$$f(i) = \frac{1}{\sigma_I \sqrt{2\pi}} \exp\left(-\frac{1}{2} \left(\frac{i}{\sigma_I}\right)^2\right), \quad (5.6)$$

where $\sigma_I = \sqrt{i^2}$ is the standard deviation of the turbulence intensity.

The solution to the integral, Eq. (5.5), is the non-central moment of turbulence intensity,

$$I_m^m = I_0^m \left[1 + \binom{m}{1} \frac{\bar{i}}{I_0} + \binom{m}{2} \frac{\overline{i^2}}{I_0^2} + \dots + \binom{m}{m} \frac{\overline{i^m}}{I_0^m} \right], \quad (5.7)$$

where $\binom{*}{*}$ are the binomial coefficients and m is a non-negative integer. Keeping only the first two terms of the right-side bracket, the following approximation for effective turbulence intensity is obtained:

$$I_m^m \approx I_0^m \left[1 + \frac{1}{2} m(m-1) \left(\frac{\sigma_I}{I_0} \right)^2 \right] \Rightarrow$$

$$I_m \approx I_0 \left(1 + \frac{1}{2} (m-1) \left(\frac{\sigma_I}{I_0} \right)^2 \right). \quad (5.8)$$

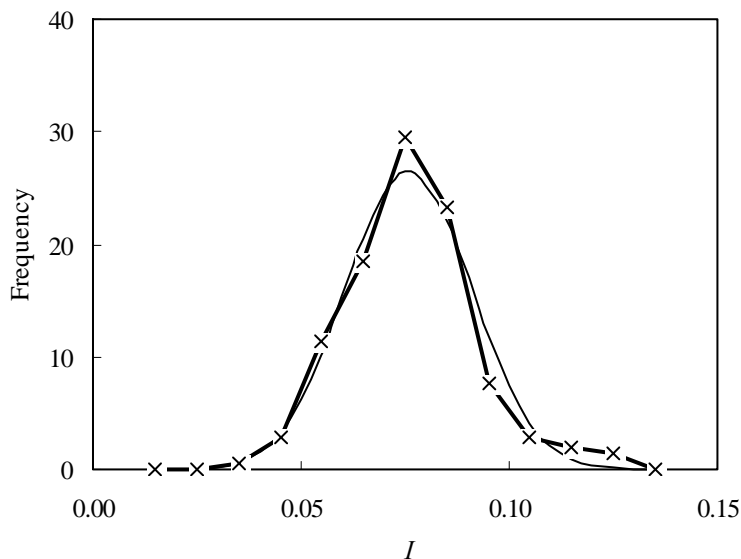


Figure 5.3 Measured PDF of turbulence intensity (line with crosses), $10 < U < 13$ m/s, compared to normal PDF fit.

The approximation, Eq. (5.8), is compared with the non-truncated solution, Eq. (5.7), in Figure 5.2 as a function of the Wöhler exponent, for $\sigma_I/I_0 = 0.2$, chosen as the typical value of the IEC prescription on design turbulence. The approximation is slightly conservative, of the order 5%, for larger values of m .

From the filtered data (right plot of Figure 5.1) for wind speeds between 10 and 13 m/s, the PDF of turbulence was derived and the result is shown in Figure 5.3. Also shown is the normal distribution, fitted to the data. What matters in the context is whether the non-central, higher order moments found directly from the data match what is found by assuming the data are normally distributed. In Figure 5.4, the effective turbulence intensity, computed directly from the data, for Wöhler exponents between 2 and 10, is plotted alongside the prediction of Eq. (5.8). The measurements yield a slightly higher estimate for the free-flow effective turbulence than what is obtained by assuming that the turbulence intensity is normally distributed and applying the approximation Eq. (5.8).

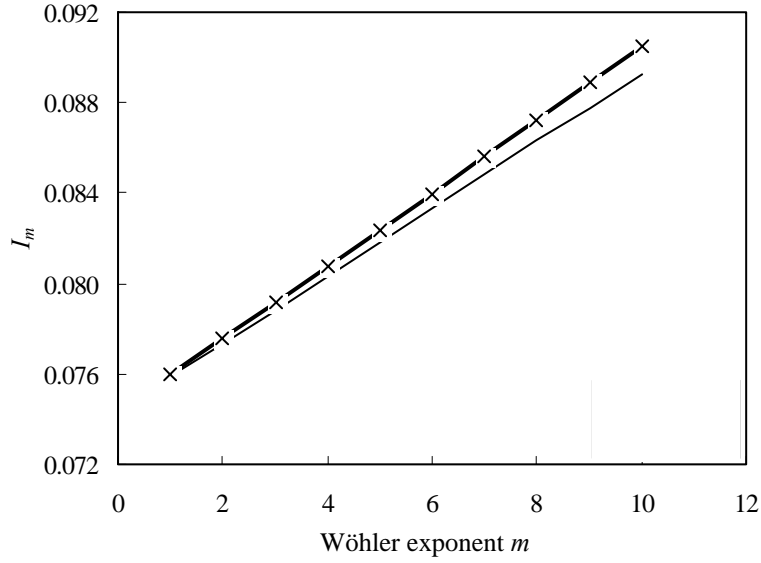


Figure 5.4 *Effective turbulence, derived from measurements (line with crosses are data corresponding to right plot in Figure 5.1) and model, Eq. (5.8).*

For free-flow turbulence, the following is noted:

For $m \leq 10$ and $\sigma_I / I_0 \leq 0.2$, the effective turbulence intensity according to Eq. (5.8) is lower than the percentile value applied in the IEC61400-1 (1999).

For the free-flow turbulence, the addition of one standard deviation already applied in e.g. the IEC-standard covers the random variations in turbulence intensity accounted for by Eq. (5.8).

The random variation in *turbulence under wake* conditions is found to be marginally less than for free-flow conditions, Frandsen et al (1996). The explanation could be that the turbulence created by the wind turbine dominates over ambient turbulence, and that wake-generated turbulence does have lesser variation since presumably it is not affected by variations in atmospheric stability.

Thus, summing up it is found in analogy to Eq. (5.3) that the effective turbulence in the free flow, conditioned on wind speed but not direction, is

$$I_{eff,free} = \left[\int_{-180}^{180} I_m^m(\theta|U) f_{wd}(\theta|U) d\theta \right]^{\frac{1}{m}}, \quad (5.9)$$

where f_{wd} is the PDF of wind direction and I_m is the m -weighted ambient turbulence intensity for wind direction θ , and mean wind speed U .

5.2 Contribution from the wakes

In a cluster of wind turbines each unit is exposed to wake conditions from its neighbour units. Rules of summation of the integrated effects of the wakes that hit one specific wind turbine unit, is sought. Firstly, a two-unit cluster is considered in order to target a suitable method of summation.

One wake, uniform wind direction and ambient distribution

In a cluster of two wind turbines, each unit will experience wake conditions from only one wind direction. The model for load-generating wake turbulence as function of wind direction, given by Eqs. (3.13) and (3.16) to (3.18), is applied to Eq. (5.3). The ambient turbulence intensity is assumed independent of wind direction, which firstly is assumed uniformly distributed. To economize with indices, the random variation in the ambient turbulence – discussed in the previous section – is assumed covered in $I_0(\theta)$. Applying the proposed wake turbulence profile, the effective turbulence conditioned on mean wind speed, but not wind direction becomes

$$I_{eff} = \left(\int_{-180}^{180} f_{wd} I^m d\theta \right)^{\frac{1}{m}} = \left(\int_{-180}^{180} f_{wd} \cdot \left(I_0 \left(1 + \alpha \exp \left[- \left(\frac{\theta}{\theta_w} \right)^2 \right] \right) \right)^m d\theta \right)^{\frac{1}{m}}, \quad (5.10)$$

where for convenience it is assumed that maximum wake turbulence appears at $\theta = 0^\circ$ and where as previously defined

$$\alpha = \sqrt{\left(\frac{I_{add}}{I_0} \right)^2 + 1} - 1. \quad (5.11)$$

The PDF of wind direction was assumed uniform, $f_{wd} = \frac{1}{360} [\text{deg}^{-1}]$, and for the considered case I_0 and α are constants. Since θ_w is small relative to the limits of the definite integral of Eq. (5.10), I_{eff}^m may be approximated as follows:

$$\begin{aligned} I_{eff}^m &= \int_{-180}^{180} f_{wd} \cdot I_0^m \left(1 + \sum_{i=1}^m \binom{m}{i} \alpha^i \cdot \exp \left(-i \left(\frac{\theta}{\theta_w} \right)^2 \right) \right) d\theta \\ &\cong I_0^m \left(1 + \frac{\theta_w}{180} \cdot \frac{\sqrt{\pi}}{2} \cdot \sum_{i=1}^m \binom{m}{i} \cdot \frac{\alpha^i}{\sqrt{i}} \right), \end{aligned} \quad (5.12)$$

where $\binom{m}{i} = \frac{m!}{i!(m-i)!}$ are the binominal coefficients. An obvious model for combining non-wake and wake condition is

$$I_{eff} = \left[\left(1 - b \frac{2\theta_w}{360} \right) I_0^m + b \frac{2\theta_w}{360} I_T^m \right]^{\frac{1}{m}}, \quad (5.13)$$

where $I_T = I_0(1+\alpha)$ is maximum wake turbulence. The quantity $2b\theta_w$ constitutes the “width” of a hypothetical rectangular wake-turbulence profile with amplitude I_T that would cause the same fatigue damage as the more realistic bell-shaped profile. Comparing the two expressions. Eqs. (5.12) and (5.13), it is found that

$$b = \frac{\sqrt{\pi}}{2} \cdot \frac{\sum_{i=1}^m \binom{m}{i} \frac{\alpha^i}{\sqrt{i}}}{\sum_{i=1}^m \binom{m}{i} \alpha^i}. \quad (5.14)$$

Multiple-wake, non-uniform ambient turbulence and wind direction

Consider next a multi-wind turbine cluster, where the considered wind turbine is exposed to wakes from several neighbouring units. Extending the one-wake model

to multiple-wake, applying the same wake shape, the turbulence intensity as function of wind direction becomes

$$I(\theta) = I_0(\theta) \cdot \left(1 + \sum_{j=1}^N \alpha_j \cdot \exp \left(- \left(\frac{\theta - \theta_j}{\theta_{w,j}} \right)^2 \right) \right), \quad (5.15)$$

where N is the number of wakes, $\alpha_j = (I_{T,j} / I_0) - 1$, and $I_{T,j}$ and θ_j are the maximum turbulence intensity and the corresponding azimuth direction of wake no. j , respectively, and $\theta_{w,j}$ is the characteristic width of wake no. j . Now I_0 is a function of wind direction, typically found from measurements. The effective turbulence is hereafter found from

$$I_{eff}^m = \int_{-180}^{180} f_{wd}(\theta) \cdot I_0^m(\theta) \cdot \left(1 + \sum_{j=1}^N \alpha_j \cdot \exp \left(- \left(\frac{\theta - \theta_j}{\theta_{w,j}} \right)^2 \right) \right)^m d\theta. \quad (5.16)$$

Even with reliable models for α and θ_w , further assumptions are needed to reduce Eq. (5.16) since f_{wd} and I_0 are arbitrary experimental functions. Thus, assuming that the wakes overlap only marginally, Eq. (5.16) may be written as

$$I_{eff}^m \cong \int_{-180}^{180} f_{wd}(\theta) \cdot I_0^m d\theta + \sum_{j=1}^N \int_{-180}^{180} f_{wd}(\theta) \cdot I_0^m(\theta) \cdot \sum_{i=1}^m \binom{m}{i} \alpha_j^i \exp \left(- \left(\frac{\theta - \theta_j}{\theta_{w,j} / \sqrt{i}} \right)^2 \right) d\theta, \quad (5.17)$$

Aiming at making the result operational, it is assumed that the PDF of wind direction and the ambient turbulence intensity may be assumed constant in an interval $[\theta_j^-; \theta_j^+]$ around each wake-centre, with values $f_{wd,j}$ and $I_{0,j}$ (in practical terms taken e.g. as the mean of the respective measured values in the interval), and that the interval is large enough to provide a good approximation to the indefinite integral, then Eq. (5.17) becomes

$$I_{eff}^m \cong I_m^m + \sum_{j=1}^N \sum_{i=1}^m f_{wd,j} \cdot I_{0,j}^m \cdot 2\theta_{w,j} \frac{\sqrt{\pi}}{2} \binom{m}{i} \frac{\alpha_j^i}{\sqrt{i}}, \quad (5.18)$$

where $I_m^m = \int_{-180}^{180} f_{wd}(\theta) \cdot I_0^m d\theta$. In analogue to Eq. (5.13), it is seen that the m 'th power of the effective turbulence can be written as

$$I_{eff}^m \cong \left(1 - \sum_{j=1}^N p_{w,j} \left(\frac{I_{0,j}}{I_m} \right)^m \right) I_m^m + \sum_{j=1}^N p_{w,j} \cdot I_{T,j}^m, \quad (5.19)$$

where

$$p_{w,j} = b_j 2\theta_{w,j} f_{wd,j} \text{ and } I_{T,j} = (1 + \alpha_j) I_{0,j}, \quad (5.20)$$

and

$$b_j = \frac{\sqrt{\pi}}{2} \cdot \frac{\sum_{i=1}^m \binom{m}{i} \frac{\alpha_j^i}{\sqrt{i}}}{\sum_{i=1}^m \binom{m}{i} \alpha_j^i}. \quad (5.21)$$

Eq. (5.19) states that the effective turbulence may be expressed as the probability-weighted maximum turbulences of the wakes and the weighted turbulence of the ambient turbulence. The weight of the (fatigue-weighted) ambient turbulence is one minus a fraction that depends on the time spent under wake condition and the ratio of ambient turbulence in the directions of the wakes and I_m taken over the whole wind rose.

Effectively, the probability for wake condition from wake no. j is $p_{w,j} = 2b_j\theta_{w,j}f_{wd,j}$. By testing a range of values of m and α , it turns out that p_w may, without significant loss of accuracy, be approximated as

$$p_{w,j} = 2b_j\theta_{w,j}f_{wd,j} \cong 22 \cdot f_{wd,j} \quad (5.22)$$

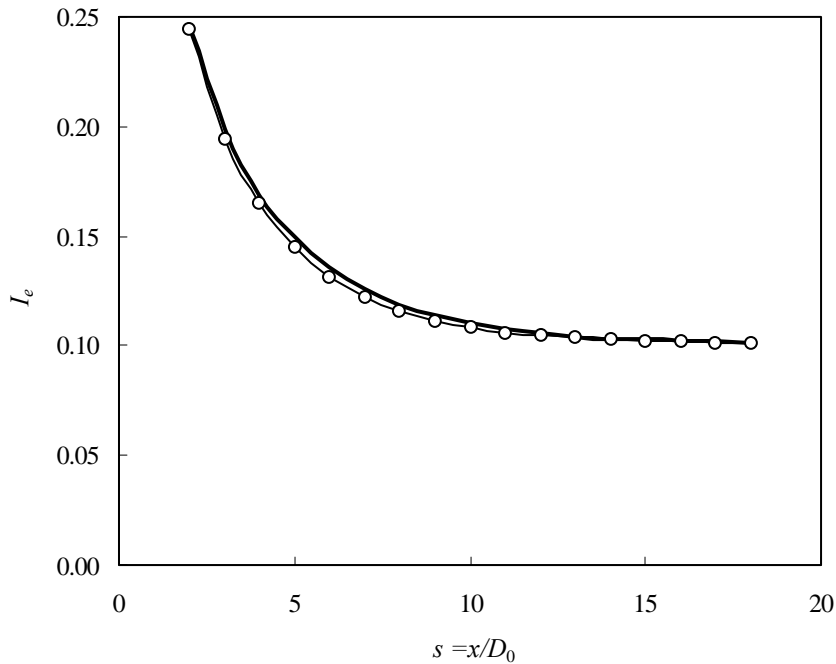


Figure 5.5 Effective turbulence as function of separation between wake-generating and wake-affected units. Ambient turbulence is 10%. The line with the circles and the bold line are the prediction with the width defined by Eq. (5.20) and Eq. (5.22), respectively.

for any value of s , I_0 and m . Apparently, this approximation is crude, claiming the “probability of wake condition” does not depend on the distance to the wake-generating unit. That the approximation is feasible can be understood from Eqs. (5.13)/(5.20). b , θ_w and I_T are functions of low powers of separations, but I_T is raised to the power m , and while θ_w gets larger for decreasing s , b gets smaller. The quality of the approximation is illustrated in Figure 5.5.

Default model

Finally, if the PDF of wind direction is assumed uniform, $f_{wd} = \frac{1}{360} [\text{deg}^{-1}]$, and the ambient turbulence intensity is independent of wind direction, Eq. (5.19) reduces to

$$I_{eff}(U) \cong \left[(1 - N \cdot p_w) I_0^m + \sum_{j=1}^N p_w I_{T,j}^m \right]^{\frac{1}{m}}, \quad (5.23)$$

where I_{T_j} is maximum turbulence intensity for wake number j and N is the number of neighbouring wind turbines taken into account and $p_w \approx 0.06$.

In the final model proposed in Section 8, the number of neighbouring wind turbines to consider is limited to 8 – the nearest few units dominate the aggregated load effect and the practical limitation does not significantly influence the result. The wind turbine units will typically be arranged to produce as little mutual wake effects as possible in order to minimize array production losses. Therefore the assumption of uniform distribution of wind direction – and thus Eq. (5.23) – tends to be conservative.

The validity of the approach as such – altering the load instead of the load effect – is not dependent on linearity between the equivalent load and standard deviation of wind speed fluctuations. However, the specific proposals, Eqs. (5.19) and (5.23), were developed under the assumption of linearity. Therefore, should a concrete design display strong non-linearity in σ_u , the specific values proposed here should be used with some caution.

For structural loads, where the mean is not insignificant and/or different under wake and non-conditions, the influence of the mean should be taken into account. The Goodman approach, which implies an alteration of the constant in the Wöhler-curve for the material in question, may be applied:

$$n_{fat} = a \cdot s_{fat}^{-m} \Leftrightarrow s_{fat} = a^{1/m} \cdot n_{fat}^{-1/m} \quad (5.24)$$

where

$$a = a_0 \left(1 - \frac{s_a}{s_u} \right)^m \quad (5.25)$$

where n_{fat} is the number of cycles to fatigue with stress range s_{fat} , s_a is the mean stress and s_u is the ultimate tensile strength. a_0 is the constant corresponding to zero mean stress. If s_a and thus a are independent of wind direction (wake or non-wake condition), then a does not enter into the derivation of the expressions for effective turbulence. If s_a varies significantly with wind direction, then it must be taken into account in the calculation of the effective turbulence.

6 Combination of extreme load cases

Though being more complicated than that, fatigue life consumption has – so to speak – the nature of averaging: all events are counted to provide an estimate of the lifetime of the structure. Obviously, ultimate loads happen extremely seldom and supporting experimental evidence will therefore be much weaker for extreme load models than for fatigue load models.

6.1 General

For practical evaluation of both fatigue and ultimate load, present-day methods include time-simulation and subsequent analysis of the response time series. For fatigue, rain-flow counting or the like is performed on time series for the design load

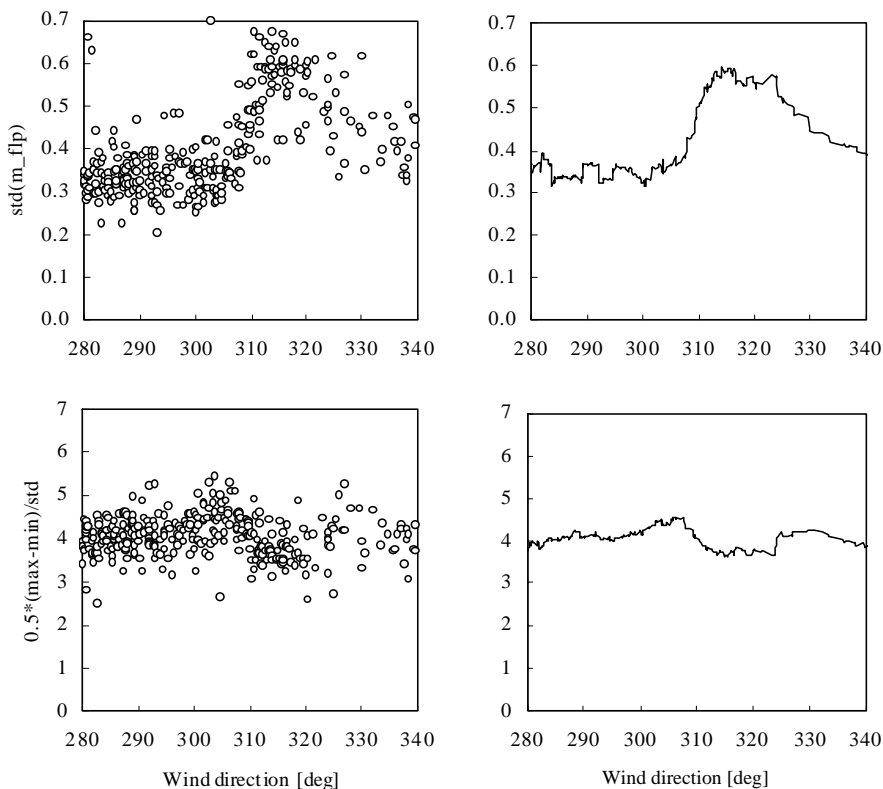


Figure 6.1 Standard deviation of flap-wise blade bending moment (upper plots, scale arbitrary) and load peak factor, as function of wind direction. Left plots are 30min statistics and right plots are running averages over 19 half-hour values. $8m/s < U < 10m/s$. From Vindeby measurements.

cases and the resulting load spectra are *added* to form the combined load spectrum. For ultimate loads, the rational approach is to establish the extreme *distribution* for each load case by means of stochastic simulations, then combine these to form the overall extreme distribution and subsequently derive the lifetime extreme from the overall distribution¹⁴. If ultimate load occurs under extreme wind conditions, no

¹⁴ The upcoming ed. 3 of the design standard IEC61400-1 applies a mixture of this approach and simply deriving the ultimate load as the largest of the extremes obtained for each load case. The latter method is strictly speaking not correct.

problems particular to the wind farm environment are encountered since the wind turbines will be parked and turbulence will deviate only marginally from free-flow conditions¹⁵.

However, should ultimate load occur during normal operation, the particulars of the wind farm flow must be taken into account. An analogy to extreme loads generated by wakes is the combination of deterministic gravity loading and stochastic turbulence loads on wind turbine blades, Madsen and Frandsen (1984b) and Madsen et al (1999).

Firstly, by means of measurements, the free-flow and wake-flow extreme response in terms of expectance value of the largest extreme is probed. The peak load factor is defined by the following expression for the extreme load as function of mean M_{mean} and standard deviation σ_M of the load,

$$E\{M_{max}\} = M_{mean} + k_p \cdot \sigma_M \quad (6.1)$$

Due to measurement uncertainties, estimation of M_{mean} is not reliable and therefore it is assumed that the PDF of response is symmetrical, i.e. $E\{M_{max} - M_{mean}\} = E\{\frac{1}{2}(M_{max} - M_{min})\}$, where M_{min} and M_{max} are minimum and maximum for the considered period, respectively. Thus, the peak factor is estimated

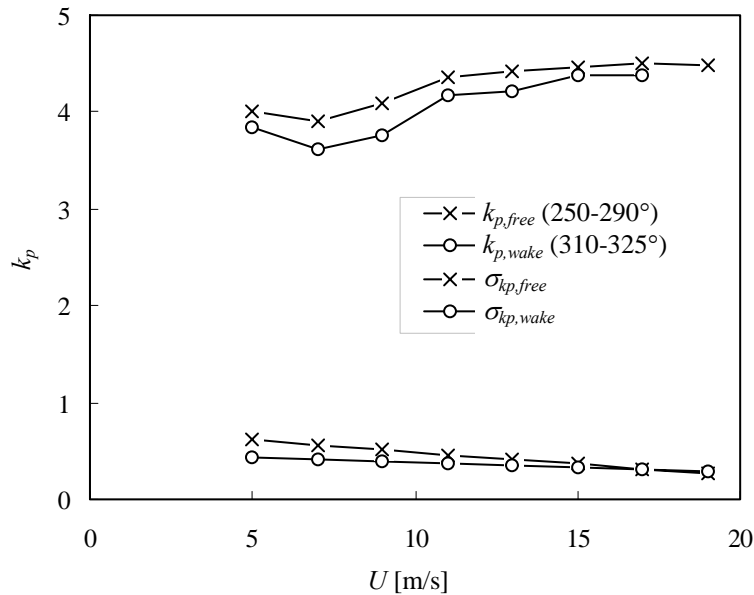


Figure 6.2 Estimate of peak load factor, taken over 30min periods, as function of wind speed for free-flow and wake conditions. The lower curves are standard deviations of k_p . From Vindeby measurements.

from measurements as:

$$\hat{k}_p = \frac{1}{2} \frac{M_{max} - M_{min}}{\sigma_M} \quad (6.2)$$

Figure 6.1 shows standard deviation of flap-wise blade bending moment and the peak load factor, respectively, as function of wind direction. It is seen that there is a

¹⁵ Applying the model for effective turbulence, with adjusted wake turbulence model, on the parked wind turbines leads for typical wind farm configurations to an increase in “ambient” turbulence, relative to free-flow turbulence, of 1-4% in absolute terms.

significant increase in the load standard deviation in the wind direction with wake conditions, 310-325°, see Figure 1.2. The peak factor (lower plots) is of the order 4, both under free-flow and wake conditions. Only in the “running average” plot, a change in k_p – when entering in and out of the wake – can be detected.

In Figure 6.2, k_p is plotted as function of wind speed for free-flow and centre-wake conditions, respectively. The estimates, derived from wind speed bins of 2 m/s, increase approx. 20% from low to high wind speeds and deviate marginally from each other. Further, it is seen that the standard deviation of k_p is marginally less for wake than non-wake conditions.

When adding fatigue life consumption from different load cases, it was utilised that the mean response most often plays an insignificant role. A similar specific approach cannot directly be applied for identification of lifetime extremes. However, it is possible to suggest workable solutions for the aggregated response for wake and non-wake conditions, the aim still being to significantly reduce the numerous response simulation runs, which are otherwise necessary.

The extreme response of the operating wind turbine depends on a number of variables describing the air flow. Let the distribution of response extremes, conditioned on the variables x_i , be denominated $F_y(y|x_1, x_2, \dots)$. Denominating the joint probability density function of the flow variables $g(x_1, x_2, \dots)$, the unconditional distribution of extremes can be expressed as

$$F_T(y) = \iint \dots \int F_y(y|x_1, x_2, \dots)g(x_1, x_2, \dots)dx_1 dx_2 \dots \quad (6.3)$$

If y represents the largest extremes in 10min periods, the expectance value of the unconditional, largest extreme y_M over $10 \cdot M$ minutes is in principle found by solving the following equation with respect to y_M :

$$F_T(y_M) = 1 - \frac{1}{M}, \quad (6.4)$$

where M is the number of 10min periods with the wind turbine in operation in its lifetime.

Primary assumptions and simplifications

For fatigue, the dominating flow variables are mean wind speed and standard deviation of wind speed fluctuations, and basically this holds for extreme response. Thus, it is assumed that the standard deviation of response is proportional to standard deviation of wind speed fluctuations, $\sigma_y \propto \sigma_u$, for wake and non-wake conditions.

Further, it is assumed that – at least in a statistical sense – extreme response is a function of U and σ_u , and that σ_u is a deterministic function of mean wind speed U and wind direction θ , $\sigma_u = \sigma_u(U, \theta)$. Therefore, disregarding other flow variables than U and σ_u , the unconditional extreme distribution of the operating wind turbine reduces to

$$F_T(y) = \int_{-180}^{180} \int_0^{\infty} F_y(y|U, \sigma_u(U, \theta))g(U, \theta)dUd\theta. \quad (6.5)$$

Let the wind direction be uniformly distributed, $f_{wd}(\theta) = \frac{1}{360}$ and the mean wind speed be Weibull distributed and independent of wind direction, with the probability density function $f_w(U)$. With this, the unconditional extreme distribution of response has conveniently been reduced to

$$F_T(y) = \int_0^{\infty} F_U(y|U) f_w(U) dU, \quad (6.6)$$

where the distribution conditioned on wind speed, but not wind direction, is

$$F_U(y|U) = \frac{1}{360} \int_{-180}^{180} F_y(y|U, \theta) d\theta. \quad (6.7)$$

6.2 Combined distribution

Let then the standard deviation of wind speed during non-wake condition be constant, $\sigma_u(\theta) = \sigma_0$, and let wake turbulence be represented by a rectangular wake with constant standard deviation of wind speed, $\sigma_u(\theta) = \sigma_w$, in the angular space $2\theta_w$, θ_w being defined by Eq. (3.18):

$$F_y(y|U, \sigma_u) = \begin{cases} F_0(y|U, \sigma_0) & \text{for } \theta < -\theta_w \text{ and } \theta > \theta_w \\ F_w(y|U, \sigma_w) & \text{for } -\theta_w < \theta < \theta_w \end{cases}. \quad (6.8)$$

With this idealisation, the distribution of extremes, conditioned on mean wind speed, becomes

$$F_U = p_0 \cdot F_0 + p_w \cdot F_w, \quad (6.9)$$

where

$$p_0 = 1 - p_w \text{ and } p_w = \frac{\theta_w}{180} \quad (6.10)$$

is the probability of non-wake and wake conditions, respectively. Note that the "probability of wake condition", p_w , is different from the definition applied in the fatigue context.

Asymptotic combined distribution

Eq. (6.9) may be rewritten:

$$\begin{aligned} F_U &= p_0 \cdot F_0 + p_w \cdot F_w \\ &= F_w \left(p_w + p_0 \frac{F_0}{F_w} \right). \end{aligned} \quad (6.11)$$

For large y , both $F_0 = 1 - \delta_0$ and $F_w = 1 - \delta_w$ are close to unity and assuming wake response "considerably" larger than non-wake response, then

$$\frac{\delta_w}{\delta_0} \gg 1 \quad (6.12)$$

and therefore

$$\frac{F_0}{F_w} \approx \frac{1}{F_w} \quad (6.13)$$

Inserting Eq. (6.13) into Eq. (6.11) yields

$$F_U = F_w \left(p_w + p_0 \frac{F_0}{F_w} \right) \approx F_w \left(p_w + p_0 \frac{1}{F_w} \right) = F_w \left(1 + p_0 \left\{ \frac{1}{F_w} - 1 \right\} \right) \quad (6.14)$$

Taking the logarithm on both sides of the equation and applying Taylor expansion, including the first two terms, yields

$$\begin{aligned}
\ln F_U &= \ln F_w + \ln \left(1 + p_0 \left\{ \frac{1}{F_w} - 1 \right\} \right) \approx \ln F_w + p_0 \left(\frac{1}{F_w} - 1 \right) \\
&= \ln F_w \left(1 + \frac{p_0}{\ln F_w} \cdot \frac{1 - F_w}{F_w} \right) = \ln F_w \left(1 + \frac{p_0}{\ln(1 + F_w - 1)} \cdot \frac{1 - F_w}{F_w} \right) \\
&\approx \ln F_w \left(1 + \frac{p_0}{F_w - 1} \cdot \frac{1 - F_w}{F_w} \right) = \ln F_w \left(1 - \frac{p_0}{F_w} \right) \approx (1 - p_0) \ln F_w \Rightarrow \\
\ln F_U &\approx \ln(F_w^{(1-p_0)}) = \ln(F_w^{p_w}) \Rightarrow
\end{aligned}$$

$$F_U \approx F_{U,as} = F_w^{p_w} \quad (6.15)$$

Thus, when extremes are significantly larger during one period than during another period, then asymptotically for increasingly large response, the extreme response distribution for the aggregated period only depends on the extreme distribution for the period with larger response and the relative probability of that period.

To proceed, it is assumed that F_w is of type EV I (Gumbel):

$$\begin{aligned}
F_{U,as}(y|U) &= (\exp(-\exp(-\alpha_w(y - \beta_w))))^{p_w} \\
&= \exp(-\exp(-\alpha_w(y - \beta_w) + \ln p_w)) \\
&= \exp\left(-\exp\left(-\alpha_w\left(y - \beta_w - \frac{\ln p_w}{\alpha_w}\right)\right)\right)
\end{aligned} \quad (6.16)$$

where α_w and β_w are scale and location parameters, respectively. Thus, asymptotically for increasing y , the extreme response distribution, conditioned on wind speed but not wind direction, is also of type Gumbel, with the parameters

$$\alpha_U = \alpha_w \text{ and } \beta_U = \beta_w + \frac{\ln p_w}{\alpha_w}. \quad (6.17)$$

The result is illustrated in Figure 6.3. For the chosen example with zero mean response for both free- and wake flow and standard deviation 1 and 3 for free- and wake flow, respectively, the asymptotic solution becomes a good approximation for

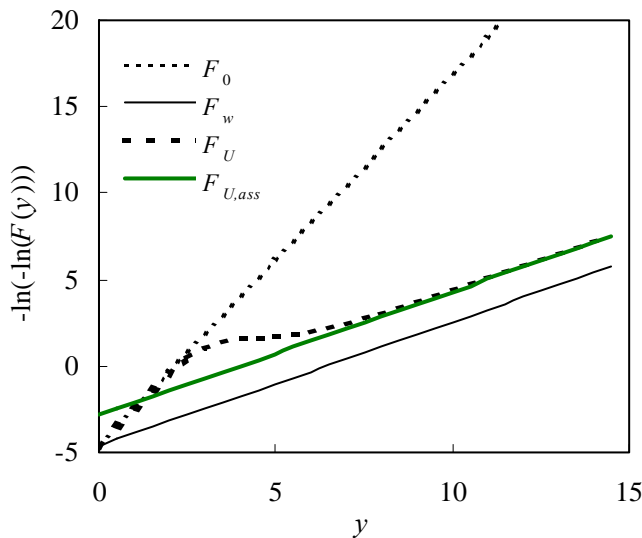


Figure 6.3 Extreme distributions for free flow (F_0), wake flow (F_w), combined (F_U) and asymptotic ($F_{U,as}$). Mean is 0, and variance 1 and $p_0=0.87$ for free flow and variance 3 and $p_w=0.13$ for wake flow.

extreme responses larger than 5-6 wake-flow standard deviations.

Extreme response conditioned on wind speed

By means of the asymptotic distribution, it is now possible to produce an estimate of the extreme response for a wind turbine exposed to one wake from a neighbouring machine and otherwise undisturbed flow. Assume that for a given, not too wide, wind speed bin there is a total of M_i 10min periods and that the parameters of the extreme distribution α_w and β_w have been determined through computer simulations with maximum wake turbulence. The best estimate of the conditional extreme response y_i is found from Eq. (6.4):

$$F_{U,as}(y_i|U) = 1 - \frac{1}{M_i} \Rightarrow y_i \approx \frac{\ln(M_i)}{\alpha_U} + \beta_U = \frac{\ln(p_w M_i)}{\alpha_w} + \beta_w \quad (6.18)$$

where p_w is determined from Eq. (6.10). Therefore, the overall extreme in a wind speed bin is found by considering only the number of 10min periods ($p_w M_i$) where there are wake conditions. If there are wakes from N neighbouring wind turbines producing approximately the same wake effects in terms of standard deviation of response, Eq. (6.18) is adjusted to account for that by replacing $p_w M_i$ by $p_w N M_i$.

The conditional extreme of Eq. (6.18) will also be a good approximation to the extreme not conditioned on wind speed in case the extreme distribution of the considered bin is ‘‘considerably’’ less steep than the distributions of the neighbouring wind speed bins.

A shortcut to the extreme distribution not conditioned on wind direction

To survey whether it is possible in the context of extreme response to find a shortcut, let us first consider the extreme value distribution of the normalised along-wind wind speed fluctuations,

$$v = \frac{u - U_0}{\sigma_u}, \quad (6.19)$$

where u is wind speed, U_0 is mean wind speed and σ_u is standard deviation of fluctuations. If v is narrow-band normally distributed, the (10min) extremes are Gumbel distributed,

$$F_v(v) = \exp(-\exp(-\alpha_0(v - \beta_0))), \quad (6.20)$$

The expectance of the 10min extreme is

$$E_{10\min}\{v\} = \beta_0 + \frac{\gamma_E}{\alpha_0}, \quad (6.21)$$

where $\gamma_E \approx 0.577$ is Euler’s constant. Alternatively, Cartwright and Longuet-Higgins (1956) and Davenport (1961) estimate the largest extreme over a period with K up crossings of zero,

$$v_K = \sqrt{2\ln(K)} + \frac{\gamma_E}{\sqrt{2\ln(K)}}. \quad (6.22)$$

Thus, if K up crossings take place in the 10min period, then $\alpha_0 = \beta_0 = \sqrt{2\ln(K)}$. Wind speed is not a narrow band process. However, Eq. (6.22) is nevertheless a fair approximation. From wind speed data, v_K is found to be approximately three, corresponding to $K \approx 50$ upcrossings/extreme events per 10min. In turn, estimates of the scale and location parameters of Eq. (6.21) are obtained, $\alpha_0 \approx \beta_0 \approx 2.8$.

By a crude, but in the context adequate, assumption of linearity between response and wind speed fluctuations, we may write

$$y = y_0 + c_u \sigma_u v \Rightarrow v = \frac{y - y_0}{c_u \sigma_u}, \quad (6.23)$$

where y_0 is mean response and c_u is a proportionality factor between wind speed and response fluctuations. Inserting this expression for v in Eq. (6.20) leads to the following distribution of response, conditioned on mean wind speed:

$$F_U(y|U) = \exp\left(-\exp\left(-\alpha_0\left(\frac{y - y_0}{c_u \sigma_u} - \beta_0\right)\right)\right) \Rightarrow, \quad (6.24)$$

$$F_U(y|U) = \exp(-\exp(-\alpha_U(y - \beta_U)))$$

where

$$\alpha_U = \frac{\alpha_0}{c_u \sigma_u} \quad \text{and} \quad \beta_U = c_u \sigma_u \beta_0 + y_0. \quad (6.25)$$

The parameters for the asymptotic distribution of the combination of wake and free flow conditions, Eqs. (6.17) and (6.25), become

$$\alpha_{U,as} = \alpha_U = \frac{\alpha_0}{c_u \sigma_w} = \frac{\alpha_0}{\sigma_y} \quad \text{and}$$

$$\beta_{U,as} = \beta_U + \frac{\ln(p_w)}{\alpha_U} = c_u \sigma_w \beta_0 + c_u \sigma_w \frac{\ln(p_w)}{\alpha_0} + y_0 \quad (6.26)$$

$$= \sigma_y \beta_0 + y_0 + \sigma_y \frac{\ln(p_w)}{\alpha_0}$$

where here σ_y is standard deviation of response with full wake turbulence, σ_w . What distinguishes these parameters for the combined extreme response distribution from the parameters corresponding to full wake response is the term $\sigma_y \frac{\ln(p_w)}{\alpha_0}$. The

term could be taken as an adjustment to the mean response to account for not having full wake conditions all the time,

$$y_{0,w} = y_0 + \sigma_y \frac{\ln(p_w)}{\alpha_0} \quad (6.27)$$

With the value for α_0 deduced above and for typical values of p_w , the adjustment term is of order $-\sigma_y$. For several wakes, the correction becomes less.

Thus, for the extreme distribution of response, conditioned on wind speed but not wind direction, the scale and location parameters can be derived. The proposed procedure is

1. Determine y_0 computationally
2. Calculate the adjustment to y_0 , Eq. (6.27)
3. Determine the proportionality constant between standard deviation of response and wind speed, c_u , by computational analysis perturbed around relevant values of σ_u (σ_w)
4. Derive the overall scale and location parameters for the distribution, Eq. (6.26).

Effective turbulence for extreme loads?

For fatigue load calculations it was meaningful and useful to define an effective turbulence intensity that causes the same fatigue as the physically appearing real turbulence intensities. For the extreme, conditioned on wind speed but not direction,

an effective turbulence of a kind can be determined. For a wind speed bin with M_i 10min extremes, the best estimate of the largest of these is

$$\begin{aligned}
 E\{y_i\} &= \frac{\ln(M_i)}{\alpha_{U,as}} + \beta_{U,as} \\
 &= \frac{\ln(M_i)}{\alpha_0} + c_u \sigma_w \beta_0 + \frac{c_u \sigma_w}{\alpha_0} \ln(p_w) + y_0 \\
 &\quad \frac{c_u \sigma_w}{\alpha_0} \\
 &= c_u \sigma_{ext} + y_0
 \end{aligned} \tag{6.28}$$

where

$$\sigma_{ext} = \sigma_w \left(\frac{\ln(M_i p_w)}{\alpha_0} + \beta_0 \right) = \sigma_w \cdot k_{ext} \tag{6.29}$$

could be taken as an equivalent measure of turbulence. Thus, the extreme within one wind speed bin, may be found by first calculating σ_{ext} for the respective bin, then use σ_{ext} as input to a computer code which in turn produces as output the mean response y_0 and standard deviation of response, $\alpha_y (= c_u \sigma_{ext})$. The sum of y_0 and α_y is the expectance value of the largest response in the considered wind speed bin during the wind turbine's lifetime. The factor k_{ext} is defined similarly to the previously defined k_p and takes values up to 10, i.e. 2-3 times typical values of k_p for expectation value of the largest extreme wind gust or response in a 10min period.

Apart from the strong requirement of linearity between response and wind speed, the concept has the conceptual weaknesses that k_{ext} depends on bin width and that it only produces a conditional extreme. For the strict derivation of the unconditional extreme, the conditional *distributions*, and not only the conditional extremes, must be known.

6.3 Overall distribution

The unconditional extreme distribution

The unconditional extreme distribution of the operating machine is found as

$$F_T(y) = \sum_i p_i F_{U,i}(y|U_i, \alpha_{U,i}, \beta_{U,i}), \tag{6.30}$$

where p_i is the probability of wind speed bin no. i . $F_{U,i}$ may be replaced with the asymptotic solutions $F_{U,as,i}$ with the parameters $\alpha_{U,as,i}$ and $\beta_{U,as,i}$ of Eq. (6.26). With the component distribution being of type EV 1, F_T will asymptotically be Gumbel distributed and its parameters can be found by means of regression analysis on the tail of the above sum of distributions.

Normality of the mother distributions

Normality of the mother distributions of wind speed and response fluctuations is the basis for the above derivations. The proposed procedure can also be applied for non-normal distributions of local extremes as long as these are of exponential type, though the parameters α_0 and β_0 must be derived by other means than proposed.

7 Verification

Apart from the Vindeby measurements, the data material to verify the model is rather scarce, either not available at all or if available, then frequently presented in a form not compatible to the present analysis. From what *is* available, comparisons are made with the predictions of the model. Doing so, the focus is on testing how well the wake response follow the proposed wake turbulence model. A critical point in verifying the model is to establish the ratio between load effect and standard deviation of wind speed fluctuations in the free flow. Except for the Vindeby and Middelgrunden data (Subsections 7.1 and 7.2), where the ratio was estimated from the measured data, the ratio was estimated by an *assumed* free-flow turbulence intensity together with the measured dynamic response.

The comparison with data is followed by comparison of the model with the recommendations previously used, Teknisk Grundlag (1992), of the Danish wind turbine type approval system. Finally, the uncertainty of the model is assessed.

7.1 Vindeby

The data from the Vindeby Wind Farm have been utilised throughout the report for verification in the case of fairly large separations of the wind turbines. While for that analysis response of the 4W unit was utilised, we here compare simultaneous measurement from both units 4W and 5E, see Figure 1.2.

In Figure 7.1, equivalent loads determined from measurements are plotted as function of wind direction. The wind speed window is 8-9m/s, i.e. the thrust coefficient is high. The crosses are measured equivalent loads on the unobstructed unit 4W and the circles are loads on unit 5E, located downwind of 4W for wind directions 240-265°. The line is the model prediction of equivalent load. It is seen that the data on average fit the model rather well, indicating that no safety margin is left at lower wind speeds.

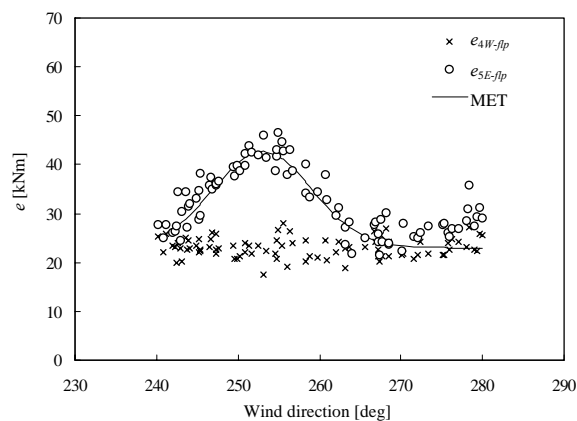


Figure 7.1 *Equivalent loads, flap-wise blade bending, based on 30min time series. Wind speeds 8-9 m/s, estimated $C_T=0.88$, separation is $8.5D_0$, $I_0=0.077$. Vindeby data.*

The same comparison of the model with data is given in Figure 7.2, this time for wind speeds between 13 and 14m/s. Here, the model somewhat over-predicts the equivalent load under wake conditions. For wind speeds in excess of 15 m/s, see Figure 4.8, the equivalent load under wake conditions approaches free-stream load, which is consistent with the results for closely spaced machines, see below.

A possible explanation is that the design turbulence model's asymptotic dependency of the thrust coefficient

($I_{add} \propto C_T^{1/2} s^{-1}$), should be stronger, i.e. the exponent should be higher than $\frac{1}{2}$, see Section 3. It is further noticed that also here the scatter of equivalent load, and thus the coefficient of variation of the quantity, appears less for wake conditions than for unobstructed load conditions.

Even though the wake effect relative to free-flow conditions is less for higher wind speeds, the absolute level is 20-30% higher in Figure 7.2 than Figure 7.1.

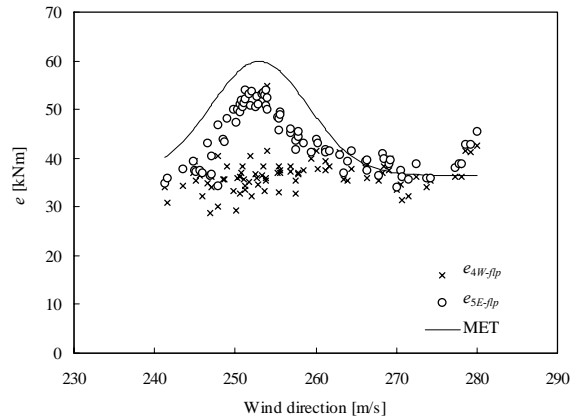


Figure 7.2 Equivalent loads, flap-wise blade bending, based on 30min time series. Wind speeds 13-14 m/s, thrust coefficient estimated at $C_T=0.64$, separation is $8.5D_0$, $I_0=0.075$. Vindeby data.

7.2 Middelgrunden

The Middelgrunden offshore wind farm is located in Øresund east of the city of Copenhagen and consists of 20 Bonus 2MW units, placed in a large-radius arc as shown in Figure 7.3. The measurement programme was initially reported by Enevoldsen and Stiesdal (2001). The active-stall controlled rotor has a diameter of 76m and hub height is 64m. The induction generator operates around two rotational speeds, 1000 and 1500 rpm. The separation of the wind turbines is 182m, corresponding to 2.4 rotor diameters. The small spacing is of particular importance for evaluating the assumption that increasing the nominal maximum wake turbulence intensity beyond its physical value may account for an increasing effect of semi-wake loading, as discussed in Section 3.4. The effect of partial wake would be expected to be significant for flapwise blade bending and yaw moment (torsion of tower) and less significant for e.g. tower overturning moment.

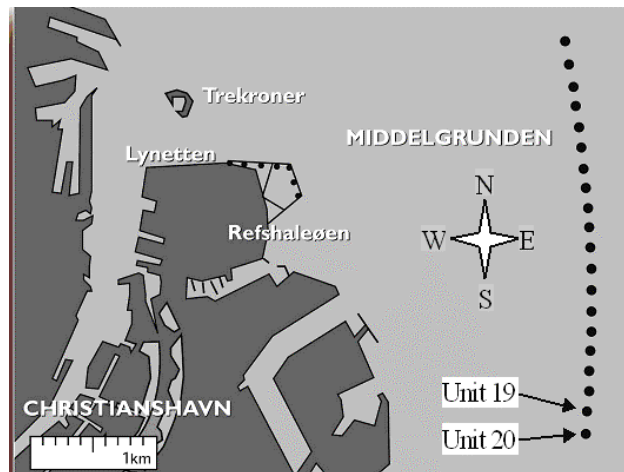


Figure 7.3 The wind farm Middelgrunden located 2-3km east of Copenhagen; water depths up to 5m.

For westerly winds turbulence intensities are high due to the closeness of Copenhagen. For other wind directions the free fetch is long, 15+ km, and the turbulence intensity is correspondingly low.

Three units are instrumented for structural measurements, of which the two most southerly units (19 and 20) are considered here. The measurements include 10min

statistics of blade and tower bending moments, power, nacelle wind speed, generator rpm, yaw position of nacelle and other sensors. The recorded statistics are mean, standard deviation, maximum and minimum of the measured quantities. Approximately 14,000 sets of 10min statistics were available for this analysis. Of these data sets only a fraction is directly relevant to the present analysis because of the scarceness of northerly and southerly wind. There is no separate met mast to measure wind characteristics, and wind speed and direction is estimated from nacelle-mounted anemometers and yaw position sensors, respectively.

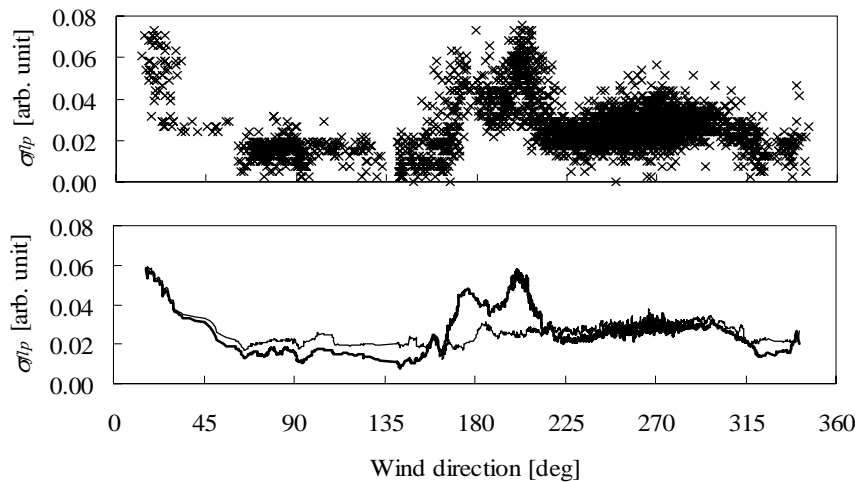


Figure 7.4 Blade flapwise bending moment standard deviation. Upper plot is estimates over 10min of standard deviation of unit 19 and lower plot is both unit 19 (thick line) and 20, smoothed over 20 10min estimates. $rpm \approx 1500$ and $5m/s < U < 23m/s$. After filtering away data with $rpm < 1500$, approx. 6000 data sets remain.

Figure 7.4 shows standard deviation of flapwise blade bending moment as function of wind direction θ for the entire wind rose. In the approximate interval $45^\circ < \theta < 135^\circ$ the dynamic response is low corresponding to the low turbulence level for easterly winds, and for $250^\circ < \theta < 315^\circ$ the dynamic response is higher due to the high turbulence generated by nearby Copenhagen. For wind directions from the North, both units experience multiple-wake conditions. For southerly wind directions, unit 20 is unobstructed and unit 19 experiences wake effects. For these closely spaced machines, it is seen that the cross-wake profile of dynamic loading deviate significantly from bell-shapes experienced with the large wind turbine spacing at Vindeby, Figure 7.1 and Figure 7.2. The discrepancy from “ideal” conditions were foreseen in Section 3, where Figure 3.6 indicates

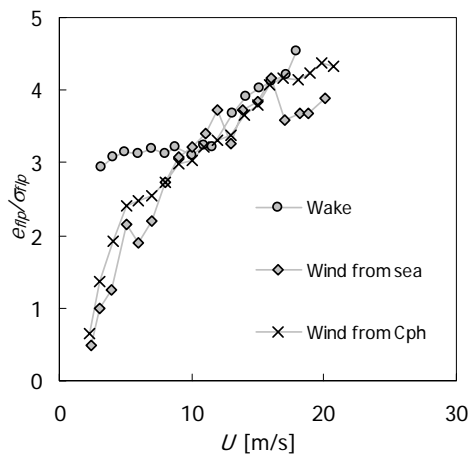


Figure 7.5 Ratio between equivalent load (e_{fp}) and standard deviation of flapwise blade bending moment (σ_{fp}) as function of wind speed¹⁶.

¹⁶ Personal communication: Peder Enevoldsen, Bonus Energy A/S

that for the wind turbine spacing at Middelgrunden the contribution from horizontal shear (half-wake) is 40-60% of the contribution from wake turbulence.

Fatigue load

The data set does not encompass equivalent load. However, a test similar to the one illustrated in Figure 4.4 was performed on the data from Middelgrunden, Figure 7.5, showing that for wind speeds larger than approx. 10 m/s, the ratio of equivalent load to standard deviation is independent of whether there are free-flow or wake flow conditions. For wind speeds less than 10 m/s the ratio taken under wake condition appears larger than for non-wake condition. As pointed out, the wind speed is measured with the nacelle anemometer, and therefore – for wake conditions – the reference wind speed is the reduced speed in the wake and not the free-flow speed. For wind speeds below 10 m/s, with the wind turbine separation in question, the wake speed deficit is estimated to be of the order 2-3 m/s. Thus, if recalculating the wind speed to free-flow speed for the wake data, the discrepancy would be less. This, together with the general observation made in Section 4 that lower wind speeds have little effect on fatigue loading, justifies using the standard deviation of response in place of equivalent load for the present investigation.

The presumption for the validity of the default model for effective turbulence, Eq. (5.23), is that the functional relationship between response and wind speed standard deviation is the same for wake and non-wake conditions. For the Middelgrunden data, under non-wake conditions, the relationship between standard deviation of wind speed fluctuations and of response is shown in Figure 7.6. Thus, by means of linear regression for non-wake conditions ($45^\circ < \theta < 160^\circ$ and $225^\circ < \theta < 325^\circ$), an empirical model for response is $\sigma_{flp} \approx 0.028 \sigma_u$ (units arbitrary), with σ_u being the value measured by the nacelle anemometer¹⁷. The data selected was here limited to medium-range wind speeds, where power limitation by means of blade pitching is not applied and linearity in the relationship should be expected, Section 4. The correlation (R) between the linear regression line and the individual data values is not high, in particular due to variation in other flow parameters not recorded or considered in the analysis. For the chosen wind speed bin, the free-flow wind speed will be of the order 1 m/s higher than the reading obtained from the nacelle anemometer.

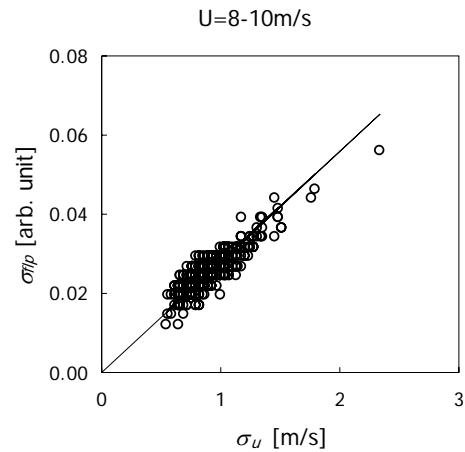


Figure 7.6 Standard deviation of flapwise blade bending moment, unit 19, for westerly winds as function of σ_u . $\sigma_{flp} \approx 0.028 \sigma_u$. After filtering, approx. 500 10min values remain, $R=0.7$.

For wake conditions, the absence of an instrument measuring the free-flow wind speed disallows making a plot similar to Figure 7.6 because the readings of the nacelle anemometer deviate significantly from free-flow wind under wake conditions. The absence of the measurement of the free-flow wind speed remains a problem when attempting to make a direct comparison of the proposed model and the measurements.

¹⁷ The uncalibrated strain gauge signal is in Volt.

Figure 7.7 shows measured flapwise blade bending moment as function of wind direction. The measured response is identical in the three plots, covering the data in the indicated wind speed interval. Also shown in Figure 7.7 is an empirical model. For modelling response under non-wake conditions, response is simply assumed to be related to the measured value of σ_u by $\sigma_{flp} \approx 0.028 \sigma_u$. For wake conditions, the standard deviation of wind speed is modelled by Eqs. (3.16) to (3.18) and in turn

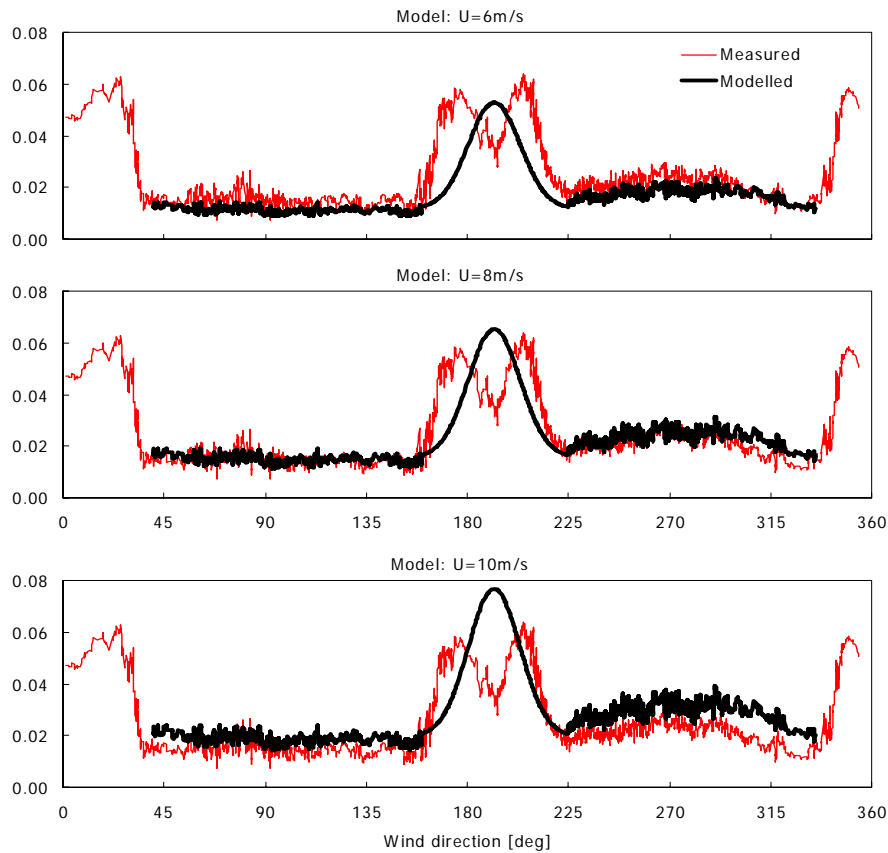


Figure 7.7 Standard deviation of flapwise blade bending moment (unit 19) as function of wind direction. The curves are running averages of the ordered measured values (thin line) and model result, respectively. For the measured response nacelle wind speed (unit 19) was $6\text{m/s} < U < 10\text{m/s}$. The total number of 10min statistics is approx. 4500.

response is determined from the regression result. It is seen that an assumed mean wind speed of 8 m/s gives the best fit to the response data for non-wake conditions, reflecting that the nacelle mean wind speed of the selected data is approx. 8 m/s.

For wake conditions, it was anticipated that the reading of the nacelle anemometer is a poor measure of the free-flow speed. Comparison of power and nacelle anemometer readings indicates that the free-flow speed, as mentioned, is of the order 2-3 m/s higher than nacelle wind speed under wake conditions.

m	6 m/s		8 m/s		10 m/s	
	All WDs	Wake	All WDs	Wake	All WDs	Wake
5	-21%	-21%	-2%	-3%	+16%	+15%
10	-16%	-16%	+4%	+4%	+23%	+23%

Table 1 Deviation between equivalent load determined from flapwise blade bending response measurements and model, respectively. For “All WDs” and “Wake” equivalent load is calculated for $45^\circ < \theta < 325^\circ$ and $160^\circ < \theta < 225^\circ$, respectively. $-m$ is the assumed Wöhler exponent. “+” indicates that the model overestimates.

Assuming perfect proportionality between standard deviation of response and fatigue loading, the equivalent load was calculated from the measured standard deviation response data and for the model, for all data minus the Northern wake sector and for the Southern wake-sector alone, applying two different Wöhler exponents. The calculations are summarized in Table 1.

Several observations may be made:

- The deviation between measured and modelled response depends little on whether or not non-wake sectors are included. This indicates – as expected – that wake conditions completely dominate fatigue loading,
- A high Wöhler exponent causes the model to yield higher response because the width of the wake response becomes less important, and not least
- Applying the model with $U = 10$ m/s (lowest plot in Figure 7.7) leads to an over-prediction of approx. 15-25%. Since the free-flow wind speed is higher than what is measured by the nacelle anemometer, the estimated over-prediction for 10 m/s is most probably the most representative.

Tower bending moment is measured by two strain gage bridges oriented approximately North-South and West-East and statistics are calculated separately for the two directions. Therefore, the along-wind tower bending (corresponding to tower deflections in the along-wind direction) can only be evaluated in a relative narrow wind direction sector around each main direction.

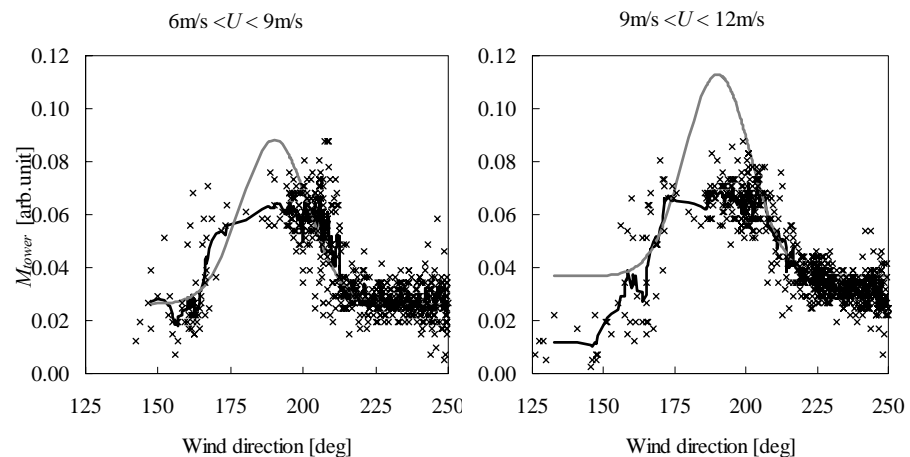


Figure 7.8 Standard deviation of tower base bending moment for two wind speed intervals. The points are measured statistics, the black lines are running averages over the data sorted by wind direction minus the “offset” and the smooth curves are prediction by model. In the free-flow sector $225^\circ < \theta < 235^\circ$ the data give $I_0 \approx 0.09$, and $M_{tower} \approx 0.04 \sigma_u$ for both wind speed bins.

Figure 7.8 shows standard deviation of tower bending moment for unit 19 for southerly wind directions when the unit is in the wake of unit 20. The data were sorted in a lower and a higher wind speed bin, and data with wind speed lower than 6m/s and higher the 12m/s were discarded. The (mean) ambient turbulence is estimated by means of the nacelle anemometer in a 10° wind direction sector next to the wake affected sector. The proportionality factor between turbulence (σ_u) and standard deviation of tower bending was estimated from the same wind direction sector.

With this information and with application of Eqs. (3.13) and (3.16) to (3.18), the effective turbulence model’s prediction of the horizontal wake-load profile is esti-

mated. In both wind speed bins, the model significantly over-predict the fatigue loading. If ambient turbulence and the proportionality factor were determined from a wind direction sector to the other (easterly) side of the wake-affected sector, the over-estimation would remain significant but less because there is a trend in the ambient turbulence intensity across the wake. As measured with the unit 20's nacelle anemometer, the change in turbulence intensity is from approx. 6% in the sector $160^\circ < \theta < 170^\circ$ to 9% in the sector $225^\circ < \theta < 235^\circ$.

With respect to tower fatigue loading – in terms of equivalent load – the model over-prediction is of the order 30-50%, i.e. the over-prediction is approx. twice as large as for flapwise blade bending.

Extreme load

Estimation of extreme loading under wake conditions was dealt with in some detail in Section 6 and data from the Vindeby Wind Farm showed that with the expectation value of 10min extreme response, normalized with standard deviation of response (peak load factor), was approximately the same for wake and non-wake conditions, Figure 6.1. For the Middelgrunden data, with the narrow spacing of the wind turbines and presumed large load contribution from (deterministic) semi-wake conditions, it is not obvious that the peak load factor would be unaffected by wake condition.

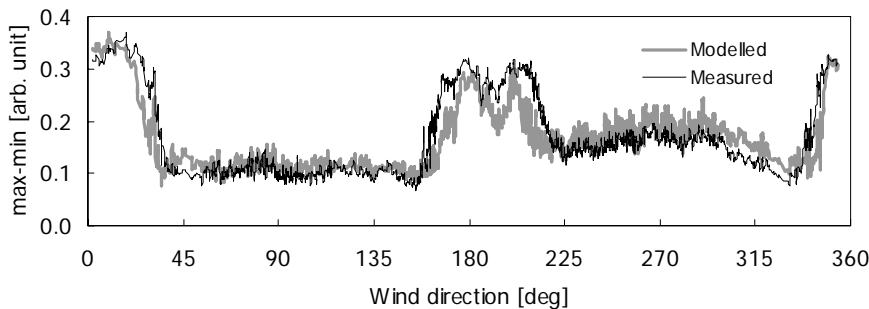


Figure 7.9 *Difference between maximum and minimum flapwise bending moment for unit 19, for wind speeds $6 < U < 10$ m/s. Data are ordered according to wind direction and smoothed over 10 estimates of 10 min. statistics. The model is $\max - \min = 0.028 \cdot \sigma_u \cdot k_p$, where σ_u is measured with the nacelle anemometer and $k_p = 8$.*

Figure 7.9 shows the difference between maximum and minimum 10min response for flapwise blade bending moment on unit 19. The measured response is compared with the response estimated by means of a peak load factor and by a proportionality factor between standard deviation of wind speed and response, both derived under non-wake conditions. The result indicates that the proposed method of extrapolation, Section 6, is a viable option also for narrow wind turbine spacing.

7.3 Other clusters

Alsvik, Danwin 180kW, $D_0 = 23$ m, stall-controlled, $s = 5, 7$ and 9.5

The small cluster of Danwind wind turbines in Alsvik in Sweden was built with the specific purpose of providing experimental evidence on wake effects, load- and production-wise. Considering the growth in economy of the wind energy industry and the obvious importance of wake effects, it gives food for thought that the Alsvik Wind Farm is unique in being built primarily for experimental purposes.

The cluster of four turbines is situated so that, for predominant wind directions, wake effects with different wind turbine separations are frequent. Thus, one unit is exposed to wake conditions with distances to the wake-generating unit of $5D_0$, $7D_0$ and $9.5D_0$, respectively. Valuable analyses on structural loading including load spectra and standard deviation of loads, have been performed, Poppen and Dahlberg (1992).

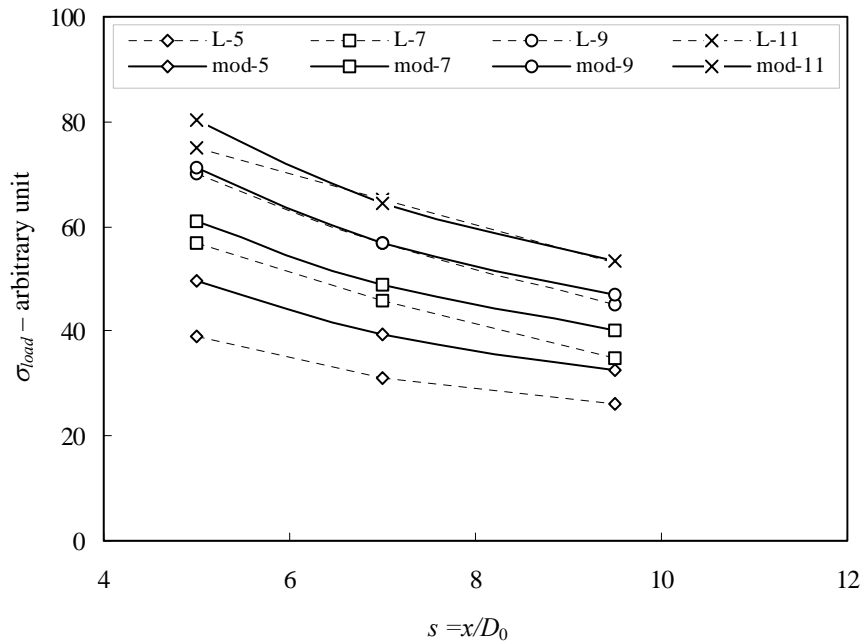


Figure 7.10 Standard deviation of maximum wake blade root bending moment as function of separation, for different wind speeds. Broken lines are measurements and full lines the effective turbulence model. Ambient turbulence intensity $I_0 = 0.06$. Alsvik wind farm: Danwind $D_0 = 23\text{m}$, $h = 30\text{m}$. Measurements are from Poppen and Dahlberg (1992).

Though not being of the same make as the turbines of the Vindeby Wind Farm, the wind turbines at Alsvik are also fixed-speed and stall controlled machines and could be expected to have similar response characteristics.

From Poppen and Dahlberg (1992), response for centre-wake conditions, in terms of standard deviation of blade root-bending moments have been deduced for mean wind speeds 5, 7, 9 and 11m/s and the three mentioned separations, Figure 7.10. Also shown are the model's response predictions, normalised to fit for non-wake conditions for the turbulence intensity $I_0 = 0.06$ corresponding to the reported very low turbulent flow approaching from the sea¹⁸. The slopes of the curves corresponding to the model fit qualitatively well for the four wind speeds. While also the level fits well for wind speeds 9 and 11 m/s, the model overestimates for the wind speed 5 m/s, the possible explanation being that the ambient turbulence is underestimated.

¹⁸ The very low turbulence intensity corresponds – for neutral atmospheric stratification – to what is observed at other offshore or shore sites at low wind speeds, with low “water roughness”. What further reduces both turbulence and load estimates is that these are based on 1min time series.

Kegnæs - Bonus 450kW, $D_0 = 36\text{m}$, stall controlled, $s = 2.5$, stall-controlled

Next, Stiesdal (1992) reported experimentally determined equivalent loads on a Bonus wind turbine with hub height 35m. The wake-exposed unit is separated $2.5D_0$ from the wake-generating neighbouring wind turbine. The equivalent load as function of wind speed is reproduced from Stiesdal (1992) in Figure 7.11. At low wind speeds, the wake equivalent load is approximately twice the free-flow equivalent load. While free-flow loads increase for the whole range of wind speeds, wake loads increases only slightly, and at approx. 17m/s free-flow and wake-flow loads are equal, indicating that free-flow and wake turbulence approach each other and that dynamic response at higher wind speeds become less dependent on turbulence in general.

Also shown in Figure 7.11 are predictions assuming that ratio between turbulence

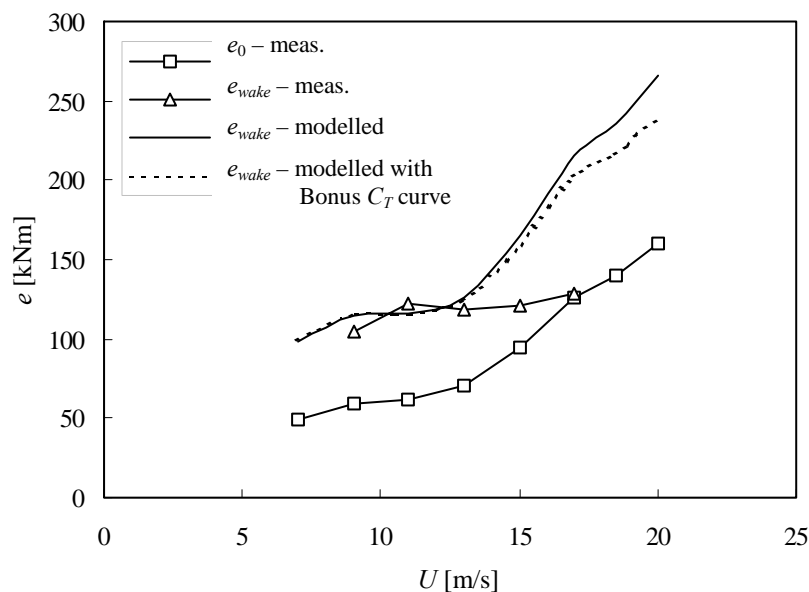


Figure 7.11 Flap-wise bending equivalent loads, Kegnæs; 2 turbines, separated $2.5D_0$. e_0 -meas and e_{wake} -meas are measured equivalent loads in free and maximum wake turbulence, respectively, and e_{wake} -modelled (solid line) and e_{wake} -modelled with Bonus C_T curve (dashed line) are modelled response with the C_T -model of this report and the C_T -values reported by Stiesdal. Measurements are from Stiesdal (1992).

and equivalent load is independent of wind speed and using for the C_T curve Eq. (3.12) and the C_T curve reported by Stiesdal (1992). The significant deviation of the prediction from measurements demonstrates that constancy of the proportionality factor is an over-simplification – and the proposed model cannot be simplified further to become unconditional of wind speed.

Kappel – Vestas/DWT Windane 34, $D_0 = 35\text{m}$, pitch-controlled, $s = 3.7$

Thomsen et al (1994) report measurements on one unit in a 24-machine array arranged in one curving row on a coastline, basically facing the south. Figure 7.12 shows equivalent loads under free-flow conditions and wake conditions, respectively, as a function of wind speed.

The pattern from Kegnæs is repeated. At lower wind speeds – here for $U < 15\text{-}20$ m/s – the model prediction (not shown) is fairly consistent with data. For the highest wind speeds with the turbine still in operation, wake effects seemingly vanish.

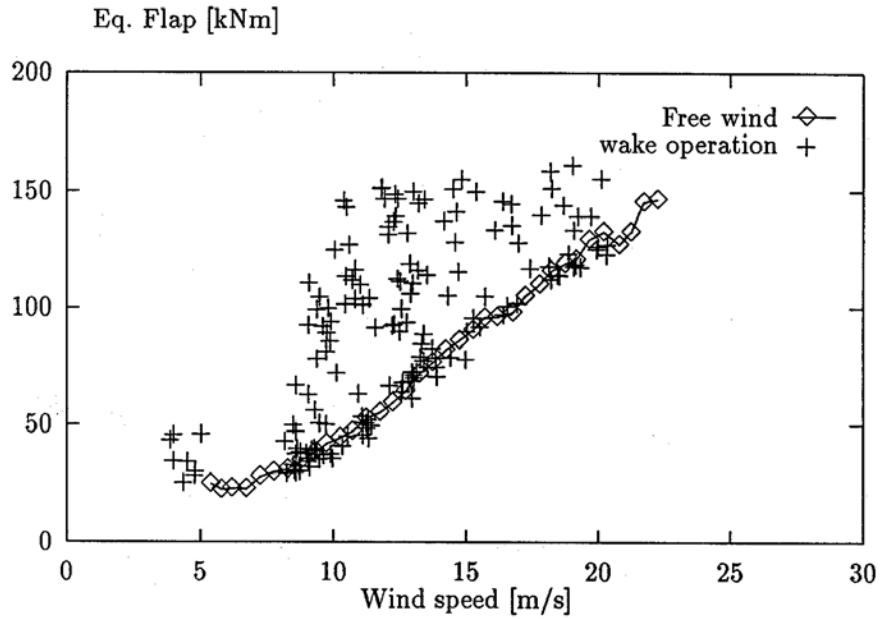


Figure 7.12 Kappel, equivalent load on flap-wise bending moment. Free-flow values are bin-averaged from 10min series and wake loads are non-averaged equivalent loads determined from the 10min series. From Thomsen et al (1994).

Risø Test Station – Vestas V27(m), pitch-controlled and Nordex 26m stall-controlled, $s = 2$

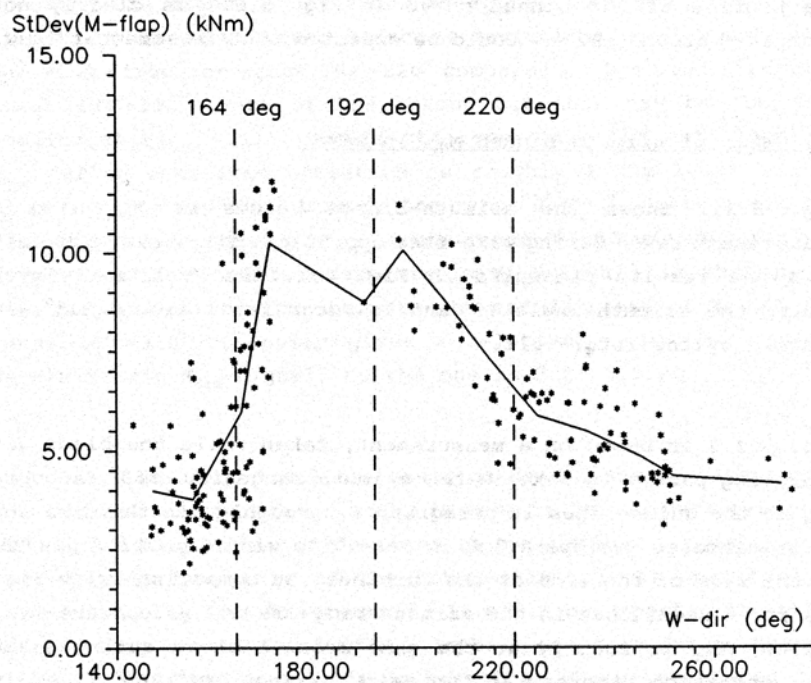


Figure 7.13 Risø Test Station for Wind Turbines; standard deviation of flap-wise bending moment on Nordex turbine, based on 80sec time series. Wind speed 6-9 m/s. From Vølund (1991).

Measurements were made on a stall-controlled Nordex wind turbine in the wake of a pitch-controlled Vestas V27 while installed at the Risø Test Station for Wind Turbines, Vølund (1991). The measurements are standard deviations of flap-wise bending moments, based on 80sec time series, Figure 7.13.

The wind direction 192° corresponds to wind directly from the Vestas to the Nordex unit. The wind speed range of the data is 6-9 m/s, signifying a high value of the thrust coefficient C_T . The increase in loading from non-wake to wake conditions is approximately from 4 to 10 kNm, which is in fair agreement with the model for effective turbulence, with $I_0 \approx 0.12-0.14$ and $I_T \approx 0.3$. It is noted that there seems to be a slight decline in loading for direct centre-line conditions (wind direction 192°) similar to what was seen at Middelgrunden, Subsection 7.2.

Summary

For half the reported sets of measurements, standard deviation was the measured quantity. The findings are summarised in Table 2, ordered with descending separations s . Only the Vestas machine at Kappel is pitch controlled.

For lower wind speeds where C_T is high, response fits well the model predictions, notably for all separations.

Site/wind turbine	Control	Measure	s	Prediction of wake loads	
				low winds	high winds
Alsvik/Danwin	stall	Std	9.5	fits**	fits**
Vindeby/Bobus	stall	E	8.5	fits	over-predicts*
Alsvik/Danwin	stall	Std	7.0	over-predicts*	over-predicts**
Alsvik/Danwin	stall	Std	5.0	over-predicts*	over-predicts**
Kappel/Vestas	pitch	E	3.7	fits	over-predicts
Kegnæs/Bonus	stall	E	2.5	fits	over-predicts
Bonus, Middelgrunden	stall***		2.4	under-predicts*	over-predicts
Risø/Nordex	stall/pi	std	2.0	fits	--

*) Slightly, **) Loads for wind speed 11 m/s, ***) active stall

Table 2. Summary of verification with respect to fatigue loading: blade flapwise bending.

For higher wind speeds, where C_T expectedly is low, the model consistently over-predicts.

The over-prediction is of lesser consequence for the large separations, where the ambient turbulence tends to dominate at higher wind speeds.

For the closely spaced wind turbines (Middelgrunden, Kegnæs and Kappel) the over-conservative model for wake turbulence is of consequence since wake-turbulence is expected to dominate throughout the wind speed range of operation. For the more reliable data at Middelgrunden, the over-prediction at moderate-high wind speeds is estimated to approx. 25% for flapwise blade bending and approx. 50% for tower bending. Whereas for the Kegnæs data, "higher wind speeds" starts at approx. 15 m/s, the Kappel data display the same features from 20 m/s and up. The Kappel wind turbines are pitch-controlled and wake loads are experienced under multiple-wake conditions. The Kegnæs wind turbine is stall-controlled and the wake loading is under single-wake condition. Nevertheless, the absence of wake

effects for high wind speeds, $U > 15-20$ m/s, unifies the two cases.

The limited experimental evidence of wake turbulence and load-effects at high wind speeds has two reasons. Firstly, wind speeds over 15-20 m/s do not occur often and even less for the wind directions with wake conditions, and secondly, the *absence* of wake effects appears less intriguing than large effects and therefore the focus has been elsewhere. At this point, it is concluded that the model appears conservative for high wind speeds, and especially for high wind speeds *and* small separations between wake generating and wake affected wind turbine.

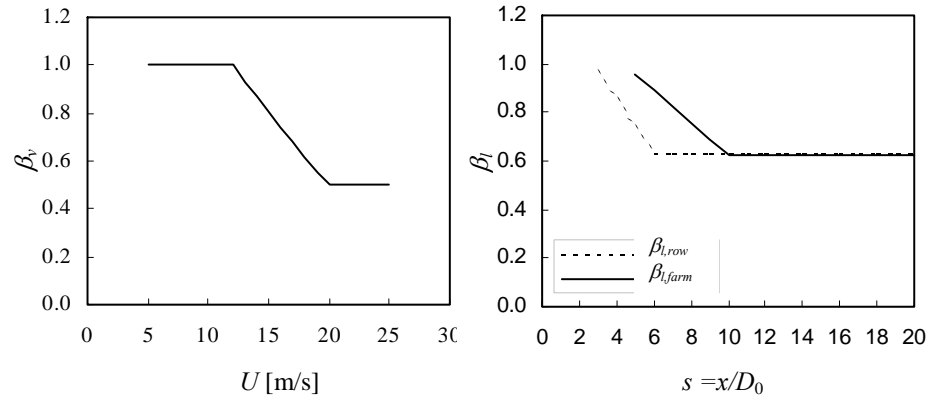


Figure 7.14 Correction factor β_v for wind speed (left) and for wind turbine separation β_l . The broken line in the right plot corresponds to a single row of wind turbines and the full line to the turbine being fully embedded in a wind farm. Deduced from *Teknisk Grundlag* (1992).

7.4 Comparison with "Teknisk Grundlag"

The recommendations attached to the Danish type approval system, *Teknisk Grundlag* (1992), have been referenced several times. *Teknisk Grundlag* (1992) prescribes the following design turbulence intensity for wind turbines in clusters:

$$I_{eff} = \sqrt{I_0^2 + I_{add}^2} , \quad (7.1)$$

where

$$I_{add} = \beta_v \cdot \beta_l \cdot 0.15 . \quad (7.2)$$

The parameters β_v and β_l are factors for wind speed and wind turbine separation, respectively. In *Teknisk Grundlag* (1992), these factors are given as graphs, see Figure 7.14. The maximum value of I_{add} is 0.15 and the minimum 0.05, the lower limit being of marginal importance when quadratically added to the ambient turbulence with values between 0.10 and 0.15.

Following *Teknisk Grundlag* (1992), wind turbine separations in single-row clusters must not be less than $s = 3$, and fully embedded in a wind farm, separations less than $s = 5$ are disallowed. It is noted that for non-constant ranges of the respective variables (U and s), $I_{add} \propto U^{-1}$ and $I_{add} \propto s^{-1}$.

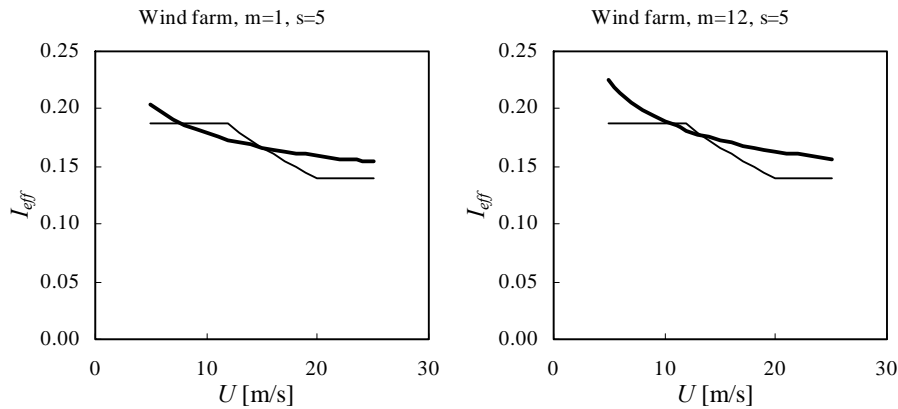


Figure 7.15 I_{eff} as function of wind speed for the Teknisk Grundlag (1992) -model (thin line) and the model of this report (bold line). Wind turbine is fully embedded, separation $s = 5$, $I_0 = 0.12$, and linear weighting (left) and Wöhler exponent $m=12$ in the right plot.

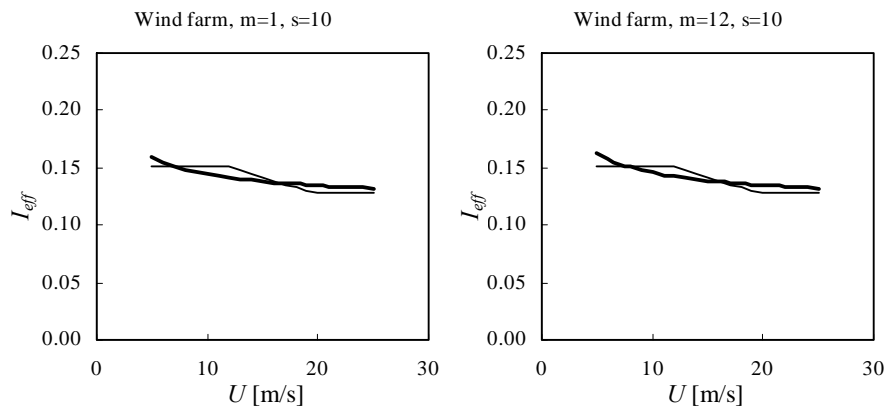


Figure 7.16 Same as Figure 7.15, but with separation $s = 10$.

Consider the case of the wind turbine fully embedded in a large array of wind turbines, Figure 7.15. The bold lines correspond to the model of this report. The difference is small for linear weighting ($m = 1$) of wake and ambient turbulence intensity. For $m = 12$, the Teknisk Grundlag (1992)-model appears non-conservative relative to the model for effective turbulence.

In Figure 7.16, the two models are compared for separations of $10D_0$. It is seen that I_{eff} for this large separation depends only marginally on the Wöhler exponent. And further, the deviation from the Teknisk Grundlag (1992)-model is small.

Turning to the Teknisk Grundlag (1992)-case of loading on a turbine in one row of turbines, i.e. with wake conditions in two directions, the model is compared with the Teknisk Grundlag (1992)-model for four combinations of the parameters s and m . The result is shown in Figure 7.17. In the two upper plots, the separation is $s = 5$ and the Wöhler exponent is $m = 4$ – corresponding to steel components – in the plot to the left and $m = 12$, corresponding to GRP¹⁹ materials, in the plot to the right. For the low value of m , the Teknisk Grundlag (1992)-model is conservative and for the high value of m the Teknisk Grundlag (1992)-model is non-conservative.

¹⁹ Glass-fibre Re-enforced Polyester.

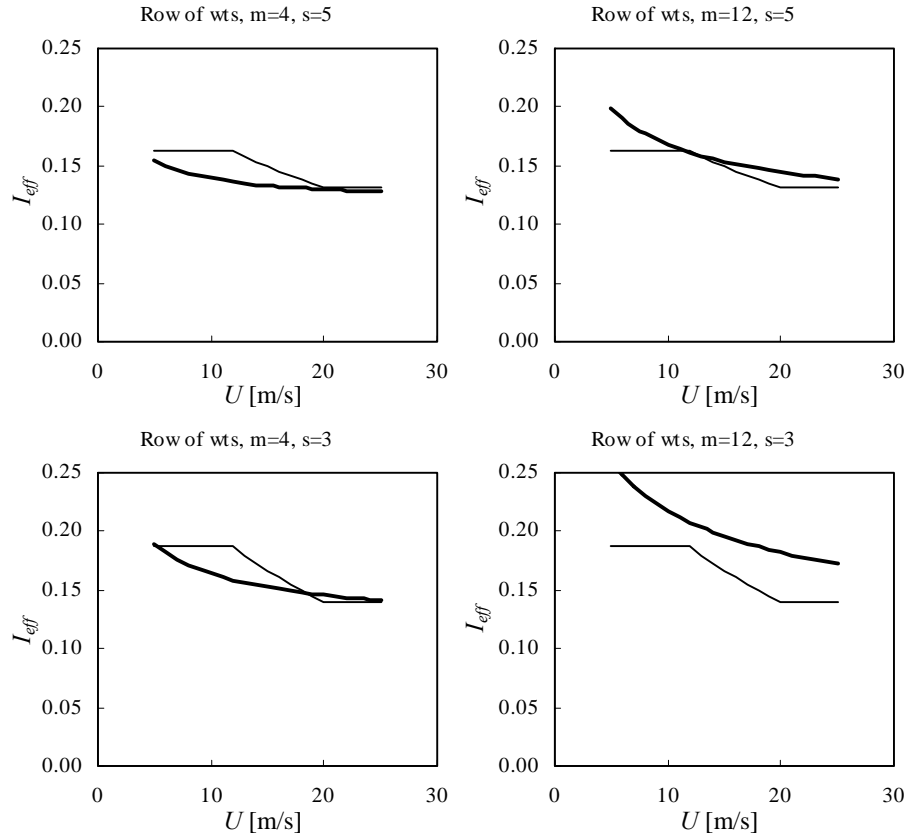


Figure 7.17 I_{eff} as function of wind speed for the Teknisk Grundlag (1992)-model and the MET. Wind turbine in one row of machines, separation $s = 3$ and 5 , $I_0 = 0.12$, and Wöhler exponent $m = 4$ and 12 .

The pattern is re-emerging for separation $s = 3$, only here the non-conservatism of the Teknisk Grundlag (1992) model is more pronounced for the high m -value. In general terms, the past's common practise was more conservative for steel components and non-conservative for GRP components.

Having identified wake effects from closely spaced wind turbines as a main deviation from past practise, it may be useful also to exceed the allowable limits of Teknisk Grundlag (1992), i.e. to investigate and compare loads at separations smaller than $3D_0$. Doing so – though *not* being a valid approach in general – it is assumed for the whole wind speed range is that $e \propto \sigma_w$, i.e. the sensitivity factor is independent of mean wind speed.

With this shortcut, a relative measure of the lifetime fatigue loading is the wind speed weighted turbulence, i.e. the effective turbulence not conditioned on wind speed:

$$I_{life} = \left(\int_5^{25} f_w(U) \cdot I_{eff}^m(U) dU \right)^{\frac{1}{m}}, \quad (7.3)$$

where f_w is the Weibull probability density distribution,

$$f_w(U) = \left(\frac{k}{A} \right) \cdot \left(\frac{U}{A} \right)^{k-1} \cdot \exp \left(- \left(\frac{U}{A} \right)^k \right). \quad (7.4)$$

For the comparison below, the values of scale and shape parameters have been chosen as $[A;k]=[9;2]$.

In Figure 7.18, the relative difference between the wind speed weighted, integrated turbulence intensity stemming from the Teknisk Grundlag (1992)-model and the model for effective turbulence are plotted as function of separation, so that when the difference is positive the model yields a larger weighted turbulence intensity than the Teknisk Grundlag (1992)-model. For the Teknisk Grundlag (1992)-model, the correction factor for turbine separation has simply been extrapolated down to $1.5D_0$ ($s = 1.5$).

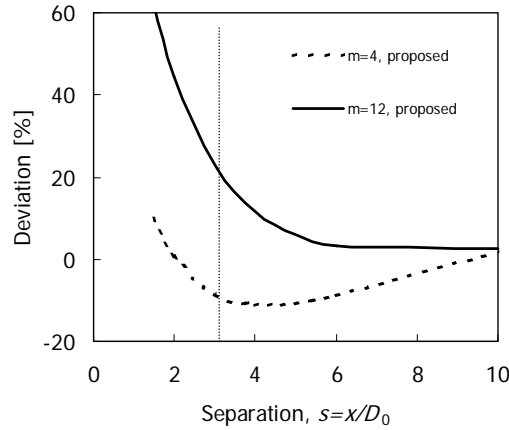


Figure 7.18 Deviation in (unconditional) equivalent load of the model from the Teknisk Grundlag (1992)-model as function of wind turbine separation for one-row cluster. The full line is for GRP components and the broken line for steel.

Figure 7.18 shows that

- For separations larger than approx. $s = 6$, the model gives the same equivalent load as the Teknisk Grundlag (1992)-model for GRP components and 0-10 % lesser equivalent loads for steel
- For separations $s = 3-6$ the loads according to the model is 0-20 % larger than what is obtained with the Teknisk Grundlag (1992)-model for GRP components and 10-15 % less for steel
- For separations smaller than $s = 3$, GRP components are loaded, as predicted by the proposed model, 20-60 % more than the (virtual) recommendations of Teknisk Grundlag (1992) and for steel the model deviates -10 % to +10 %.

Quantitatively, the most significant deviation from past practise is considerably larger fatigue design loading of the blades²⁰ for small wind turbine separations.

7.5 Uncertainties related to the model

A profound quantitative uncertainty analysis is difficult and not included in this report. The uncertainties directly quantifiable from measurements – like statistical uncertainty in wake-turbulence measurements – may well be of lesser importance than non-measurable uncertainty sources. If the component uncertainties are non-correlated, the combined, relative uncertainty can be expressed as, ISO (1993)²¹,

$$\frac{u_e}{e} = \sqrt{\sum_{i=1}^M \alpha_i^2 \cdot \frac{u_i^2}{x_i^2}}, \quad \text{where} \quad \alpha_i = \frac{x_i}{e} \cdot \frac{\partial e}{\partial x_i}, \quad (7.5)$$

where M is the number of component uncertainties, x_i are the input quantities, u_i the uncertainties in terms of standard deviations and α_i are the sensitivity coefficients.

²⁰ All major manufacturers use GRP blades.

²¹ The ISO (1993) is designed for tests where most x_i are *measured* input quantities. The approach can, however, be applied also in the present context.

From Eq. (7.5) it is noted that square summing of non-correlated component-uncertainties will strongly emphasise the largest component-contributions.

However, it has become increasingly clear to this author during the work with verification of the model for effective turbulence, with previous work related to performance of wind turbines, Frandsen et al (2000), and through discussions with colleagues working in the area of e.g. computational fluid dynamics and structural modelling that verification of design tools by means of full-scale measurements is a generic bottleneck in improving the quality of the models. In effect, the combined measurement uncertainties are often larger than the quantity (e.g. an incremental decrease in loading or increase in efficiency) that is sought for in the measurements.

Measurement uncertainty in general terms

In any case, identification of the largest uncertainties is of the utmost importance, and the presumed major uncertainty components are briefly discussed below.

Materials' model

Considerable uncertainties are attached to the experimental determination of the Wöhler-curves, to the non-homogeneity in materials, to the modelling concept and to the validity of the partial damage hypothesis. The model of this report relies to some extent on whether it is reasonable to neglect the mean load level, but basically the uncertainties related to forecasting of material fatigue are not specific to the model for effective turbulence.

Equivalent load concept

In developing the rules for weighting of turbulence intensities, it was assumed that the Wöhler curves of the materials in the whole range of load cycles follow $n_{fat} \propto s_{fat}^{-m}$, where n_{fat} is the number of load cycles causing fatigue at load amplitudes s_{fat} , and m is the Wöhler exponent. Real Wöhler curves are most often up- and downward limited and this may cause some distortion of the result. Quantification of the uncertainty is not offered here.

Explicit components in the model

In the model, the following explicit variables occur:

- Ambient turbulence I_0 ,
- Wind farm ambient turbulence I_0^* ,
- Wind turbine separations s_i ,
- Number of neighbouring wind turbines N ,
- Wöhler exponent m ,
- Wake probability p_w ,
- Rotor thrust coefficient C_T , modelled as $7/U$.

It is presumed that s_i and $-m$ relative to free-flow conditions have little or no uncertainty attached.

In Section 5, it was demonstrated that the model is robust to the choice of value of p_w .

In the final proposal, the number of wind turbines to take into account is limited to 8 and inclusion of more may increase I_{eff} only marginally.

The ambient turbulence I_0 (I_0^*) should be derived by weighting, see Section 5. Doing so, the distribution of I , conditioned on wind speed, was assumed normal. Taking into account that I is downward limited – by the application of e.g. a log-normal or Weibull distribution – may in general terms be important. However, the considerations in Subsection 5.1 point to lesser importance in the context of fatigue.

The last explicit factor in the model is the thrust coefficient C_T . With the assumption that turbulence intensity *does* serve well as primary input variable, two potential problems emerge. Firstly, is C_T alone sufficient to characterise wake turbulence intensity? And secondly, is the proposed model-formulation for effective turbulence adequate?

The first question is dealt with in some detail in Section 3 and Appendix A.3, but although turbulence immediately behind the rotor may be a function of also the power coefficient, the indication is that somewhat downwind, turbulence primarily *is* a function of C_T . As to the second question, the evidence indicates that the C_T model is conservative, i.e. the response is less than predicted by the model for high wind speeds. A model of wake turbulence intensity that would better fit the results of the verification could be

$$I_T = \sqrt{\frac{1}{(1.5 + 0.8 \cdot s / C_T)^2} + I_0^2} . \quad (7.6)$$

Eq. (7.6) provides approximately the same turbulence intensity (and thus loads) for low wind speed and less turbulence for high wind speeds. The model for effective turbulence and the alternative model for maximum wake turbulence intensity, Eq. (7.6) - when applied in the model for effective turbulence - are compared in Figure 7.19, arbitrarily referring to Teknisk Grundlag (1992). The figure shows the difference for $m = 12$, integrated over wind speed, the difference being approx. 20% for small turbine separations. Should the alternative model be applied, then the uncertainty decreases and is still biased toward conservatism.

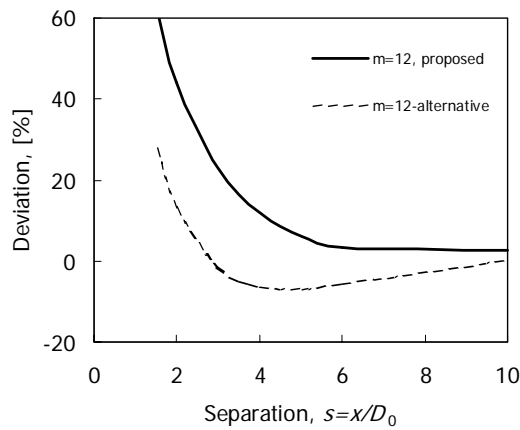


Figure 7.19 Comparison with Teknisk Grundlag of (unconditional) equivalent load. The solid line is the model and the broken line corresponds to the alternative model for I_T .

Wake-turbulence as the only descriptive wake parameter

For the validity of the model for effective turbulence to work, the impact of wind speed fluctuations is assumed dominating. This basic hypothesis was discussed extensively in Section 4, and previously in this section it was noted that fatigue loads at high wind speeds seem lower than what is found with the wake-turbulence model. The introduced uncertainty (biased toward conservatism) is estimated to be of the order 10-20%.

Different response from different machines

In Section 4, the relative importance of the various flow characteristics was investigated and discussed on basis of measurements on one particular wind turbine. The measurements on other wind turbines presented previously in this section indicate that a range of wind turbine types respond similarly. However, there may be wind turbine designs that respond differently to the same set of flow variables. Presently, no direct evidence is offered to that end.

Random variability in annual mean wind speed

Fatigue loading will (because $\sigma_u \propto U$) increase with annual mean wind speed, which varies 5-10% from year to year. The uncertainty of the experimental determination of the long-term average of the annual mean wind speed is typically of the

order 5-10% and the uncertainty of response in terms of equivalent load will be of the same magnitude.

Combined uncertainty for fatigue loading

The aggregated standard uncertainty²² of the model – still disregarding uncertainties related to the models for material fatigue – is estimated to be of the order 20-40%. With the chosen formulation and the chosen parameter values, the model is expected to be conservative. Choosing the alternative model for wake turbulence, Eq. (7.6), the uncertainty *and* conservatism will be less.

With the choices of parameters in the model, the largest uncertainty is found for narrow spacing of the wind turbines and/or for high wind speeds.

Uncertainty of model relative to ultimate load

Thus, the uncertainty in fatigue loading under wake conditions is large, but not serious compared to e.g. the stipulated uncertainty of ultimate loading during extreme wind conditions. The characteristic extreme wind speed is defined as the 98% percentile of the annual extreme 10min average wind speed, which in turn corresponds approximately to the wind speed with 50 year return period. The partial safety coefficient for ultimate load is set at 1.5 in DS 410 (1998), corresponding to a coefficient of variation of 40%, covering both variability in occurrence of the estimate of extreme wind speed and uncertainty in the load model(s).

²² ISO (1993): the uncertainty expressed by its relative standard deviation – the coefficient of variation.

8 Proposal for standard

Following discussions in the IEC working group responsible for the revision, the formulation of the model for effective turbulence appears slightly different in the edition 3 of the international wind turbine design standard, IEC61400-1 (2005). The differences are can be summarized as follows.

- Calculating the proposed effective turbulence the exponent of the Wöhler curve is applied. For that reason, the effective turbulence cannot be regarded as a physical quantity describing the *external conditions* to which the wind turbine is exposed. In the revised standard, IEC61400-1 (2005), this was dealt with by specifying as usual design values of turbulence in the section for external conditions, and by requiring that design values of standard deviation of wind speed fluctuations shall be less severe that the effective turbulence.
- The model is expressed in terms of standard deviation of wind speed fluctuations rather than turbulence intensity.
- The maximum *added* wake turbulence has been reduced with a factor $\sqrt{0.9}$.
- The maximum total wake turbulence has effectively been increased by a term to be added $1.25 \cdot \hat{\sigma}_\sigma$, where $\hat{\sigma}_\sigma$ is the estimated standard deviation of standard deviations of wind speed fluctuations.

The remaining text of this section is excerpts from IEC61400-1 (2005) of sections regarding wake effects:

(from: 7.3 Load calculations)

Loads as described in 7.3.1 through 7.3.4 shall be taken into account for each design load case. Where relevant, the following shall also be taken into account:

- *wind field perturbations due to the wind turbine itself (wake induced velocities, tower shadow etc.)*

.....

(section: 11.4 Assessment of wake effects from neighbouring wind turbines)

Wake effects from neighbouring wind turbines during power production shall be considered. The assessment of the suitability of the wind turbine at a site in a wind farm shall take into account the deterministic and turbulent flow characteristics associated with single or multiple wakes from upwind machines, including the effects of the spacing between the machines, for all ambient wind speeds and wind directions relevant to power production.

The increase in loading generally assumed to result from wake effects may be accounted for by the use of an effective turbulence intensity, which shall include adequate representation of the effect on loading of ambient turbulence and discrete and turbulent wake effects.

For fatigue calculations, the effective turbulence intensity I_{eff} may be derived according to Annex D.

For ultimate loads, I_{eff} may be assumed to be the maximum of wake turbulence intensities from neighbouring wind turbines as defined in Annex D.

$$I_{eff} = \frac{1}{V_{hub}} \max\{\hat{\sigma}_T\} \quad (33)$$

It should be noted that for wind turbine spacing less than 3 diameters the validity of such models is uncertain and caution shall be exercised.

(from: 11.9 Assessment of structural integrity by reference to wind data)

.....

An adequate assessment of wake effects can be performed by verifying that turbulence standard deviation σ_1 from the normal turbulence model is greater or equal to the estimated 90 % quantile of the turbulence standard deviation (including both ambient and wake turbulence) for $0,6 V_r < V_{hub} < V_{out}$ i.e.:

$$\sigma_1 \geq I_{eff} \cdot V_{hub} + 1.25 \hat{\sigma}_\sigma \quad (35)$$

where I_{eff} for fatigue load and extreme load calculations follows from 11.4.

.....

(from: 11.10 Assessment of structural integrity by load calculations with reference to site specific conditions)

.....

In case of wake effects it shall be verified that structural integrity is not compromised for DLC 1.1 and 1.2 in which σ_1 in the normal turbulence model is replaced by the actual wake turbulence. This may be estimated by

$$\sigma_{wake} \geq I_{eff} \cdot V_{hub} + 1.25 \hat{\sigma}_\sigma \quad (36)$$

where I_{eff} for fatigue load and extreme load calculations follows from 11.4.

Since for fatigue load calculations, I_{eff} as defined in Annex D depends on the Wöhler curve exponent m of the material of the considered component, the loads on structural components with other material properties shall either be recalculated or assessed with the appropriate value of m .

For extreme load calculations it is permitted to take into account the frequency of the wake situations and modify the load extrapolation in DLC 1.1 accordingly.

Annex D (Informative) Wake and Wind Farm Turbulence

Wake effects from neighbouring wind turbines may be taken into account during normal operation for fatigue calculation by an effective turbulence intensity I_{eff} , Frandsen (2003). The effective turbulence intensity – conditioned on hub height mean wind speed - may be defined as

$$I_{eff}(V_{hub}) = \left\{ \int_0^{2\pi} p(\theta|V_{hub}) I^m(\theta|V_{hub}) d\theta \right\}^{\frac{1}{m}} \quad (D-1)$$

where p is the probability density function of wind direction, I is the turbulence intensity combined of ambient and wake flow from wind direction θ , and m is the Wöhler (SN-curve) exponent for the considered material.

In the following a uniform distribution $p(\theta|V_{hub})$ is assumed. It is also acceptable to adjust the formulas for other than uniform distribution²³. No reduction in mean wind speed inside the wind farm shall be assumed.

²³ In the case of non-uniform wind direction distribution p_w may be adjusted by a factor equal to the ratio of the actual probability of the wind direction in the direction of the neighbouring turbines and the probability associated with uniform wind direction distribution.

if $\min\{d_i\} \geq 10 D$:

$$I_{eff} = \frac{\hat{\sigma}}{V_{hub}} \quad (D-2)$$

if $\min\{d_i\} < 10 D$:

$$I_{eff} = \frac{\hat{\sigma}_{eff}}{V_{hub}} = \frac{1}{V_{hub}} \left[(1 - N p_w) \hat{\sigma}^m + p_w \sum_{i=1}^N \hat{\sigma}_T^m(d_i) \right]^{\frac{1}{m}}; p_w = 0,06 \quad (D-3)$$

where

$\hat{\sigma}$ is the ambient estimated turbulence standard deviation

$$\hat{\sigma}_T = \sqrt{\frac{0,9V_{hub}^2}{(1,5 + 0,3d_i\sqrt{V_{hub}/c})^2} + \hat{\sigma}^2}$$

is the maximum center-wake, hub height turbulence standard deviation,

d_i is the distance, normalised by rotor diameter, to neighbouring wind turbine no. i ,

c is a constant equal to 1 m/s,

I_{eff} is the effective turbulence intensity

N is the number of neighbouring wind turbines, and

m is the Wöhler curve exponent corresponding to the material of the considered structural component.

Wake effects from wind turbines “hidden” behind other machines need not be considered, e.g. in a row only wakes from the two units closest to the machine in question are to be taken into account. Depending on the wind farm configuration the number of nearest wind turbines to be included in the calculation of I_{eff} is given in the table below.

The wind farm configurations are illustrated in the below figure D for the case “Inside a wind farm with more than 2 rows”.

Wind farm configuration	N
2 wind turbines	1
1 row	2
2 rows	5
Inside a wind farm with more than 2 rows	8

Inside large wind farms, the wind turbines tend to generate their own ambient turbulence. Thus, when 1) the number of wind turbines from the considered unit to the “edge” of the wind farm is more than 5, or 2) the spacing in the rows perpendicular to the predominant wind direction is less than $3D$, then the following ambient turbulence shall be assumed:

$$\hat{\sigma}' = \frac{1}{2} \left(\sqrt{\hat{\sigma}_w^2 + \hat{\sigma}^2} + \hat{\sigma} \right) \quad (D-4)$$

where

$$\hat{\sigma}_w = \frac{0.36V_{hub}}{1 + 0.2\sqrt{d_r d_f}/C_T} \quad (D-5)$$

C_T is the thrust coefficient, d_r and d_f are separations in rotor diameters in rows and separation between rows, respectively.

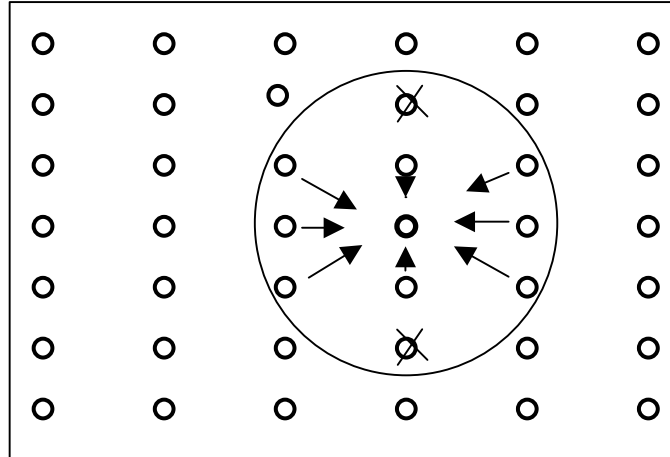


Figure D - Configuration - Inside a wind farm with more than 2 rows.

Reference documents:

S. Frandsen (2003) Turbulence and turbulence generated loading in wind turbine clusters, Draft Risø report R-1188.

9 Efficiency of large wind farms

In this section, the *array efficiency* of large wind farms is considered. The array efficiency is the power production of the cluster of wind turbines relative to the production of the same number of machines not obstructed by other machines. Although being a diversion from the main topic of the report, the array efficiency is obviously paramount to the economy of a wind farm. The considerations of Subsection 2.4, aimed at estimation of the turbulence level, were assumed valid for a position deep enough inside a wind farm for the flow to be horizontally homogeneous. Also the mean hub height wind speed, which is essential to the calculation of power production, were estimated, Eq. (2.13).

However, wind farms are not infinitely large and therefore an adequate model for the transient zone at the edge of the wind farm is needed. Present engineering tools for calculation of array efficiency – based on equations for momentum and continuity around the *individual* wind turbines and subsequent schemes for summation of the resulting wake effects – do not support accurate estimation of energy production for very large wind farms.

9.1 Roughness-change models

Aiming at linking models for array efficiency of a few wind turbines to the infinitely large clusters, a model for the transient zone between the upwind end and the far downwind end of the wind farm must be devised. Doing so, techniques related to those applied in models for the effect of change of terrain surface roughness are useful.

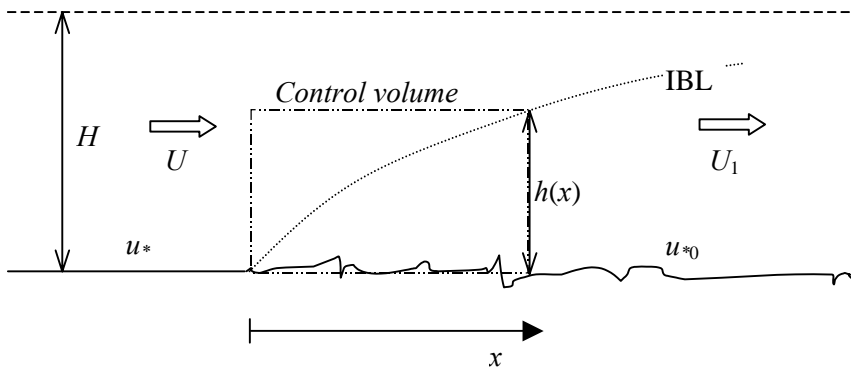


Figure 9.1 Schematics of roughness change model incl. development of an internal boundary layer (IBL).

An early model for roughness change was proposed by Elliott (1958), who equates the loss of momentum by advection to the difference between the surface shear stress and the shear stress at the top of the control volume as illustrated in Figure 9.1. In the model, it is assumed that an internal boundary layer is developed in which the flow characteristics only depend on the new surface roughness. Outside the IBL the flow is identical to the upwind flow. Elliott further assumes that the vertical mean wind speed profile is logarithmic above and inside the internal boundary layer, that there is continuity of the profile across the surface separating the internal boundary layer and the outside layer and that the friction velocity in the internal boundary layer is height-independent, but a function of downwind distance

from roughness change. The result is an implicit analytical solution for the internal boundary layer height h , which Elliott finds “extremely clumsy” and he suggests the following alternative expression for h :

$$h \approx 0.8 \cdot x^{\frac{4}{5}} z_{00\infty}^{\frac{1}{5}} \Leftrightarrow x \approx 1.2 \cdot h^{\frac{5}{4}} z_{00\infty}^{\frac{1}{4}} \quad (9.1)$$

where h is the height of the internal boundary layer the distance x downwind of the roughness change and $z_{00\infty}$ is the terrain roughness downwind of the roughness change.

Panofsky (1973) applies the same assumptions and argues that the growth of the internal boundary layer height is proportional to the standard deviation of vertical wind speed fluctuations, σ_w , and inversely proportional to the mean wind speed:

$$\frac{\partial h}{\partial x} = A \cdot \frac{\sigma_w(h)}{U(h)}, \quad (9.2)$$

where A is a constant. With the assumption of logarithmic vertical mean wind profile in the internal boundary layer and assuming that $\sigma_w \propto u_{*0}$, where u_{*0} is the friction velocity in the internal boundary layer, integration of Eq. (9.2) leads to the following implicit equation for the height of the internal boundary layer as function of x :

$$x = \frac{1}{c} h \left(\ln \left(\frac{h}{z_{00\infty}} \right) - 1 \right), \quad c \approx 0.9. \quad (9.3)$$

The constant c is empirical and was evaluated by means of data, Sempreviva et al (1990). The same study showed that the growth with x of the internal boundary layer is slower than what is obtained directly from Eq. (9.3), and therefore the European Windatlas, Troen and Petersen (1989), applies for the height of the internal boundary layer $0.3h$.

What *de facto* allows for the result of Eq. (9.3) – contrary to the result of Elliott (1958) – is the assumption that σ_w is proportional to the friction velocity of the internal boundary layer. This is not an obvious assumption. However, assuming friction velocity independent of x , a result similar to Eq. (9.3) is obtained, which is demonstrated in the following.

Consider the control volume indicated in Figure 9.1, which starts at a terrain roughness change and extends the distance x downwind. The height of the control volume corresponds to the height of the internal boundary layer $h(x)$ the distance x downwind.

To proceed, a set of assumptions is made:

- After the roughness change, close to the roughness elements, the surface friction (ρu_*^2) immediately assumes a new and constant value (ρu_{*0}^2) not dependent on x . Thus, it is assumed that the flow speed in between the roughness elements quickly reaches a balance and so does the drag on the roughness elements.
- The vertical mean wind speed profile above and under $h(x)$ is logarithmic.
- There is continuity at h : $U_1(h) = U(h)$.
- Turbulence in terms of standard deviation of wind speed fluctuations is isotropic and constant under and above $h(x)$, $\sigma_{u0} = 2.5 \cdot u_{*0} \approx u_{*0} / \kappa$ and $\sigma_u \approx u_* / \kappa$, respectively.

- The turbulent, vertical (downward) momentum flux is ρu_*^2 above h and ρu_{*0}^2 below h .
- A possible horizontal pressure gradient and the changing of the wind direction with height can be neglected.
- The height H of the boundary layer is constant, see below.

The simplified geostrophic drag law, Appendix A.2, is applied to determine the friction velocities, corresponding to the roughness of the upwind terrain roughness and the terrain roughness *far* downwind in the internal boundary layer, respectively:

$$u_* = \frac{\kappa G}{\ln\left(\frac{G}{f' z_0}\right)} \quad \text{and} \quad u_{*0} = \frac{\kappa G}{\ln\left(\frac{G}{f' z_{00\infty}}\right)}, \quad f' = 1.2 \cdot 10^{-4} \cdot e^4 = 6.5 \cdot 10^{-3}, \quad (9.4)$$

where z_0 and $z_{00\infty}$ are the surface roughness of the upwind terrain and the roughness corresponding to equilibrium far downwind, respectively. G is the geostrophic (gradient) wind speed and f' is the Coriolis parameter multiplied by e^4 .

With Eq. (9.4) and by assuming logarithmic mean wind speed profile throughout the boundary layer, the height where wind speed reaches the geostrophic value, $U(H) = G$, is

$$H = \frac{G}{f'}. \quad (9.5)$$

Thus, a consequence of the *simplified* geostrophic drag law, is that the height of the boundary layer is independent of surface roughness. With constant surface shear stress the continuity of the wind speed profile at $h(x)$ is obtained by defining the following surface roughness under $h(x)$:

$$z_{00} = z_0^\alpha \cdot h(x)^{1-\alpha}, \quad \alpha = \frac{u_*}{u_{*0}}, \quad (9.6)$$

Thus, for $z > h$ the vertical wind profile is given by:

$$U(z) = \frac{u_*}{\kappa} \ln\left(\frac{z}{z_0}\right), \quad (9.7)$$

and in the internal boundary layer, for $z < h$:

$$U_1(z) = \frac{u_{*0}}{\kappa} \ln\left(\frac{z}{z_{00}}\right). \quad (9.8)$$

Including also horizontal turbulent momentum flux, see Appendix A.3, leads to the following momentum balance for the control volume (z_L is the height, where the wind speed nominally becomes zero):

$$\rho(u_{*0}^2 - u_*^2) \cdot x = \int_{z_L}^h \rho \left[U^2 - U_1^2 + U(h)(U_1 - U) + \sigma_u^2 - \sigma_{u0}^2 \right] dz \Rightarrow \quad (9.9)$$

$$(\sigma_{u0} + \sigma_u) \frac{\kappa^2 x}{h} = U(h) - 3(\sigma_{u0} + \sigma_u) \Rightarrow$$

$$x \cong \frac{\alpha}{\kappa^2(1+\alpha)} h \cdot \left(\ln\left(\frac{h}{z_0}\right) - 3\left(\frac{1+\alpha}{\alpha}\right) \right). \quad (9.10)$$

Thus, in principle the same expression as obtained by Panofsky (1973). Excluding (as is usually done) horizontal turbulent momentum flux alters Eq. (9.10) in that 2 shall replace the factor 3.

In Figure 9.2, the result is compared with the Panofsky model (hp), the modified Panofsky model (0.3*hp) and the model of Elliott, (we). (+turb) corresponds to Eq. (9.10) and (-turb) is the result if horizontal turbulent momentum flux is neglected.

The following is noted regarding the model's height of internal boundary layer:

- The model, Eq. (9.10), fits the Panofsky and Elliott models closely at the initial 1000m and thereafter there is some diversion of the three models
- When disregarding horizontal momentum flux (-turb), the growth rate of the model is lesser than what is obtained with the Elliott and Panofsky models.
- The modified Panofsky (hp*0.3) displays a slower growth rate.
- Thus, inclusion of horizontal turbulent momentum flux (+turb) increases the growth rate of the internal boundary layer significantly.
- The consequences of choosing either constant roughness or constant friction velocity seem marginal as to the growth rate of the internal boundary layer.

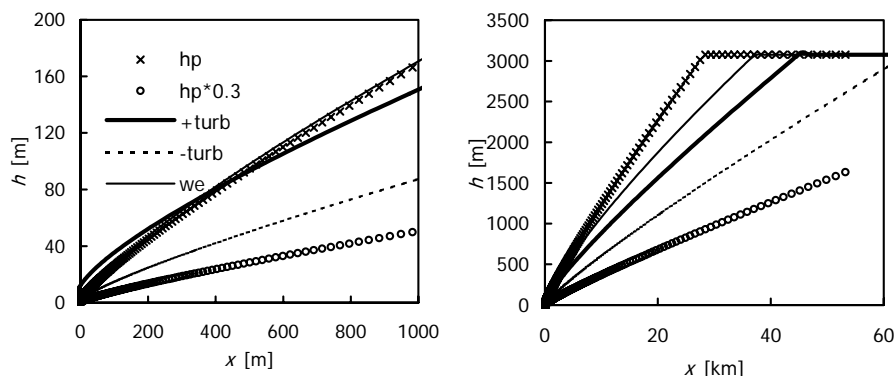


Figure 9.2 Growth of internal boundary layer after change of roughness. Upwind and downwind roughness is 0.01m and 0.3m, respectively. +turb and -turb are results from the model presented above with and without inclusion of horizontal turbulent momentum flux. hp and hp*0.3 corresponds to unadjusted and adjusted Panofsky model and we to the Elliott model.

The modified Panofsky model as applied in Troen and Petersen (1989) (the application also includes a transient layer between the internal and external boundary layer) and the above model (+turb and -turb) are also compared in Figure 9.3, which shows the development of the wind speed at 36m height after a roughness change from 0.01m to 0.3m (corresponding to a wind farm with $C_T = 0.66$ and separations $7 \times 7D_0$). Both models reach balance around 50km downwind and – for the case of an infinite wind farm – this is supported by computer simulations, Crespo et al (1999b). The modified Panofsky model yields faster reduction in wind speeds over the first few kilometres and thereafter the two models converge. For larger hub heights, the differences between the models become marginally less.

Bossanyi et al (1980) offered solutions to infinite clusters and also to finite clusters. The approaches are similar to Elliott's, typically solving the momentum equation row-by-row. In Bossanyi et al (1980), the results for the four models reviewed in the paper deviate substantially from each other. These models are compared with the Panofsky/WasP model and the model derived above in Table 3, where the wind farm (array) efficiencies of clusters with 10, 15 and infinite number of rows are compared. Determining the wind farm efficiencies, it was assumed that the thrust coefficient is constant at $C_T = 0.66$, hub height is 36 m and the spacing between the wind turbines is $7 \times 7D_0$ (the small wind turbine size was chosen to match the presentation of Bossanyi et al (1980)). "Bossanyi low" and "Bossanyi high" are lowest and highest efficiencies of the models compared in Bossanyi et al (1980), respectively. Notably, the roughness change model proposed above and "Bossanyi high" yield similar results. For the 10 row cluster the efficiency estimates vary from 64-77% to 85-88% and for the 15 row cluster, from 50-70% to 80-83%.

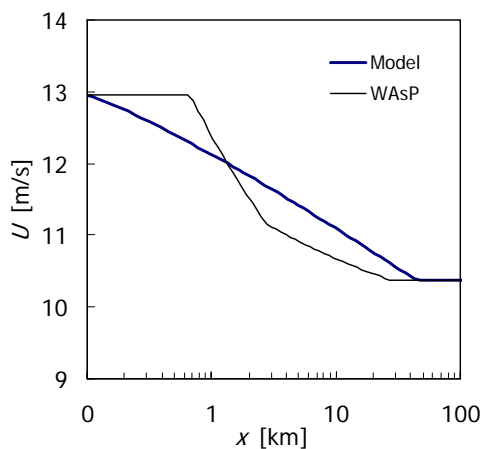


Figure 9.3 *Wind speed at height 36m as function of distance from roughness change. Up-wind and downwind roughness is 0.01m and 0.3m, respectively.*

For the infinite cluster, all models agree well. As stated previously, it is expected that turbulence – when the roughness elements are wind turbines – will be in balance much sooner than the mean wind speed. This can be understood in analogue with the initial development of flow speed between the “conventional” roughness elements. At the edge of the wind farm facing the incoming flow, the momentum transport into the individual wakes of the wind turbine units happens from all sides. A little further into the wind turbine cluster, the influx from below is strongly reduced, and even further into the cluster, the only direction from which momentum can be induced is from above the cluster. How far into the cluster is this point of balance reached? It is stipulated that this happens when wakes merge so that there is no reminiscence of free-flow between units. Depending on the specific geometry, the distance in terms of wind turbine rows would be of the order 5-10, corresponding to 3-5km with contemporary wind turbine sizes.

Model	10 wt rows	15 wt rows	∞ wt rows
Model presented herein	85%	80%	42%
Panofsky/WasP	77%	70%	42%
Bossanyi, high	88%	83%	38%
Bossanyi, low	64%	50%	38%

Table 3 *Comparison of cluster models: array efficiency.*

Summing up, when the roughness-change models presently available are applied to realistic sizes of wind farms, these yield highly diverging results.

9.2 An integrated model

The engineering models presently applied for calculating production losses due to wake effects from neighbouring wind turbines are based on local unit-by-unit momentum equations, disregarding a two-way interaction with the atmosphere. On the other hand, the models discussed in Subsection 9.1, which did not reach

engineering maturity, predict the array efficiency of very large large wind farms by viewing the wind turbines as roughness elements. A third option is to apply CFD²⁴ schemes which are promising but presently lack resolution and are computationally uneconomic.

The following is an outline of a model-complex that links the small and large-scale features of the flow in wind farms. Thus, if successful it will be applicable for any size of wind farm. The model will be evaluated and adjusted and calibrated with measurements in the near future. In essence, the model was presented by Frandsen et al (2004).

As it is often needed for offshore wind farms, the model handles *a priori* a regular array-geometry with straight rows of wind turbines and equidistant spacing between units in each row and equidistant spacing between rows. Firstly, the base case with the flow direction being parallel to rows in a rectangular geometry is considered by defining three flow regimes. Secondly, when the flow is not in line with the main rows, solutions may be found for the patterns of wind turbine units emerging corresponding to each wind direction. The solutions are in principle the same as for the base case, but with different spacing in the along wind direction and different distance to the neighbouring rows.

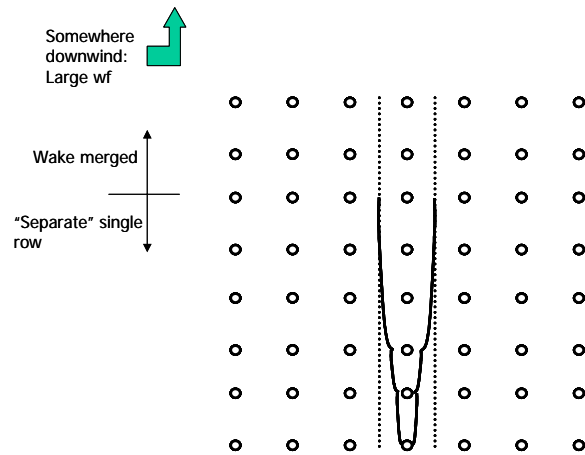


Figure 9.4 Illustration of the regimes of the proposed model. The wind comes from the “South”, parallel to the direction of the rows.

Returning to the base case and counting from the upwind end of the wind farm, the model encompasses 3 main regimes as illustrated in Figure 9.4:

- In the first regime, the wind turbines are exposed to multiple-wake flow and an analytical link between the expansion of the multiple-wake and the asymptotic flow speed deficit are derived.
- The second regime materializes when the (multiple) wakes from neighbouring rows merge and the wakes can only expand upward. This regime corresponds (but is not identical) to the flow after a simple roughness change of terrain.
- The third regime is when the wind farm is “infinitely” large and flow is in balance with the boundary layer.

Additional regimes may need to be defined when the model is to be practically applied, i.e. the first row facing the wind is obviously not exposed to wake conditions, and most frequently the wake hits the ground before it merges with the wakes from the neighbouring rows. However, it is here chosen to disregard these in order to produce a clearer presentation.

The regimes outlined above are discussed in detail in the following, and models for each regime are described.

²⁴ Computational Fluid Dynamics – numerical solutions to the equations of motion of the fluid.

Single wake

Initially, the flow through and around one wind turbine rotor is considered. Lanchester (1915) and Betz (1920) derived expressions that link thrust and power coefficients of the wind turbine to the flow speed deficit of its wake. The main device of these classical derivations was a control volume with no flow across the cylinder surface. Alternatively – and this is practical in the present context – a cylindrical control volume with constant cross-sectional area equal to the maximum wake area and with horizontal axis parallel to the mean wind vector is defined, see Appendix A3. With the approximations applied in Appendix A3, the rotor thrust becomes

$$T = \int_A \rho U (U_0 - U) dA, \quad (9.11)$$

where A is the wake cross-sectional area, which may be definite or infinite. U_0 and U are the free and wake flow wind speed, respectively. Next step in evaluation of the wake characteristics is to assume self-similarity of the wake flow speed profiles, i.e. the wake wind profile can be written as

$$U = U_w(x) \cdot f(r/R), \quad (9.12)$$

where U_w is the minimum wake flow speed at the distance x downwind of the rotor, r is the distance from the centre of the wake and R is a characteristic of the wake width at the distance x downwind of the rotor. Assuming the wake is axis-symmetric, inserting Eq. (9.12) in (9.11) and introducing polar coordinates and the substitution $y = r/R$ yield

$$T = \int_0^{2\pi} \int_0^{\infty} \rho r U_w f(r/R) (U_0 - U_w f(r/R)) dr \Rightarrow$$

$$T \propto \rho R^2 \int_0^{\infty} y U_w f(y) (U_0 - U_w f(y)) dy \quad (9.13)$$

The integral in (9.13) only depends on the minimum wake flow speed U_w . Therefore, Eq. (9.13) can be written as

$$T = C_{f1} \cdot \rho \pi R^2 U_w' (U_0 - U_w'), \quad (9.14)$$

where $U_w' = C_{f2} \cdot U_w$ is a characteristic flow speed and C_{f1} and C_{f2} are constants depending only on the integrals $\int_0^{\infty} f(y) dy$ and $\int_0^{\infty} f^2(y) dy$ i.e. the *shape* of the wake profile. The assumption of self-similarity throughout the wake is not only questionable in general terms for the regions of the wake, which is of interest in the present context. It is definitely wrong in the near-wake region. However, the self-similarity assumption maintained here is justified because the wake-affected wind turbine's rotor “integrates” the wake over a sizable fraction of its area, thus making the finer details less important.

Thus, the self-similarity allows that any actual wake shape can be represented by a rectangular distribution of the flow speed without violating the general principles of the above derivations:

$$T = \rho A U (U_0 - U), \quad A = \frac{\pi}{4} D^2, \quad (9.15)$$

where $D=D(x)$ is the diameter of the rectangular wake flow speed profile and A is the area of the wake. The thrust may also be expressed as

$$T = \frac{1}{2} \rho A_0 U_0^2 C_T, \quad A_0 = \frac{\pi}{4} D_0^2, \quad (9.16)$$

where A_0 is the swept area of the rotor, D_0 is the rotor diameter and C_T the thrust coefficient. Denominating the induction factor²⁵ $a = 1 - U_a / U_0$, where U_a is the flow speed in the wake after the initial wake expansion, then the thrust coefficient is related to the induction factor by

$$C_T = a(2 - a) \Rightarrow a = 1 - \sqrt{1 - C_T}, \quad C_T < 1, \quad (9.17)$$

and the wake cross-sectional area immediately after wake expansion, A_a , is related to the rotor area by

$$\frac{A_a}{A_0} = \frac{1 - (a/2)}{1 - a}, \quad (9.18)$$

Combining Eqs. (9.15), (9.16), (9.17) and (9.18) yields

$$A_a = \beta \cdot A_0 \text{ and } D_a = \sqrt{\beta} \cdot D_0, \text{ where } \beta = \frac{1}{2} \cdot \frac{1 + \sqrt{1 - C_T}}{\sqrt{1 - C_T}}. \quad (9.19)$$

As indicated previously, the result applies to the wake area at the position downwind where the pressure in the wake has regained the free-flow value. In real terms, it is difficult to identify exactly that position. For the model, we choose to assume that the wake expands immediately. Thus, the assumption is that $A(x=0) = A_a$. The assumption ensures a solution for all C_T values between 0 and 1 of the combined Eqs. (9.15) and (9.16).

The following expression for the wake flow speed is found:

$$\frac{U}{U_0} = \frac{1}{2} \pm \frac{1}{2} \sqrt{1 - 2 \frac{A_0}{A} C_T}. \quad (9.20)$$

For $A(x=0) = A_a$, Eq. (9.20) has solutions for $0 \leq a \leq 1$, where the “+” applies for $a \leq 0.5$ and “-“ for $a > 0.5$. Assuming monotonous expansion of the wake for increasing x , Eq. (9.20) only has solutions for $a \leq 0.5$. A frequently applied approximation to Eq. (9.15) for small wake flow speed deficits is $T \approx \rho A U_0 (U_0 - U)$, which in turn modifies Eq. (9.20):

$$\frac{U}{U_0} \approx 1 - \frac{1}{2} C_T \frac{A_0}{A} \approx 1 - a \frac{A_0}{A}. \quad (9.21)$$

Mathematically, this equation has solutions for all distances downwind and for all $0 \leq a \leq 1$. The above considerations allow only direct estimation of the initial wake flow speed deficit. In order to estimate the deficit any distance downwind, a model for the wake expansion must be found.

Schlichting (1968), Engelund (1968) and others point to a solution where $D \propto x^{1/3} \Rightarrow A \propto x^{2/3}$ for $x \rightarrow \infty$.

The result stems from several assumptions (most prominent constant eddy viscosity in the wake and self-similarity of the wake flow speed deficit and turbulence profiles, respectively) and is only valid in the far wake where also the approximation of Eq. (9.21) is valid. In any case, it is useful to adopt a model for expansion of the wake cross-sectional area as function of distance downwind that has the form:

$$D(x) = \left(\beta^{k/2} + \alpha \cdot s \right)^{1/k} D_0, \quad s = x / D_0, \quad (9.22)$$

²⁵ Usually, the induction factor is defined through the flow speed in the rotor plane.

where the initial wake diameter is $\sqrt{\beta} \cdot D_0$. If the Schlichting solution is accepted, then $k=3$. The constant α must be experimentally determined. An initial estimate can be obtained by comparing Eq. (9.21) with a fairly successfully model developed by Jensen (1983) and further by Katic et al (1986):

$$D_{(noj)}(x) = (1 + 2\alpha_{(noj)} \cdot s)D_0, \quad \frac{U}{U_0} = 1 - a \frac{A_0}{A_{(noj)}} = 1 - a \frac{D_0^2}{D_{(noj)}^2} \quad (9.23)$$

where $\alpha_{(noj)} \approx 0.1$. In this model, the initial expansion of the wake has been neglected. Presumably, the linear wake expansion is too large. However, having the “right” wake diameter at a representative distance downwind a variety of values of n will lead to plausible results. Matching the expressions (9.22) and (9.23) for wake flow speeds the distance s downwind yields

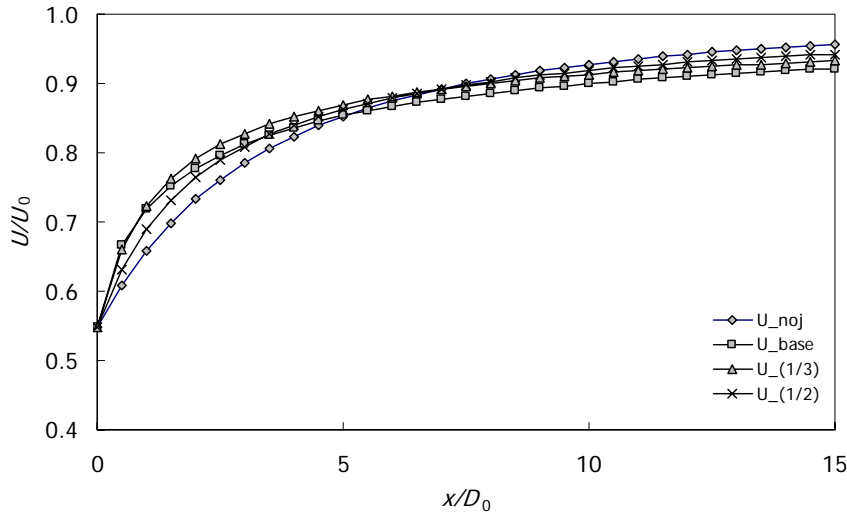


Figure 9.5 Comparison of single-wake models: “ $U_{(1/2)}$ ” is the proposed mode, “ $U_{(1/3)}$ ” is the Schlichting model, “ U_{noj} ” is Jensen (1983) model and “ U_{base} ” is Schlichting model with no linearization of momentum equation. The figure gives the relative speed in the wake as function of downwind position. $C_T=0.7$, $2\alpha_{(noj)}=0.1$; the flow speed deficits were matched at $s=x/D_0=7$.

$$\begin{aligned} (\beta^{n/2} + \alpha s)^{2/k} &= \beta \cdot (1 + 2\alpha_{(noj)}s)^2 \Rightarrow \\ \alpha &= \beta^{k/2} \cdot [(1 + 2\alpha_{(noj)}s)^k - 1] \cdot s^{-1} \end{aligned} \quad (9.24)$$

Figure 9.5 shows the relative wake wind speed as function of downwind distance from the wake generating wind turbine for different wake shapes and with and without the linearization of Eq. (9.20).

Obviously, the decay factor depends of the distance downwind chosen to match the flow speeds. For small C_T and large s , the decay factor α is of order $10\alpha_{(noj)}$.

The square root shape ($k=2$) may be a convenient choice for reasons given below.

Multiple wake, single row (regime 1)

The case of multiple-wake is dealt with as illustrated in Figure 9.6. Firstly, the possible effects from boundaries such as the ground are included implicitly through the area growth, $dA_n = A_{n+1} - A_n$. We consider a single row of wind turbines and in that row, the wake between wind turbine no. n and wake $n+1$ is described. Outside

(and in) the cylinder surface of control volume the flow speed is U_0 . The wake

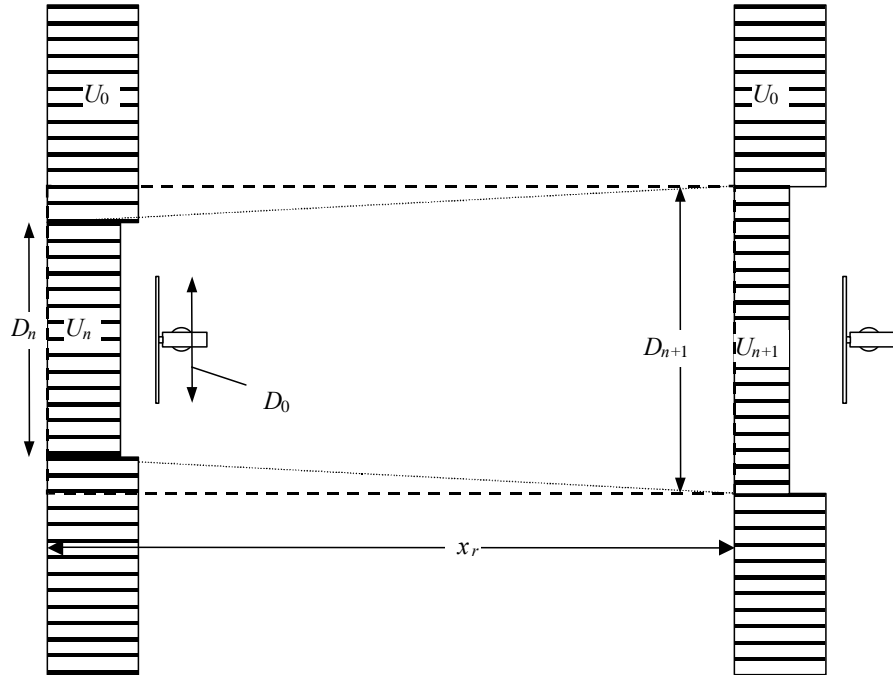


Figure 9.6 Flow between two units in a long row of wind turbines.

flow-speed profile is assumed rectangular. The flow speed at the ends of the cylinder surface is denominated as indicated in Figure 9.6. The areas corresponding to the diameters D^* are denominated A^* and is now referring to a position just in front of each unit. Note also that the cross section of the control volume need not be a circular cylinder.

Without the approximation of Eq. (9.21), we get for momentum conservation:

$$\begin{aligned} \rho A_{n+1} U_{n+1} (U_0 - U_{n+1}) &= \rho (A_{n+1} - A_n) U_0 (U_0 - U_0) \\ &\quad + \rho A_n U_n (U_0 - U_n) + T \Rightarrow \\ A_{n+1} U_{n+1} (U_0 - U_{n+1}) &= \\ A_n U_n (U_0 - U_n) + \frac{1}{2} A_R U_n^2 C_T &\Rightarrow \\ c_{n+1} (1 - c_{n+1}) &= \frac{A_n}{A_{n+1}} c_n (1 - c_n) + \frac{1}{2} \frac{A_R}{A_{n+1}} c_n^2 C_T, \quad c_n = \frac{U_n}{U_0}, c_{n+1} = \frac{U_{n+1}}{U_0} \end{aligned} \quad (9.25)$$

where

$$A_n = A_n(s) = A_n(n \cdot s_r), \quad s_r = x_r / D_0$$

is a function of the dimensionless distance s from the first wind turbine. *With* the approximation of the flow speed deficit, Eq. (9.21), the recursive equation becomes

$$c_{n+1} = 1 - \left[\frac{A_n}{A_{n+1}} (1 - c_n) + \frac{1}{2} \cdot \frac{A_R}{A_{n+1}} C_T c_n \right]. \quad (9.26)$$

For a solution, an explicit model for the wake expansion is needed.

Asymptotically for $n \rightarrow \infty$

For an infinite large number of wind turbines, it must be assumed that there is an asymptotic, non-zero flow speed: if the flow speed becomes zero then the thrust on the wind turbines becomes zero and the flow would accelerate etc. Therefore, for $n \rightarrow \infty$, $(c_n - c_{n+1}) \rightarrow 0$. Denominating the asymptotic value of the wind speed ratio $c_w = c_n|_{n \rightarrow \infty}$, an equation that links the asymptotic wake area and wake flow speed is obtained:

$$c_w(1 - c_w) = \frac{A_n}{A_{n+1}} c_w(1 - c_w) + \frac{1}{2} \frac{A_R}{A_{n+1}} c_w^2 C_T \Rightarrow A_{n+1} - A_n = \frac{1}{2} A_R \frac{c_w}{1 - c_w} C_T \quad (9.27)$$

In Eq. (9.27) the term $\frac{1}{2} A_R \frac{c_w}{1 - c_w} C_T$ is a constant, and thus – asymptotically – wake cross-sectional area is expanding linearly with x . Eq. (9.27) points to an interesting result: without assumptions regarding eddy viscosity, it is possible to derive the wake expansion for an infinite row of two-dimensional obstacles: $D \propto x^{\frac{1}{2}}$. This expansion is the only shape that asymptotically will ensure a non-vanishing and non-increasing flow speed.

By assuming the wake cross-section circular, it is now possible to link the decay factor α in Eq. (9.22) to the asymptotic flow speed ratio c_w . With the wake model of Eq. (9.22) with $n=2$ corresponding to the square root expansion of wake diameter, the increase in wake cross section is

$$A_{n+1} - A_n = \frac{\pi}{4} D_0^2 (\beta + \alpha \cdot s_r \cdot (n+1)) - \frac{\pi}{4} D_0^2 (\beta + \alpha \cdot s_r \cdot n) = A_R \cdot \alpha \cdot s_r \quad (9.28)$$

where s_r is the dimensionless distance between the wind turbines in the row. Inserting Eq. (9.28) into Eq. (9.27) yields

$$\alpha = \frac{1}{2} \cdot \frac{C_T}{s_r} \cdot \frac{c_w}{1 - c_w} \quad (9.29)$$

Thus, if the asymptotic, relative flow speed in the wake is known, then the decay constant is given. Conversely, the relative wake flow speed is given as

$$c_w = \frac{\alpha}{\alpha + \frac{1}{2} \frac{C_T}{s_r}} \quad (9.30)$$

In Figure 9.7, the result of applying Eq. (9.25) is compared with data from the wind farm Nørrekær Enge II. It is seen that the flow speed ratio (i.e. also c_w is approximately constant) is only marginally dependent on free-flow mean wind speed. With a C_T measured on a wind turbine similar to the units in question and with that curve approximated by $C_T \approx 3.5 \cdot (2U - 3.5) \cdot U^{-2}$, it is found that the decay constant must be proportional to C_T to satisfy Eq. (9.29). The full line in Figure 9.7 is the

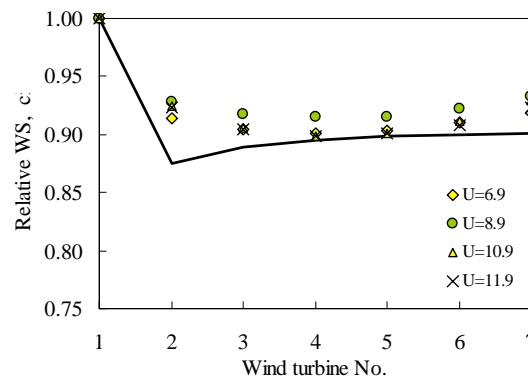


Figure 9.7 Measurement of wind speed ratio, c_i , at Nørrekær Enge II. Wind speeds are derived from power signals. Average is taken over six rows with each 7 units. $s_r \approx 7$. The wind farm consists of 42 300kW units.

average of the average value of C_T for the 4 different wind speeds.

The consequence of the flow speed ratio c_w not depending on wind speed, is that the decay constant α is a function of C_T , i.e. the initial wake deficit/turbulence.

Multiple wake, merged (regime 2)

When the wakes from different rows meet, the lateral wake expansion is stopped and the wake area can only expand upward. Since the area must expand linearly to satisfy Eq. (9.27), the height of the wake must increase linearly. This implies that the growth of wake-height, being equivalent to the internal boundary layer for roughness change models, asymptotically has $h \propto x$.

Also in regime 2, the “wake area” must expand linearly for the flow speed to approach a non-zero value. Since the wake cannot expand laterally, the incremental growth of the internal boundary layer in regime 2 in the limit for $n \rightarrow \infty$ is

$$\Delta h = \frac{dA}{s_r D_0} = \frac{A_{n+1} - A_n}{s_r D_0}, \text{ and} \quad (9.31)$$

$$\begin{aligned} \frac{\partial h}{\partial x} &\approx \frac{A_{n+1} - A_n}{\Delta x} \frac{1}{s_r D_0} = \frac{1}{2} A_R \frac{c_{mw}}{1 - c_{mw}} C_T \frac{1}{\Delta x} \cdot \frac{1}{s_r D_0} \\ &= \frac{1}{2} \frac{c_{mw}}{1 - c_{mw}} \frac{\pi}{4} D_0^2 C_T \frac{1}{s_f D_0} \cdot \frac{1}{s_r D_0} \Rightarrow \end{aligned}$$

$$\frac{\partial h}{\partial x} \approx \frac{c_{mw}}{1 - c_{mw}} c_t \Rightarrow h = \frac{c_{mw}}{1 - c_{mw}} c_t \cdot (x - x_0) + h_0, \quad (9.32)$$

$$c_t = \frac{\pi C_T}{8 s_r s_f}$$

where s_f is the dimensionless distance to the neighbouring rows, c_{mw} is the relative flow speed in the wake and x_0 and h_0 are integration constants to be determined. We want to make a comparison of this result with Elliott (1958), Eq. (9.1). For the purpose we choose the following values of the parameters:

$$s_r = s_f = 7, C_T = 0.5, D_R = h_H = 100m .$$

Where h_H is hub height. Assuming that the asymptotic wind speed is identical to the flow speed for the infinitely large wind farm, Appendix A.2, it is found that

$$z_{00} = 0.55m, \quad \frac{\partial h}{\partial x} = 0.019,$$

where z_{00} is the apparent roughness for the infinitely large wind farm. Further, assuming that in Regime 1 the wake expands similarly in lateral and vertical direction from hub height, we estimate the height of the multiple-wake when these merge with neighbouring rows to

$$h_0 \approx h_H + \frac{1}{2} \cdot s_f D_R \approx 5D_R \approx 500m \text{ (when better experimental information becomes available, the suggested wake expansion model should be used) The distance downwind from the edge of the wind farm to where the wakes merge is here estimated as:}$$

$$x_0 \approx 10h_0 = 5000m .$$

In Figure 9.8 the growth of the internal boundary layer is plotted for Elliott’s model and for the proposed model, which are given by Eqs. (9.1) and (9.32), respectively. As to the functional dependency on distance downwind, the model compares well with Elliott (1958), who suggests the approximation $h \propto x^{\frac{4}{5}}$. The Elliott model estimates the internal boundary approx. 3 times higher than the proposed model. However, correcting the Elliott model in accordance with the experimental data of Sempreviva et al.(1990), the proposed model fits Elliott’s model well.

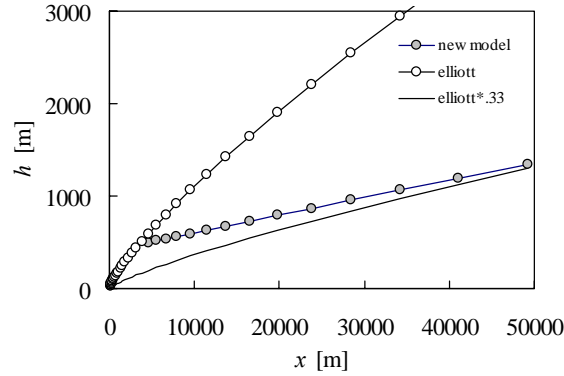


Figure 9.8 Growth of internal boundary layer as function of downwind distance to front end edge of wind farm.

Wind farm in balance with boundary layer (regime 3)

Regime 3 corresponds to the flow inside a wind farm so far away from the upwind boundary of the wind farm that the boundary layer is fully developed. A model for the infinitely large wind farm has to some extent been reported previously, Frandsen (1993) and Frandsen and Madsen (2003), Subsection 2.4 and Appendix A.2. The apparent “wind farm roughness” may be expressed as:

$$z_{00} = h_H \cdot \exp\left(-\frac{\kappa}{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}\right), \quad (9.33)$$

where h_H is wind turbine hub height, z_0 is terrain surface roughness, κ is the von Karman constant and c_t is the drag coefficient per m^2 terrain surface from the wind turbines. In particular for large wind speed deficits, this result differs significantly from Newman (1977) in that it predicts a lesser deficit. From the large-wind-farm solution the hub height wind speed and thus the ratio of the flow speed to the free flow speed, c_{wf} , may be found. The way the model is built, c_{wf} must be the asymptotic value for regime 2.

Other wind directions

In the regular wind turbine array other wind directions are to be treated similarly: for each wind direction new rows (with larger wind turbine spacing) will form, with new (smaller) distances between rows. The *general* applicability of the proposed model will depend on whether the model – once calibrated by means of measurements – works for all wind directions.

9.3 Summary

Summarizing, the integrated model has the following components:

1. From wake no. 2-3 to where the wake merge with neighbour-row wakes, use “row of wts” wake shape expanding in 2 directions:

$$\frac{D_R}{D_1} = \frac{1}{(\beta + \alpha \cdot s_f)^{1/2}}. \text{ The asymptotic relative wake speed deficit,}$$

$c = U / U_0$, has – if the row is long enough – an asymptotic value, c_1 . The

specific value is found from experiments and this value determines the decay constant α .

2. From the point of neighbour-wake merging and onward, the merged wake expands linearly upward, $h = \frac{c_{mw}}{1-c_{mw}} c_t \cdot (x-x_0) + h_0$, where x_0 and h_0 in principle is derived from the characteristics of the flow exiting regime 1.
3. Determine the relative flow speed deficit from the model from the infinitely large wind farm, c_{wf} . The first approximation is that $c_{mw}=c_{wf}$.

It is believed that the suggested model will – with appropriate experimental “calibration” – encompass the flow characteristics of large wind farms in a realistic and consistent manner.

The model will be verified/calibrated by means of existing data and data from the large offshore demonstration project at Horns Rev and Nysted.

To verify experimentally the flow speed deficit in the infinitely large wind farm, c_{wf} , will be difficult and is presently viewed as a major challenge.

Presently, the model only allows simple geometries and there is a need to extend it to irregular geometries.

Apart from determining the efficiency of the wind farm, the estimation of the growth of the internal boundary layer is needed to determine what happens downwind.

10 Concluding remarks

Fatigue loading

The model was developed in an attempt to fill out an empty space in national and international standards by accounting for increased loading caused by the wakes of neighbouring wind turbines in wind farms.

It has been demonstrated that the model encompasses sufficient details to adequately reflect the flow environment of the wind farm. The model is composed of sub-models derived from experimental evidence and theoretical considerations.

Since the model is aimed at material fatigue, an important feature is to take into account the slope of the Wöhler curve. The model includes weighting of turbulence from wind directions with and without wakes in accordance with the Wöhler exponent of the considered material. This leads – relative to practise of yesterday – to less conservatism for steel and higher safety for fibreglass materials, which presently are the preferred materials for wind turbine blades.

There are indications that the model could be conservative for the higher wind speeds. It is felt, however, that the experimental evidence should be re-enforced before a relaxation of the requirements of the model is introduced.

In deriving the expression for summation of turbulence intensities, it was assumed that the applied values for free-flow and wake turbulence already were adjusted to account for random variability of turbulence. *If* the input values of free flow- and wake turbulence intensities are mean values rather than already weighted turbulence, then the expression for the effective turbulence should be slightly changed. Thus, the applied free-flow turbulence intensity should have been, see Section 5:

$$I_{eff,free} \approx I_0 \left(1 + \frac{1}{2}(m-1) \left(\frac{\sigma_I}{I_0} \right)^2 \right), \quad (10.1)$$

where I_0 is best estimate of the mean turbulence intensity, and $(\sigma_I / I_0) = \delta_I$ its coefficient of variation, and since it appears from data that the coefficients of variation of I under wake conditions and free-flow conditions are fairly similar, the effective wind farm turbulence becomes

$$I_{eff} \cong \left(1 + \frac{1}{2}(m-1) \cdot \delta_I^2 \right) \cdot \left[(1 - N \cdot p_w) I_0^m + \sum_{i=1}^N p_w I_i^m \right]^{\frac{1}{m}}. \quad (10.2)$$

Note that this expression is – so to speak – the best estimate of the effective turbulence and that introduction of safety margins must be done separately. Should more detailed information on δ_I be available to the designer, then Eq. (10.2) could be used.

Future regarding the fatigue model

It was found that should other wake-flow variables be of importance, these could be indirectly included by “calibrating” the model for I_T . The variable wake-deficit was found to play a role. However, the role was found to be limited and wake deficit was disregarded, still leaving a model, which is assumed to be conservative.

The model can readily be calibrated should new relevant experimental evidence emerge.

Ultimate load

Ultimate load in the wind farm has been touched upon in this report. Extreme response is the sum of a mean and a stochastic component and for that reason it is not possible to directly devise an effective turbulence for extreme loads. However, indications are that the issue can be dealt with rather straightforwardly, though there may be a problem – in common with the free-flow case – in extrapolation from 10min statistics to longer periods.

Efficiency of large wind farms

The efficiency of offshore wind farms – extending over many kilometres – is becoming an important issue over the next few years, not least in connection with the Danish offshore projects. The experimental evidence on the interaction between wind farm and boundary layer is scarce.

It is believed that the presented model has the potential to fill in the gap between existing models for small wind farms and models for infinite wind farms.

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12 Nomenclature

It has been necessary to “recycle” both the Latin and the Greek alphabet, in some cases several times. Apart from below, the symbols are defined where used.

a	Slope parameter for the IEC turbulence std. model
a	Constant in model for added turbulence “over” wind farm
a	Material constant
a	Material constant for zero mean stress
a	Rotor induction factor
A	Constant in model for internal boundary layer growth after roughness change
A	Constant in geostrophic drag law
A	Weibull scale parameter
A	Wake cross-sectional area [m ²]
A^*	Constant in simplified geostrophic drag law
A	Constant in model for response
A_T	Surface area of control volume [m ²]
A_r, A_0	Swept rotor area [m ²]
A_a	Wake cross-sectional area after initial wake expansion
b	Wake induction factor/reduction factor
b	Constant in model for added turbulence “over” wind farm
b	Constant in expression for “effective” width of wake
b_j	Constant in expression for “effective” width of wake no. j
B	Constant in geostrophic drag law
c	Constant in Panofsky’s roughness-change model
c	Weighting factor in the Teknisk Grundlag (1992) model
c_1	Constant in model for added wake turbulence
c_2	Constant in model for added wake turbulence
c_t	Drag coefficient for wind farm – drag per-area-unit terrain surface
c_u	Proportionality factor
c_w	Wind speed ratio
c_{mw}	Wind speed ratio
C	Cord length [m]
C_D	Drag coefficient
C_L	Lift coefficient
C_P	$= P / (\frac{1}{2} \rho U^3 A_r)$, Power coefficient of wind turbine
$C_{P,max}$	Maximum power coefficient of wind turbine
C_T	$= T / (\frac{1}{2} \rho U^2 A_r)$, Thrust coefficient of wind turbine
C^*	Constants
$C(t)$	Time-dependent damping-matrix of wind turbine structure
D	Characteristic wake-width [m]
D	Width of cylinder
D_0	Wind turbine rotor diameter [m]
D_i	Partial damage
D_T	Total damage
D_r	Damage rate
$E\{*\}$	Expectation value
e	Equivalent load
e_0	Equivalent double-amplitude load
e_w	Weighted equivalent load
e_{wake}	Equivalent load in wake
e_{free}	Equivalent load in free-flow
f'	Coriolis parameter multiplied by e^4

f	Coriolis parameter
f	PDF of turbulence intensity
f	Frequency scale of turbulence [Hz]
f	Wake shape function
f_{fat}	Wöhler curve
f_m	PDF of response amplitude process
f_w	Weibull PDF
f_{wd}	PDF of wind direction
$f_{wd,j}$	PDF of wind direction in direction of wake no. j
F	Force on cylinder [N]
F_{adm}	Structural admittance function
\bar{F}	Generalised mean force on cylinder [N]
F_L	Lift force on blade section [N/m]
F_D	Drag force on blade section [N/m]
F_y	Distribution of extreme response
F_U	Distribution of extreme gusts
$F_{U,as}$	Asymptotic distribution of extreme gusts
F_0	Distribution of extreme gusts under non-wake conditions
F_w	Distribution of extreme gusts under wake conditions
F_v	Extreme distribution of 10min normalised gust
g	PDF of external conditions' input variables
g	Acceleration of gravity [m/s ²]
G	Geostrophic wind speed [m/s]
h_H	Hub height of wind turbine [m]
h	Height of internal boundary layer after roughness change [m]
h_p	Height of internal boundary layer, Panofsky [m]
H	Height of boundary layer [m]
H_C	Height of cylinder [m]
H	Frequency response function
i	Deviation of I from its mean
i_{addwf}	Added turbulence intensity as function of downwind distance from row of wind turbines
i_T	Total turbulence intensity as function of downwind distance from row of wind turbines
I	Turbulence intensity
I_0^*	Wind farm "ambient" turbulence intensity in wind farm
I_{add}	Added turbulence intensity in wake
I_{15}	Constant in hub-height IEC turbulence intensity
I_m	Fatigue-weighted turbulence intensity, free-flow
I_0	Ambient (free-flow) turbulence intensity
$I_{0,j}$	Ambient (free-flow) turbulence intensity in direction j
\bar{I}_0	Mean of ambient turbulence
I_{addwf}	Added "ambient" wind farm turbulence
I_{eff}	Effective turbulence intensity
$I_{e,free}$	Effective turbulence in the free flow
$I_{T,j}$	Maximum turbulence for wake no. i
I_{ijfe}	Turbulence intensity, not conditioned on wind speed
I_T	Maximum wake turbulence intensity
$I_{T,wf}$	Wind farm turbulence intensity
I_{ext}	Extreme I during period T
$I_0^{(*)}$	Modified Bessel function
k	Constant in wake expansion model
k	Weibull shape parameter
k_p	Peak factor
k_{ext}	"Extreme" peak factor
K	No. of upcrossings

$\mathbf{K}(t)$	Time-dependent stiffness-matrix of wind turbine structure
L	Length scale of (along-wind) component of wind velocity fluctuations [m]
m	Exponent of power law Wöhler curve
m_e	Equivalent mass of cylinder [kg]
$\mathbf{M}(t)$	Time-dependent mass-matrix of wind turbine structure
M	No. of uncertainty components
M	No of 10min period in lifetime
M	Momentum flux [$\text{kg m}^{-1} \text{s}^{-2}$]
$M_{vertical}$	Vertical momentum flux [$\text{kg m}^{-1} \text{s}^{-2}$]
M_{cyl}	Momentum flux through ends of control volume [$\text{kg m}^{-1} \text{s}^{-2}$]
M_e	Momentum flux through cylinder surface of control volume [$\text{kg m}^{-1} \text{s}^{-2}$]
M_{mean}	Mean bending moment [Nm]
M_{min}	Expectation mean of maximum bending moment during 30min period [Nm]
M_{max}	Expectation mean of minimum bending moment during 30min period [Nm]
M_{ext}	Expectation extreme bending moment during period T [Nm]
$M(*,*,*)$	Confluent hypergeometric function
$n1$	Exponent of C_T in model for wake turbulence
$n2$	Exponent of s in model for wake turbulence
n_i	No. of load cycles at stress range s_i
n_{fat}	No. of cycles causing fatigue failure at stress range s_{fat}
n_t	Integrated number of load cycles
N	Number of neighbouring wind turbines (No. of load cycle series)
N	Number of load sequences
p	Energy flux [W/m^2]
p	Pressure [N/m^2]
$\{p\}$	Load vector
p_j	Probability of load
p_0	Probability of non-wake condition
p_w	Probability of wake condition
$p_{w,j}$	Probability of wake no. j
P	Wind turbine power [W]
PDF	Probability density function
q	Mass flux [$\text{kg}/(\text{m}^2\text{s})$]
Q	Volume flow (flux) [kg/m^2]
Q_{cyl}	Volume flow (flux) out of cylinder surface of control volume [kg/m^2]
Q_e	Volume flow (flux) out of ends of control volume [kg/m^2]
r	Distance axis to position on blade [m]
r_1	Distance axis to position on blade [m]
r_2	Distance axis to position on blade [m]
R	Randomly, Rayleigh-distributed amplitude
R_0	Radius of wind turbine rotor [m]
R_x	Amplitude process
s	Separation in rotor diameters to wake-generating wind turbine
s_a	Mean stress
s_u	Mean stress
std	Standard deviation
s_i	Distance to wind turbine no. i
s_f	separation between rows, in rotor diameters
s_r	separation of units in the rows, in rotor diameters
s_{fat}	Stress range causing fatigue failure for n_{fat} load cycles [N/m^2]
SMS	Sea Mast South, Vindeby
SMW	Sea Mast West, Vindeby
Std.	Standard deviation
S	Instantaneous flow speed [m/s]

S	Flow speed experienced by blade section [m/s]
S_y	Power spectrum of response of cylinder
S_x	Power spectrum of response
S_u	Power spectrum of along wind wind speed fluctuations [m^2/s]
S_u	Cross spectrum of u [m^2/s]
St	Strouhal number
t	time [sec]
t	$= \rho c u_h^2$, distributed thrust (per-area-unit land area) [N/m^2]
T	$= 1/2 \rho C_T A_r u_h^2$, thrust on wind turbine [N]
\vec{T}	Force vector acting on control volume [N]
u	Wind speed fluctuation, along wind component [m/s]
u^*	Friction velocity [m/s]
u^*_i	Friction velocity under hub height [m/s]
u^*_{i0}	Friction velocity over hub height in infinite wind farm [m/s]
u^*_{i0}	Friction velocity under $h(x)$ after roughness change [m/s]
u^*_{addwf}	Friction velocity including wind farm [m/s]
u_e	Uncertainty in equivalent load
u_t	Wind speed fluctuations perpendicular to flow direction [m/s]
U	Mean wind speed [m/s]
U_0	Free-flow mean wind speed [m/s]
U_b	Mean wind speed in the wake [m/s]
U_h	Mean wind speed at hub height in wind farm [m/s]
U_i	Mean wind speed over 30min period [m/s]
U_1	Mean wind speed under $h(x)$ after roughness change [m/s]
U_{48}	Mean wind speed at height 48m [m/s]
U_{20}	Mean wind speed at height 20m [m/s]
$U_{38,SMW}$	Mean wind speed at 38m in SMW [m/s]
$U_{38,SMS}$	Mean wind speed at 38m in SMS [m/s]
U_h	Hub height mean wind speed [m/s]
U_r	Mean wind speed in rotor plane [m/s]
U_w	Wake mean wind speed [m/s]
v	Wind speed fluctuations, lateral to wind direction [m/s]
v	Normalised gust, along wind [m/s]
v_K	Largest extreme of K upcrossings
w	Wind speed fluctuations, vertical [m/s]
x	Distance downwind from wind turbine [m]
x	Distance downwind from roughness change [m]
x	Stochastic load
x_r	Separation between wind turbines in the rows [m]
x_f	Separation between rows [m]
x_i	Input variables
X	Control volume [m^3]
$\{y\}$	Vector of deflections and rotations
y	Generalised deflection
y	Combined response of stochastic and sinusoidal load
y_0	Mean response
z	Height above ground [m]
z_0	Terrain surface roughness [m]
z_{00}	Apparent, combined roughness of wind turbines and terrain [m]
$z_{00\infty}$	Constant roughness downwind of a roughness change [m]
z_L	Height, where the logarithmic profile yields zero-wind speed [m/s]
α	Ratio of friction velocities at roughness change
α	Constant in model of wake turbulence
α_j	Constants in model of wake turbulence
α^*	Sensitivity coefficients
α^*	Gumbel shape parameter
α	Parameter in constant for wake expansion

β^*	Gumbel scale parameter
β	Constant in model for growth of internal boundary layer
β	Constant in model of wake turbulence
β	View angle of neighbouring wind turbine [deg]
β	Ratio of wake and rotor diameter
β_0	Constant in model for wake expansion
β_v	Correction factor in Teknisk Grundlag (1992) for wind speed
β_l	Correction factor in Teknisk Grundlag (1992) for wind farm configuration
δ	Coefficient of variation of σ_u
δ_l	Coefficient of variation of I
Δ	Palmgren-Miner sum
ε_i	Incremental increase of std. of wind speed fluctuations [m/s]
ϕ	Latitude on Earth [deg]
$\phi_l(\zeta)$	Std. of wind speed fluctuations, distance ζ downwind of row of wind turbines [m/s]
ϕ_0	$=\sigma_0$ Spatially averaged turbulence “over” wind farm [m/s]
$\phi_w(\zeta)$	Increase in std. of wind speed fluctuations [m/s]
γ_E	Euler’s constant
$\Gamma(*)$	Gamma function
φ	Angle of attack [deg]
κ	von Karman’s constant
λ_x	Spectral moment of order x
ν	Frequency of up crossing of mean [Hz]
ν_0	Frequency of up crossing of mean [Hz]
θ	Wind direction [deg]
θ_j	Wind direction for wake no. j [deg]
θ	Phase in narrow band response [rad/s]
θ_w	Characteristic view angle of wake [deg]
$\theta_{w,j}$	Characteristic view angle of wake no. j [deg]
ω	Frequency [rad/s]
ω_0	Frequency of sinusoidal load [rad/s]
ω_0	Eigenfrequency of cylinder [rad/s]
Ω	Angular speed of Earth [rad/s]
ξ	Amplitude in narrow-band process
ξ	Damping ratio
ρ	Air density [kg/m ³]
σ_{add}	Added turbulence in wake [m/s]
σ_b	Std. of wind speed fluctuations in wake [m/s]
$\sigma_{u,eff}$	Weighted std. of wind speed fluctuations in wake [m/s]
$\sigma_{T,wf}$	Std. of wind speed fluctuations over wind farm [m/s]
$\sigma_{u,IEC}$	IEC model for wind speed fluctuation model [m/s]
$\sigma_{u,T}$	Std. of total wind speed fluctuations in wake [m/s]
σ_u	Std. of along-wind wind speed fluctuations [m/s]
σ_{ut}	Std. of wind speed fluctuations transverse to mean flow direction [m/s]
σ_{i0}	Std. of wind speed fluctuations in internal boundary [m/s]
σ_v	Std. of lateral wind speed fluctuations [m/s]
σ_w	Std. of vertical wind speed fluctuations [m/s]
σ_w	Std. of along wind wind speed fluctuations in wake [m/s]
σ_l	Std. of turbulence intensity [m/s]
σ_i	Std. of wind speed fluctuations after wind turbine row i [m/s]
$\bar{\sigma}_u$	mean of σ_u , conditioned on U [m/s]
$\Delta\sigma_l$	Std. of σ_l [m/s]
$\Delta\sigma_u$	Std. of σ_u , conditioned on U [m/s]
σ_l	Std. of turbulence intensity [m/s]

σ_0	Std. of wind speed fluctuations in ambient turbulence, hub-height [m/s]
σ_0	Std. of wind speed fluctuations in wind farm “ambient turbulence” [m/s]
σ_{addwf}	Added wind farm “ambient” turbulence [m/s]
σ_y	Std. of response y of wind turbine structure
σ_{ext}	Extreme turbulence [m/s]
σ_x	Std. of response x
σ_M	Std. of bending moment [Nm]
τ	Difference in mean wind speed from lowest to highest blade-tip position [m/s]
$\vec{\tau}$	Flow shear vector [N/m^2]
τ_w	Mean wind speed deficit in wake [m/s]
ζ	$=x/D_0$, Non-dimensional distance from row of wind turbines

Appendices

A.1 Basic fatigue load concepts

A.2 Flow in the infinitely large wind farm

A.3 Momentum and energy balance in wake

A.1 Basic fatigue load concepts

Material fatigue occurs when a material is exposed to repeated, cyclic loading. It is initialised by micro-cracks that develop into larger cracks, which eventually cause failure of the considered structural component. Being of utmost importance in structural design, considerable scientific efforts have been invested in clarifying the issue. In the context of the model for effective turbulence, only *basic* material fatigue concepts are applied. These concepts are outlined below.

Properties of materials

When exposing under lab-conditions a material to repeated sinusoidal stress cycles with constant amplitude, failure occurs after a number of cycles, n_{fat} . By repeating the experiment with different stress amplitudes, the material's S - N or Wöhler curve is determined, as illustrated in Figure A.1. Over a large range of n_{fat} , the relation between the stress amplitude s_{fat} and the corresponding number of cycles to fatigue follows a power law:

$$n_{fat} = n_{fat}(s_{fat}) = a \cdot s_{fat}^{-m}, \quad (\text{A.1.1})$$

where a and m are material constants. The Wöhler-curve exponent m is for steel of the order 3-5 and for fibreglass 10-12. For most materials and application environment of these, there is a lower and higher limit of s_{fat} for low and high values of n_{fat} , respectively.

In addition to the stress amplitude, the lifetime depends on the mean stress. Under certain restrictions, the effect of the mean may be taken into account by the Goodman criterion, effectively by altering the material constant a in E.q. (A.1.1).

Linear Damage Hypothesis

The Wöhler curve may be used directly when designing for a load with constant amplitude. For N series of sinusoidal loads sequences with different amplitudes, s_i , and number of cycles with that amplitude, $n_i(s_i)$, the damage is evaluated by the Palmgren-Miner sum, Miner (1945):

$$D_T = \sum_i^N D_i, \quad D_i = \frac{n_i(s_i)}{n_{fat}(s_i)}, \quad (\text{A.1.2})$$

where D_i is the partial damage and $n_{fat}(s_i)$ is the number of cycles at the stress range s_i that would cause fatigue at that stress range. Failure occurs when the sum D_T exceeds 1. The function $n_i(s_i)$ is called the (inverse) load spectrum. When more realistically the stress variations are not neat, consecutive sinusoidal series, but an irregular process (e.g. random), the load spectrum $n_i(s_i)$ is determined from the rainflow counting method, originally described by Matsuishi and Endo (1968). The method is found to represent the fatigue mechanism well. In practical terms, the sustainability to fatigue is evaluated by computationally derived load spectra, representative for the lifetime of the considered structural component.

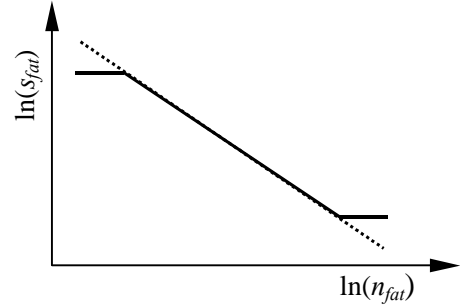


Figure A.1 S - N curve; the solid line is real data and dotted line is idealised curve, Eq. (A1.1).

Combined sinusoid and random process

The loading in terms of stresses of any wind turbine component is a combination of a periodic part and a random part. Present-day response analyses use “brute force” for fatigue load analysis by doing rainflow analysis on simulation results. Alternatively, for comprehension of the combined periodic and random response, it is useful to take an analytical approach.

The periodic response component of any wind turbine component basically has the frequency ω_0 of the rotor revolution. Assuming that the periodic response component is a sinusoid, the combined response may be written as

$$y(t) = y_0 + A \cdot \cos \omega_0 t + x(t), \quad (\text{A.1.3})$$

where y_0 is mean response, A is the amplitude of the sinusoidal response component and x is the stochastic component of the response with zero mean. The stochastic part of the response stems primarily from the dynamic loading of turbulence and for off-shore wind turbines from wave loading. A characteristic frequency of a process is defined by the up-crossing rate of the mean of the process. For a random narrow-band process the rate of up-crossing is given by the expression

$$\nu_0 = \frac{1}{2\pi} \sqrt{\frac{\lambda_2}{\lambda_0}}, \quad \lambda_n = \int_0^\infty \omega^n S_x(\omega) d\omega, \quad (\text{A.1.4})$$

where λ_n are the moments of the power spectrum S_x of x . For the sinusoidal component of the response alone, the up-crossing rate is $\nu_0 = \omega_0/2\pi$. For a stochastic process, the response spectrum must first be evaluated. Thus, consider flap-wise response of a rotating blade mounted on a stiff hub. Assuming only one degree of freedom of the blade, the spectrum of response can be written as

$$S_x(\omega) \propto \int_0^{R_0} \int_0^{R_0} H^2(\omega) S_u(r_1, r_2, \omega) dr_1 dr_2, \quad (\text{A.1.5})$$

where H is the flap-wise frequency response function of the blade and S_u is the cross spectrum of turbulent fluctuations as experienced on the rotating blade, at distances r_1 and r_2 from the wind turbine hub. If the combined structural and aerodynamic damping is low, the up-crossing rate will be close to the eigenfrequency of the blade. If the damping is high, the up-crossing rate will be closer to the frequency scale of turbulence. For the rotating blade the experienced power spectrum of turbulent fluctuations is shifted from lower frequencies to the frequency of rotor revolution and multiples hereof, Kristensen and Frandsen (1982). In all, the response is dominated by three frequencies: the scale of ambient turbulence, the frequency of revolution ω_0 and the structural eigenfrequency. Below, it is assumed – with fair support in practical experience – that the up-crossing rate of the combined process $y(t)$ can be approximated as $\nu_0 \approx \omega_0/2\pi$.

Atmospheric turbulence may – for practical purposes – be considered to be Gaussian distributed, but not narrow banded. However, the response resulting from the turbulent loading will more or less be a narrow band process with frequencies around the structural eigenfrequencies and/or ω_0 . The combined sinusoidal and narrow-band response can be re-written as

$$y(t) = y_0 + A \cdot \cos \omega_0 t + R_x(t) \cdot \cos(\omega_0 t + \theta(t)), \quad (\text{A.1.6})$$

where $R_x(t)$ is a random, Rayleigh distributed amplitude process and $\theta(t)$ a random, uniformly distributed phase process. The discrete Palmgren-Miner sum can be given an integral form in the following way, Crandall and Mark (1963). In the time period T_T for the narrow-band process with up-crossing frequency ν_0 , the average number of peaks are $\nu_0 T_T$. Denoting the PDF of the peaks $f_m(\xi)$, then the number of

peaks that lies in the interval ξ to $\xi + d\xi$ is $n(\xi) = \nu_0 T_T f_m(\xi) d\xi$ and the fractional damage in the interval becomes

$$\frac{n(\xi)}{n_{fat}(\xi)} = \nu_0 T_T \frac{f_m(\xi)}{n_{fat}(\xi)} d\xi. \quad (\text{A.1.7})$$

The expected total damage is then obtained by inserting the S - N curve, Eq. (A.1.1), in Eq. (A.1.7) and integrating the right side of the equation over all amplitudes:

$$E\{D_T\} = \nu_0 T_T \int_0^\infty \frac{f_m(\xi)}{n_{fat}(\xi)} d\xi = \frac{\nu_0 T_T}{a} \int_0^\infty (2\xi)^m f_m(\xi) d\xi \quad (\text{A.1.8})$$

Crandall and Mark (1963) calculate from Eq. (A.1.8) in closed form the expected total damage for the narrow-band random process, corresponding to zero amplitude of the sinusoid in Eq. (A.1.6), $A = 0$. For the combined process of Eq. (A.1.6), Rice (1944) firstly derives the amplitude/peak PDF as

$$\begin{aligned} f_m(\xi) &= \int_0^{2\pi} \frac{\xi}{2\pi\sigma_x^2} \exp\left(-\frac{\xi^2 + A^2 - 2A\xi \cos\theta}{2\sigma_x^2}\right) d\theta \\ &= \frac{\xi}{\sigma_x^2} \exp\left(-\frac{\xi^2 + A^2}{2\sigma_x^2}\right) \cdot I_0\left(\frac{\xi A}{\sigma_x^2}\right) \end{aligned} \quad (\text{A.1.9})$$

where ξ is the amplitude value, σ_x is the standard deviation of the random part of the response and $I_0(*)$ is the modified Bessel function. Secondly, Eq. (A.1.9) is inserted into Eq. (A.1.8) to provide the expected total damage for the combined process:

$$E\{D_T\} = \frac{\nu_0 T_T}{a} \int_0^\infty (2\xi)^m f_m(\xi) d\xi = \frac{\nu_0 T_T}{a} \cdot e_0^m, \quad (\text{A.1.10})$$

where

$$e_0 = 2\sqrt{2} \cdot \sigma_x \cdot \left[\Gamma\left(1 + \frac{m}{2}\right) \cdot M\left(-\frac{m}{2}; 1; -\left(\frac{A}{\sqrt{2}\sigma_x}\right)^2\right) \right]^{1/m} \quad (\text{A.1.11})$$

is the double-amplitude of the sinusoidal, with frequency ν_0 , that causes the same damage as the combined process. a and m are the previously defined material constants, $\Gamma(*)$ is the gamma function, and $M(*;*;*)$ is the confluent hypergeometric function. The function e_0 has the asymptotic values for vanishing random and sinusoidal component, respectively:

$$e_0 \rightarrow 2 \cdot A \text{ for } \sigma_x \rightarrow 0 \quad \text{and} \quad e_0 \rightarrow 2\sqrt{2}\sigma_x \left(\Gamma\left(1 + \frac{m}{2}\right)\right)^{1/m} \text{ for } A \rightarrow 0. \quad (\text{A.1.12})$$

It is noted that for zero amplitude of the sinusoid and conditioned on m the characteristic amplitude e_0 is simply proportional to the standard deviation, σ_x , of the response of the random component alone.

Equivalent width concept

Using a fixed number of reference cycles n_T (e.g. chosen to be equal to the integrated number of range cycles $n_T = \sum n_i$, which in turn is close to the number of rotor revolutions during the considered period, or the above-mentioned up-crossing rate times lifetime), the concept of equivalent load may be introduced:

$$D_T = \sum_i D_i = \sum_i \frac{n_i}{n_{fat}(s_i)} = \frac{n_T}{n_{fat}(e)} \Rightarrow e = n_{fat}^{-1}(D_T^{-1} n_T), \quad (\text{A.1.13})$$

i.e. the equivalent load is the width that creates the same partial damage as the real stress sequences. The above e_0 is the equivalent load with $n_T = v_0 T_T$.

Combination of equivalent widths, discrete formulation

By inserting Eq. (A.1.1) into Eq. (A.1.2) we get

$$D_T = \sum_i \frac{n_i}{a \cdot s_i^{-m}} = \frac{n_T}{a} \sum_i \frac{n_i}{n_T} s_i^m = \frac{n_T}{a} \sum_i p_i s_i^m \quad (\text{A.1.14})$$

where s_i is the stress range that causes failure at the number of cycles n_i and p_i is the probability of the stress range s_i during the total number of load cycles. Analogously, for a series of determined equivalent load, e_i , the resulting equivalent load is determined by weighting each component with its probability:

$$\frac{n_T}{a} e^m = \frac{n_T}{a} \sum_i p_i e_i^m \Rightarrow e = \left(\sum_i p_i e_i^m \right)^{\frac{1}{m}}. \quad (\text{A.1.15})$$

A.2 Flow in the infinitely large wind farm

A model for the effect of a very large wind farm on the planetary boundary layer, Frandsen (1992), Emeis and Frandsen (1993), Frandsen and Madsen (2003) is summarised in this appendix. The first treatment the problem was given by Templin (1974). At the time, the approach was by most people considered farfetched, since wind farms extending many kilometres seemed totally unrealistic.

Based on the simplified geostrophic drag law, Jensen (1978), the approach presented below offers an explicit expression for the apparent roughness of a large wind farm. The advantage of this model over other models, Bossanyi (1980), is that the formulation for “wind farm roughness” is consistent with roughness-change models developed in other research sectors, and that it also offers of an estimate of standard deviation of wind speed fluctuations in the wind farm environment.

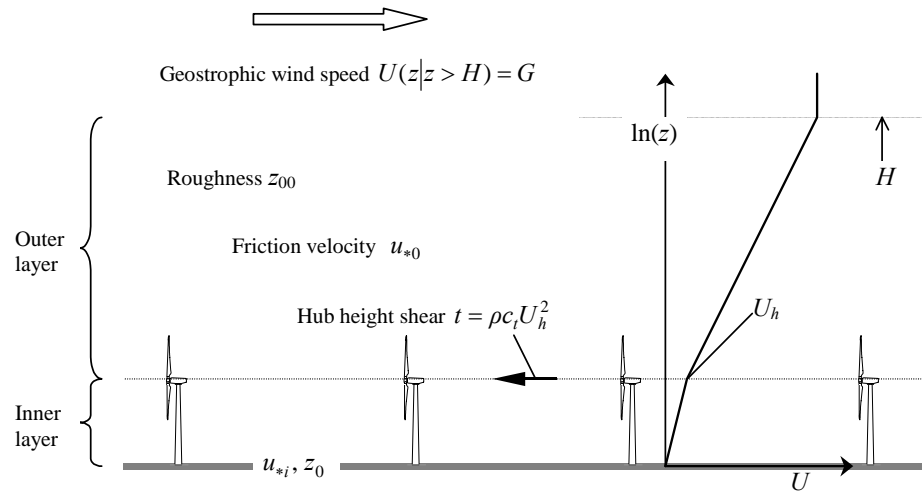


Figure A.2. The impact of an “infinitely” large wind farm on the planetary boundary layer. The difference between G and U_h is exaggerated.

The geostrophic drag law is derived by assuming inertial and viscous forces small (low Rossby and Ekman number) relative to the Coriolis and friction forces, respectively, and pressure force. The drag law for neutral atmospheric stratification can be written as, Tennekes and Lumley (1972),

$$G = \frac{u_*}{\kappa} \sqrt{\left\{ \ln\left(\frac{u_*}{fz_0}\right) - A \right\}^2 + B^2}. \quad (\text{A.2.1})$$

Here, u_* is the friction velocity, z_0 is the surface roughness, $f = 2\Omega \sin \phi$ is the Coriolis parameter and G is the geostrophic wind speed. Ω is the angular speed of Earth and ϕ is the latitude. A and B are constants, which by Troen and Petersen (1989) are estimated to be $A = 1.8$ and $B = 4.5$.

Eq. (A.2.1) is implicit in u_* and for practical purposes an approximation is useful. Jensen (1978) proposed such an approximation, and with an adjustment to that approximation proposed by Emeis and Frandsen (1993), the geostrophic drag law becomes

$$G \approx \frac{u_*}{\kappa} \left(\ln\left(\frac{G}{fz_0}\right) - A_* \right) \Leftrightarrow u_* \approx \frac{\kappa G}{\ln \frac{G}{fz_0} - A_*} = \frac{\kappa G}{\ln \frac{G}{(f \cdot e^{A_*})z_0}}, \quad (\text{A.2.2})$$

where the constant by comparison with Eq. (A.2.1) is estimated to $A_* \approx 4$ at latitude 55° .

Alternatively, when knowing z_0 and u_* at one site, Kristensen et al. (2000) propose the following expression to determine the friction velocity at another site with the same geostrophic wind speed, but with different surface roughness:

$$u_*' \approx u_* \left(\frac{z_0'}{z_0} \right)^{0.07}. \quad (\text{A.2.3})$$

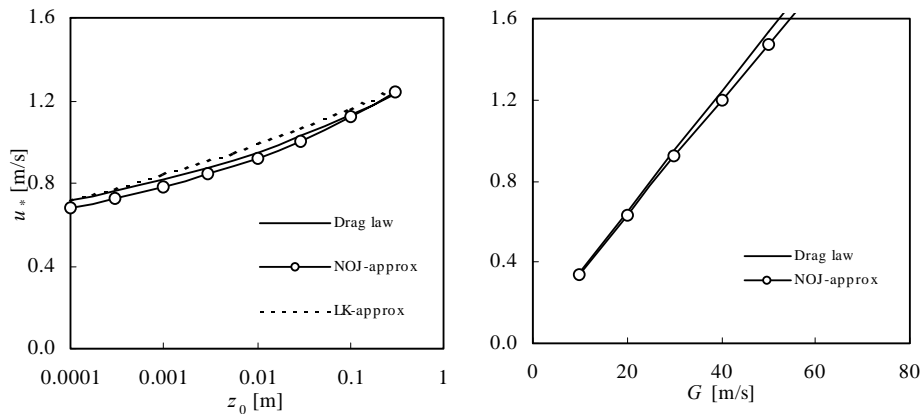


Figure A.3 Comparison of the approximations with the geostrophic drag law. Left plot: friction velocity as function of surface roughness for $G = 30$ m/s. For the LK-approximation, the friction velocity at $z_0 = 0.0001$ m is taken as reference condition. Right plot: friction velocity as function of geostrophic wind speed for $z_0 = 0.01$ m.

The approximations are evaluated in Figure A.3 Both approximations work well.

The wind vector's direction is changing as function of height. The geostrophic drag law does provide estimation of the total direction change, but not as function of height. Evaluation of data²⁶ suggests that – in average – the direction change over the height of a wind turbine is small.

Returning to the model for the influence of the wind farm on the local wind climate, the following assumptions are made:

- The wind farm is large enough for the horizontally averaged, vertical wind profile to be horizontally homogeneous.
- The thrust on the wind turbine rotors is assumed concentrated at hub height.
- The horizontally averaged vertical wind profile is logarithmic over hub height and logarithmic under hub height. This assumption is similar to the assumption for the development of the internal boundary layer after a change of surface roughness.
- The vertical wind profile is continuous at hub height.
- The height of the planetary boundary layer is considerably larger than wind turbine hub height: $H \gg h_H$.
- Horizontally averaged turbulent wind speed fluctuations are horizontally homogeneous.

²⁶ From the met tower at Risø National Laboratory. The tower is 123m high and extensively instrumented.

The log-profile assumption implies that the horizontal shear stress is constant and height-independent over and under hub height, respectively. The friction velocities over and under hub height are denominated u_{*0} and u_{*i} , respectively, and thus the shear stresses over and under hub height are $\rho \cdot u_{*0}^2$ and $\rho \cdot u_{*i}^2$, where ρ is air density.

The vertical wind speed profiles over and under hub height h are given as

$$\frac{U(z)}{u_{*0}} = \frac{1}{\kappa} \ln\left(\frac{z}{z_{00}}\right) \quad \text{for } z \geq h_H \text{ and} \quad (\text{A.2.4})$$

$$\frac{U(z)}{u_{*i}} = \frac{1}{\kappa} \ln\left(\frac{z}{z_0}\right) \quad \text{for } z \leq h_H. \quad (\text{A.2.5})$$

The thrust of each wind turbine unit is

$$T = \frac{1}{2} \rho C_T U_h^2 A_r, \quad (\text{A.2.6})$$

where C_T is the thrust coefficient, $U_h = U(h_H)$ is hub height wind speed and A_r is the swept rotor area. Assume then, that the wind turbines are arranged in rows with separation x_r and separation between rows x_f . The shear-jump at hub height is:

$$t = \frac{T}{x_r x_f} = \frac{\frac{1}{2} \rho C_T U_h^2 \frac{\pi}{4} D_0^2}{x_r x_f} = \rho c_t U_h^2, \quad c_t = \frac{\pi C_T}{8 s_r s_f}, \quad s_r = \frac{x_r}{D_0} \text{ and } s_f = \frac{x_f}{D_0}. \quad (\text{A.2.7})$$

The shear stress over and below hub height is linked by

$$\rho u_{*0}^2 = \rho u_{*i}^2 + \rho c_t U_h^2. \quad (\text{A.2.8})$$

For the flow above hub height, the geostrophic drag law, Eq. (A.2.2), is applied. Doing so, the vertical profile of the horizontal velocity may be deduced.

The assumption of continuity of the wind profile at hub height gives the following equations:

$$\frac{U_h}{u_{*i}} = \frac{1}{\kappa} \ln\left(\frac{h_H}{z_0}\right) \quad \text{and} \quad \frac{U_h}{u_{*0}} = \frac{1}{\kappa} \ln\left(\frac{h_H}{z_{00}}\right). \quad (\text{A.2.9})$$

From these equations, the following convenient expression for the apparent roughness of the wind farm area is found:

$$z_{00} = h_H \cdot \exp\left(-\frac{\kappa}{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}\right). \quad (\text{A.2.10})$$

Finally, the above-hub-height friction velocity and the hub height wind speed are found as:

$$u_{*0} = \frac{\kappa G}{\ln\left(\frac{G}{f' h_H}\right) + \frac{\kappa}{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}} \quad \text{and} \quad (\text{A.2.11})$$

$$U_h = \frac{G}{1 + \ln\left(\frac{G}{f' h_H}\right) \frac{\sqrt{c_t + (\kappa / \ln(h_H / z_0))^2}}{\kappa}}, \quad (\text{A.2.12})$$

where $f' = (1.2 \cdot 10^{-4} \cdot e^4)$ at latitude 55° . Assuming that the contribution from terrain surface to wind farm roughness, z_{00} , is negligible, i.e.

$$z_{00} = h_H \cdot \exp\left(-\frac{\kappa}{\sqrt{c_t}}\right), \quad (\text{A.2.13})$$

an approximation leading to the added wind farm turbulence I_{addwf} (Eq (2.18) in Section 2) may be obtained:

$$I_{addwf} \approx \frac{\frac{1}{\kappa} u_{*addwf}}{U_{h,free}} \approx \frac{\ln\left(\frac{G}{f' \cdot z_0}\right)}{\ln\left(\frac{h_H}{z_0}\right)} \cdot \frac{1}{\ln\left(\frac{G}{f' \cdot h_H}\right) + 0.64 \cdot \sqrt{s_r s_f / C_T}} \quad (\text{A.2.14})$$

Thus, the turbulence intensity added due to the presence of the wind farm is a function of s_r , s_f and C_T , and to a lesser extent also hub height of the wind turbines h_H , roughness of the terrain surface z_0 and the geostrophic wind speed G .

A.3 Momentum and energy balance

The first part of this appendix is textbook material, included to support the author's own fading memory. Possibly, it may serve as an aid to the readers too. The proceeding part comprises considerations as to maximum initial wake turbulence by application of both momentum and energy balance.

Momentum and energy-flux balance for control volume around wind turbine rotor, with non-turbulent wake

A cylindrical control volume with its horizontal axis parallel to the mean wind vector is defined, Figure A.4. From Engelund (1968), the equation for balance of forces acting on the control volume (the momentum equation) in vector form for the flow volume X with the surface area A_T is

$$\int_X \rho \frac{\partial \vec{U}}{\partial t} dX + \int_{A_T} \rho \vec{U} (\vec{U} d\vec{A}) = - \int_{A_T} p d\vec{A} + \int_X \rho \vec{g} dX + \vec{T} + \int_{A_T} \vec{\tau} dA. \quad (\text{A.3.1})$$

where the acceleration term (first on the left side), the pressure term (first on the right hand side) and the gravity term (second on the right hand side) often, as is done in the following, are neglected in basic considerations. Further, the cylinder extends upwind and downwind far enough²⁷ for pressure at the right end of the control volume to be equal to the free-stream pressure. \vec{U} is the instantaneous flow velocity vector and \vec{T} is the sum of forces from obstacles acting on the interior of the control volume and the last term on the right hand side is the turbulent shear forces acting on the control volume surface.

Should it be acceptable to neglect shear forces in the cylinder surface and assume that pressure downwind has regained the free-stream level, the momentum equation reduces to

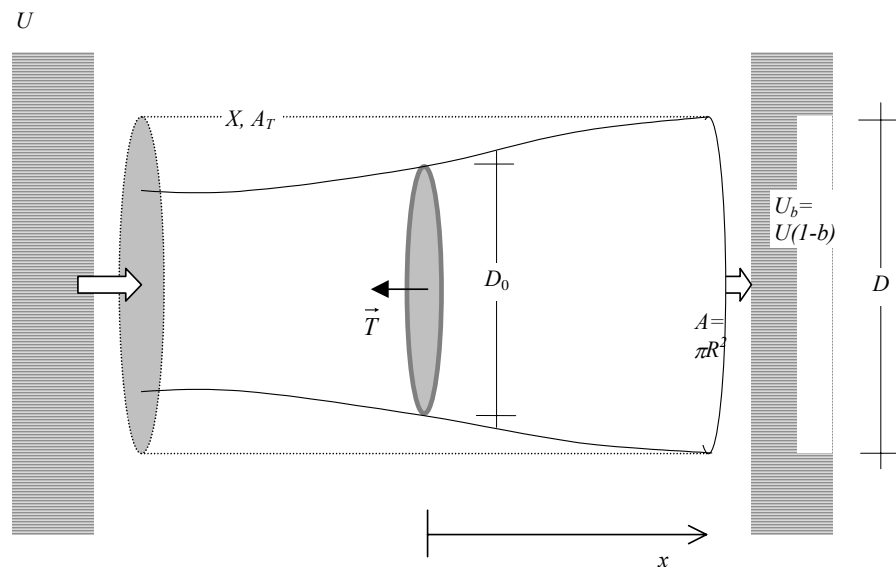


Figure A.4 Control volume around a wind turbine rotor. The cylindrical control volume has the surface area A_T and volume X . The Betz control volume (full line) follows the streamlines.

²⁷ Upstream of the order $\frac{1}{2}$ to 1 rotor diameter and downwind 2-3 rotor diameters.

$$\bar{T} = -\int_{A_r} \rho \bar{U} (\bar{U} d\bar{A}) = -\int_{A_r} \bar{U} dQ, \quad (\text{A.3.2})$$

where dQ is the volume flow out of the surface area dA .

Assuming the wake to be non-turbulent, then the flow vector in the control volume surface may be approximated by the mean flow speed and the expression can be developed further. The momentum flux out of the cylinder surface is found the following way: the volume flow (per sec.) out of the control volume's cylinder surface is equal to minus the volume flow Q_e out of the ends of the cylinder (with area $A = \pi R^2$),

$$Q_{cyl} = -Q_e = \int_A \rho U dA - \int_A \rho U_b dA, \quad (\text{A.3.3})$$

where U is the free-flow mean speed and U_b is the flow speed in the wake. Assuming that the radius of the cylinder is sufficiently large for the flow speed in the cylinder surface to be approximated as U , then the momentum flux out of the cylinder surface becomes

$$M_{cyl} = U \cdot Q_{cyl}. \quad (\text{A.3.4})$$

The rotor thrust becomes

$$T = M_e + M_{cyl} \Rightarrow T = \int_A \rho U_b (U - U_b) dA. \quad (\text{A.3.5})$$

This expression is the classical starting point for development of wake models, e.g. Engelund (1968), Schlichting (1968) and Tennekes and Lumley (1972). In a similar way, the energy-flux budget for the same control volume can be determined. With the assumption of no dissipation, the net energy out-flux plus the drainage in the form of power extraction by the wind turbine P shall be zero:

$$\int_{A_r} \frac{1}{2} \rho U^2 (\bar{U} d\bar{A}) + P = 0. \quad (\text{A.3.6})$$

With application of the same assumptions as above, the below expression for the wind turbine's power is obtained:

$$P - \int_A \frac{1}{2} U^2 (\rho U dA) + \int_A \frac{1}{2} U_b^2 (\rho U_b dA) + \frac{1}{2} \left(\int_A \rho (U - U_b) dA \right) \cdot U^2 = 0 \Rightarrow$$

$$P = \int_A \frac{1}{2} \rho U_b (U^2 - U_b^2) dA \quad (\text{A.3.7})$$

Assuming rectangular distribution of the flow-speed deficit downwind, the expressions for rotor thrust and power from the wind turbine become

$$T = \rho \pi R^2 U_b (U - U_b) \quad \text{and}$$

$$P = \frac{1}{2} \rho \pi R^2 U_b (U^2 - U_b^2) \quad (\text{A.3.8})$$

Momentum and energy flux in isotropic turbulent flow

Before proceeding, consider the momentum flux through a unit-area perpendicular to the mean flow vector. The length of the component in the mean flow direction is $U+u$. The turbulent perturbations are denominated u, v and w , $\bar{u} = \bar{v} = \bar{w} = 0$. The average momentum flux through the unit-area is

$$M = \overline{q(U+u)} = \rho \overline{(U+u)(U+u)} = \rho U^2 + \rho \sigma_u^2, \quad \sigma_u^2 = \overline{u^2}, \quad (\text{A.3.9})$$

where over-bars denote time averaging. The quantity q is the mass flux (kg/s).

Similarly, with the additional assumption that fluctuations of speed in one direction are independent of fluctuations in another direction, the average energy flux through the unit-area can be estimated:

$$\begin{aligned} p &= \overline{q \cdot (U + u)} = \frac{1}{2} \rho \cdot \overline{((U + u)^2 + v^2 + w^2)} \cdot (U + u) \\ &= \frac{1}{2} \rho \cdot \overline{(U^2 + 3U^2u + 3Uu^2 + u^2)} + \frac{1}{2} \rho U \overline{(v^2 + w^2)} \end{aligned} \quad (\text{A.3.10})$$

With the assumption of isotropy, $\sigma_u = \sigma_v = \sigma_w (= \sigma_b)$ ²⁸, the energy flux becomes

$$p \approx \frac{1}{2} \rho U^3 + \frac{5}{2} \rho U \sigma_u^2 = \frac{1}{2} \rho U^3 (1 + 5I^2), \quad I = \frac{\sigma_u}{U}, \quad (\text{A.3.11})$$

where I is the turbulence intensity.

Momentum and energy-flux balance for control volume, with turbulent wake

With the addition of turbulence in the wake (other control volume characteristics remain unchanged), the average thrust and power extraction of the rotor are

$$T = \int_A \rho U_b (U - U_b) dA - \int_A \rho \sigma_b^2 dA \quad \text{and} \quad (\text{A.3.12})$$

$$P = \int_A \frac{1}{2} \rho U_b (U^2 - U_b^2) dA - \int_A \frac{5}{2} U_b \sigma_b^2 dA. \quad (\text{A.3.13})$$

With the assumptions of rectangular profiles for flow speed deficit and wake turbulence – both $2R$ wide – the quantities become

$$T = \rho \pi R^2 [U_b (U - U_b) - \sigma_b^2] \quad \text{and} \quad (\text{A.3.14})$$

$$P = \frac{1}{2} \rho \pi R^2 [U_b (U^2 - U_b^2) - 5U_b \sigma_b^2]. \quad (\text{A.3.15})$$

Lanchester/Betz solution for a wind turbine rotor

In the wind turbine context, thrust and power is by convention referred to the free-flow speed U , the swept area of the rotor, $A_r = \pi R_0^2$, and the thrust coefficient C_T and power coefficient C_P of the rotor:

$$C_T = \frac{T}{\frac{1}{2} \rho U^2 A_r} \quad \text{and} \quad C_P = \frac{P}{\frac{1}{2} \rho U^3 A_r}. \quad (\text{A.3.16})$$

When disregarding wake turbulence thrust and power coefficients can be expressed as

$$C_T = \frac{2}{A_r} \int_A b(1-b) dA, \quad \text{and} \quad C_P = \frac{1}{A_r} \int_A b(1-b)(2-b) dA, \quad (\text{A.3.17})$$

where $b = 1 - \frac{U_b}{U}$ is the wake induction factor. In order to arrive at the Lanchester/Betz result for the wind turbine rotor's maximum efficiency on basis of the above traditional derivations, one must make probable that the wind speed deficit in the rotor plane is half the wake deficit²⁹, $U_r = (1 - \frac{1}{2}b) \cdot U$. This – together with

²⁸ For the free boundary layer, this assumption of equal standard deviation of the three components is crude. However, in the wake where the flow has been “stirred up” by the rotor, the assumption may fit better than in the free boundary layer.

²⁹ In the Betz formulation, the flow speed in the rotor plane is argued by equating the work done on the rotor with the change in kinetic energy in the flow when passing the rotor.

continuity within the Betz stream tube, see Figure A.4 – gives the ratio between the rotor diameter and the immediate downwind diameter of the wake:

$$\left(\frac{D_0}{D}\right)^2 = \frac{1-b}{1-(b/2)}. \quad (\text{A.3.18})$$

Applying this and assuming rectangular flow-speed deficit yields the following relations between the thrust and power coefficients of the rotor for the non-turbulent wake:

$$C_T = b(2-b) \quad \text{and} \quad C_P = \frac{1}{2}(2-b)^2 b. \quad (\text{A.3.19})$$

Finding the maximum of C_P by means of differentiation leads to the famous maximum efficiency of a wind turbine rotor, $C_{p,max} = 16/27 \approx 59\%$ ³⁰. The validity of this result is of course limited by the massive amount of assumptions. Nevertheless it has served as a landmark in developing also modern wind turbine rotors, where presently the maximum obtainable efficiency seems to be approx. 45+%, i.e. 45% of the kinetic energy passing through the rotor's swept area can be extracted.

Extending the Lanchester/Betz results to turbulent wake conditions

Still assuming rectangular wake deficit, but including turbulence in the wake, the expressions for thrust and power coefficients become

$$C_T = b(2-b) - \frac{2-b}{1-b} I^2 \quad \text{and} \quad C_P = \frac{1}{2}(2-b)^2 b - \frac{5}{2}(2-b) I^2. \quad (\text{A.3.20})$$

The maximum efficiency is found from differentiation of C_P with respect to b :

$$\frac{\partial C_P}{\partial b} = (2 + \frac{5}{2} I^2) - 4b + \frac{3}{2} b^2 = 0 \quad \text{for} \quad b = \frac{4 - \sqrt{4 - 15 \cdot I^2}}{3}. \quad (\text{A.3.21})$$

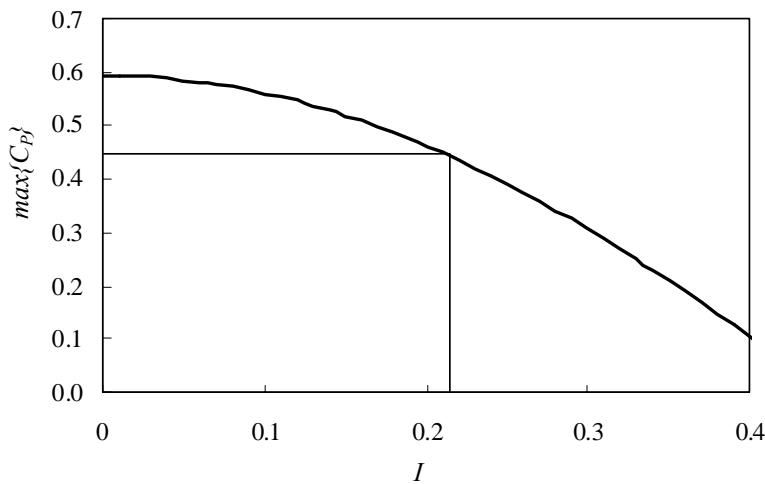


Figure A.5 Maximum C_P as function of turbulence intensity immediately downwind of the rotor.

³⁰ Larger efficiencies *can* be achieved by augmenting the tips of the blades or by building a flow-concentrator around the whole rotor, thus altering the “effective” swept rotor area and thereby changing the pre-conditions imposed herein. Such designs have so far proven uneconomic.

Thus, the maximum efficiency drops off with increasing turbulence behind the rotor, Figure A.5. As pointed out, the state-of-the-art performance of a wind turbine rotor is an efficiency of approx. 45%, corresponding to a turbulence intensity of 22%, i.e. if the near-wake turbulence is 22%, the efficiency cannot be larger than 45%

Next, for the Equations (A.3.20), the range of solutions for wake-induced velocity and C_T , conditioned on C_P are found, Figure A.6. The minimum values of b and C_T correspond to the Lanchester/Betz solution, i.e. with $I = 0$.

The maximum values, which allow solution of Eqs. (A.3.20), effectively also correspond to maximum near-wake turbulence, Figure A.7.

It is noted that the maximum initial wake turbulence is approx. 37%. Seemingly, the value can – with the chosen assumptions – only be achieved at low rotor efficien-

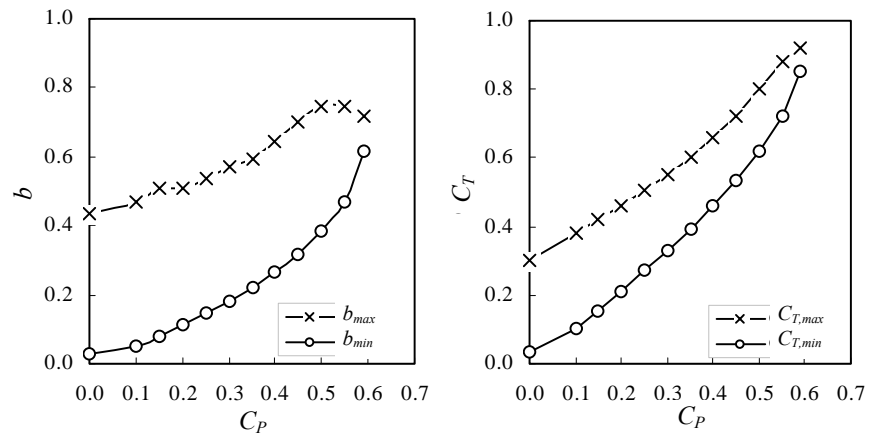


Figure A.6 Maximum and minimum wake induction (left) and thrust coefficient, as function of wind turbine efficiency.

cies.

It is also interesting to note that for high wind speeds, where $C_P \rightarrow 0$, the equations (A.3.20) yield

$$C_T \cong 3 \cdot I^2 \Rightarrow I \cong \sqrt{\frac{C_T}{3}}, \quad (\text{A.3.22})$$

e.i. for the near-wake at high wind speeds, wake turbulence is proportional to the square root of the thrust coefficient as is the case for the chosen model for wake turbulence. Further, in the limit $C_P \rightarrow 0$, the maximum, possible turbulence intensity is $\frac{1}{\sqrt{3}} \approx 0.45$ (45%).

From a set of measurements of power and tower bending moment, the thrust and power coefficients were deduced. With known C_T and C_P , the equations (A.3.20) may be solved. The result is shown in Figure A.8, in which also the maximum possible turbulence and the values corresponding to the model proposed in Section 3 are shown. The usual approach dictates that wake turbulence is a function of C_T alone and in that light, the result is surprising with the near-wake turbulence being more or less constant over the wind speed range considered – despite that C_T drops off when the wind speed increases.

It is difficult to verify the findings, since very shortly downwind of the rotor, the primary source of turbulent production changes to radial shear.

These results should be taken with significant caution because of the many simplifying assumptions. In particular, non-rectangular cross-wake distribution of deficit and turbulence may affect the outcome.

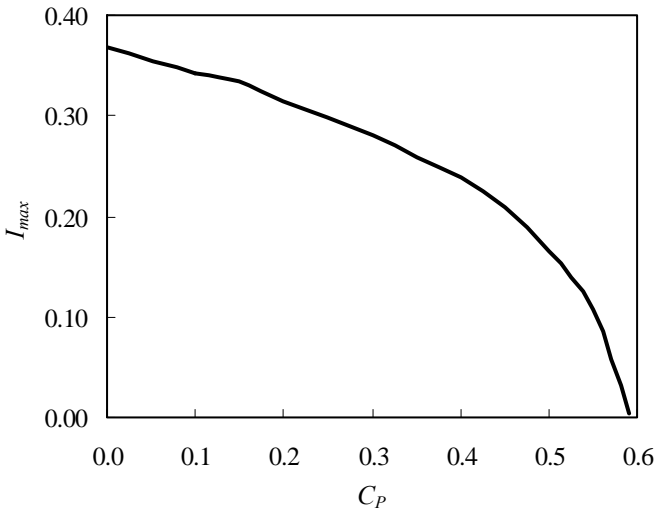


Figure A.7 Maximum possible turbulence intensity as function of power coefficient, C_P .

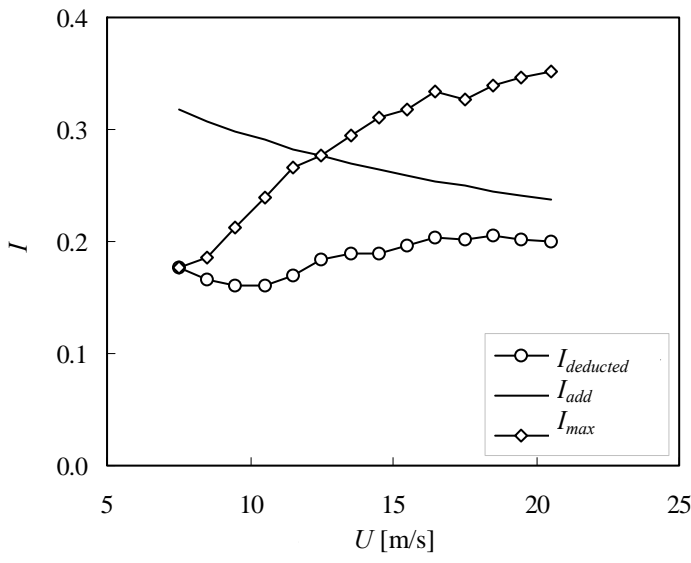


Figure A.8 Turbulence intensity as function of wind speed; $I_{deducted}$ is derived from Eqs. (A3.20), I_{max} is the maximum possible and I_{add} corresponds to the proposed model for added wake turbulence for $s=2$.

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Title: Turbulence and turbulence-generated structural loading in wind turbine clusters
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Abstract (max. 2000 char.):

Turbulence – in terms of standard deviation of wind speed fluctuations – and other flow characteristics are different in the interior of wind farms relative to the free flow and action must be taken to ensure sufficient structural sustainability of the wind turbines exposed to “wind farm flow”. The standard deviation of wind speed fluctuations is a known key parameter for both extreme- and fatigue loading, and it is argued and found to be justified that a model for change in turbulence intensity *alone* may account for increased fatigue loading in wind farms. Changes in scale of turbulence and horizontal flow-shear also influence the dynamic response and thus fatigue loading. However, these parameters are typically – negatively or positively – correlated with the standard deviation of wind speed fluctuations, which therefore can, if need be, represent these other variables. Thus, models for spatially averaged turbulence intensity inside the wind farm and direct-wake turbulence intensity are being devised and a method to combine the different load situations is proposed. The combination of the load cases implies a weighting method involving the slope of the considered material’s Wöhler curve. In the context, this is novel and necessary to avoid excessive safety for fatigue estimation of the structure’s steel components, and non-conservatism for fibreglass components. The proposed model offers significant reductions in computational efforts in the design process. The status for the implementation of the model is that it became part of the Danish standard for wind turbine design DS 472 (2001) in August 2001 and it is part of the corresponding international standard, IEC61400-1 (2005).

Also, extreme loading under normal operation for wake conditions and the efficiency of very large wind farms are discussed.

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